1 Assessment of overlay masonry strengthening system under in-plane

## 2 monotonic and cyclic loading using the diagonal tensile test

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### 7 ABSTRACT

8 The development of novel strengthening techniques to address the seismic vulnerability 9 of masonry elements is gradually leading to simpler, faster and more effective 10 strengthening strategies. In particular, the use of fabric reinforced cementitious matrix 11 systems is considered of great potential, given the increase of ductility achieved with 12 simple and economic strengthening procedures. To assess the effectiveness of these 13 strengthening systems, and considering that the seismic action is involved, one important 14 component of the structural behaviour is the in-plane cyclic response. In this work is 15 discussed the applicability of the diagonal tensile test for the assessment of the cyclic 16 response of strengthened masonry. The results obtained allowed to assess the 17 contribution of the strengthening system to the increase of the load carrying capacity of 18 masonry elements, as well as to evaluate the damage evolution and the stiffness 19 degradation mechanisms developing under cyclic loading.

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1 **Keywords**: infill masonry, high ductility, strengthening system, shear capacity, in-plane

2 cyclic tests, experimental characterization

3

# 4 HIGHLIGHTS

- The efficiency of FRCM overlays was assessed as part of a masonry
  strengthening system.
- The strength increment of masonry walls was quantified by diagonal tensile tests.
- Monotonic and cyclic in-plane loading was applied to evaluate the behaviour of
  the walls.
- The key failure modes obtained experimentally and the basic mechanisms were
  examined.
- Experimental results were discussed and compared to the analytical model
   predictions.
- 14

# 1 Nomenclature:

3	FRCM	Fabric reinforced cementitious matrix
4	$A_f$	Area of the mesh reinforcement by unit width
5	$A_m$	Interface loading area between the steel shoe and the specimen
6	$A_n$	Net area of the specimen cross section
7	A <sub>n,r</sub>	Net area of the render cross section
8	A <sub>n,PFRM</sub>	Net area of the PFRM cross section
9	$E_f$	Tensile modulus of elasticity of the cracked FRCM specimen
10	$f_{fv}$	Design tensile strength of FRCM reinforcement
11	$f_{fv,ult}$	Ultimate tensile strength of FRCM reinforcement
12	$f'_m$	Compressive strength of the masonry
13	$f'_t$	Tensile strength of the masonry
14	g	Vertical gage length
15	G	Modulus of stiffness in shear
16	h	Height of the brick units
17	L	Length of the wall in the direction of the shear force
18	n	Number of fabric layers
19	<i>n</i> *	Percentage of the net area
20	Р	Applied load
21	S <sub>s</sub>	Shear stress
22	t	Thickness of the specimen
23	$V_{ref}^{exp}$	Shear capacity of the reference specimens obtained experimentally
24	$V_{str}^{exp}$	Shear capacity of the strengthened specimens obtained experimentally
25	Van	Shear capacity of the specimen calculated following the ACI549.4R-13
26	Van VACI+PFRM	Shear capacity of the specimen calculated following ACI549.4R-13 plus
27		the contribution of the PFRM layers

1	$V^{an}_{Sup}$	Shear capacity of the specimen considering ultimate strain of the FRCM
2		plus the contribution of the PFRM layers
3	V <sub>dt</sub>	Shear capacity due to diagonal tension failure
4	$V_f$	Shear capacity of the FRCM system
5	V <sub>m</sub>	Shear capacity of the masonry specimen
6	V <sub>sf</sub>	Shear capacity due to shear friction failure
7	V <sub>ss</sub>	Shear capacity due to shear sliding failure
8	V <sub>f,ult</sub>	Shear capacity of the FRCM system considering ultimate tensile strain
9	$V_r$	Shear capacity of the render layer
10	V <sub>PFRM</sub>	Shear capacity of the PFRM layer
11	$u_v$	Vertical shortening
12	u <sub>h</sub>	Horizontal elongation
13	W	Length of brick units
14	γ	Shearing strain
15	Efv	Design value of the tensile strain of FRCM reinforcement
16	E <sub>fv,ult</sub>	Tensile strain ultimate value of FRCM reinforcement
17	$\mu_0$	Coefficient of internal shear friction in mortar joint
18	$\mu_m$	Modified coefficient of internal shear friction in mortar joint
19	$ au_0$	Shear bond strength of the mortar joint
20	$ au_{0,m}$	Modified shear bond strength of the mortar joint
21	θ	Angle between horizontal and main diagonal of the wall
22		

#### 1 1. INTRODUCTION

2 Masonry constructions are composed of brittle or quasi-brittle materials, and generally 3 have low resistance to seismic events. The current performance requirements 4 prescribed by the design codes are frequently not accomplished by existing 5 constructions, either because these requirements became more demanding, or because 6 the negative effects of aging in the long-term behaviour of the materials resulted in a 7 substantial decrease of the load carrying capacity of the masonry elements. Therefore, 8 the development of effective procedures to retrofit existing masonry constructions, in 9 order to upgrade their load bearing capacity and increase their ductility response, is still 10 of great importance.

11 This work presents the experimental program developed with the aim of characterizing 12 and quantifying the contribution of a strengthening system based on fabric reinforced 13 cementitious matrix (FRCM) to the increase of the load carrying capacity and 14 deformability of masonry elements subjected to in-plane loading. In addition, this 15 research investigates the adequacy and effectiveness of the application of the diagonal 16 tensile test for the evaluation of the in-plane cyclic behaviour of strengthened masonry 17 elements. The test procedure, which will be detailed in the following sections, was 18 adapted in order to consider unidirectional cycles of loading and unloading.

19

### 20 1.1. Overlay strengthening techniques

21 The application of additional overlays to the existing masonry elements is a common 22 strengthening technique, especially in areas of considerable seismic activity. Generally, 23 the strengthening overlay consists of a mortar, applied manually or mechanically. These 24 surface treatments, as designated by Elgaway et al., [1], typically incorporate different 25 types of steel, polymers, carbon or glass fibres and fibre meshes used to enhance tensile 26 behaviour and ductility of the strengthening overlay, [2-6]. The application of these 27 reinforced strengthening overlays improves both the in-plane and the out-of-plane load 28 carrying capacity, [7]. A different concept of overlay strengthening system was developed

1 by using materials with tensile strain-hardening behaviour in the hardened state, avoiding 2 the use of reinforcement meshes. These materials, with the designation of strain 3 hardening cement composites (SHCC), have a tensile strength higher than the stress at 4 crack initiation, and a tensile strain at tensile strength higher than 1%, with the capacity 5 of developing diffuse crack patterns of a maximum crack width does not exceeding 0.1 6 mm in the hardening phase. The SHCC can be applied using the shotcreting technique 7 or manually, [8-9]. This technique can lead to the increase of the shear capacity of the 8 masonry, to the improvement of its deformability and to the enhancement of its energy 9 dissipation capacity during cyclic loading, [10].

10 Some of the advantages and disadvantages of the masonry strengthening techniques 11 based on the addition of strengthening overlays to the original masonry element are 12 presented by Elgaway et al, [1], [11]. The advantages identified include the low cost, the 13 durability, the uniform behaviour, the increase of in-plane strength up to 3.6 times, the 14 improvement of the out-of-plane stability and the increase of the energy dissipation ability 15 before failure. The increase of the dead weight of the strengthened elements, the 16 requirement of surface treatments, the architectural changes of the structure, and the 17 high disturbance during works are the main disadvantages identified, [1], [11]. The 18 increase of the stiffness in shear walls due to the application of thick strengthening 19 overlays may also lead to alterations of the structural behaviour. These alterations often 20 result in substantial increments of the stresses at these elements, which in turn may lead 21 to the increase of their strength requirements. In addition, the greater stiffness of the 22 strengthening overlays, when compared to the original substrates, is especially 23 demanding at the level of the interface, where high stresses are generated and the 24 delamination of the overlay is promoted.

25

#### 26 1.2. In-plane experimental characterization

The characterization of the structural behaviour of masonry elements is typically dividedin two main different types, depending on the loading configuration: the in-plane and the

1 out-of-plane behaviours, [12-13]. Concerning the in-plane characterization, the 2 monotonic shear behaviour of elements is often assessed by means of the diagonal 3 tensile tests, using masonry specimens built in the laboratory, [14-15], or in-situ [16-17]. 4 Different types of strengthening systems have already been evaluated using this type of 5 test, [10], [17-20]. As discussed in previous publications, Almeida et al. [21] the cyclic 6 shear behaviour of masonry panels can be characterized by means of the diagonal 7 tensile test. Brignola et al [16] conducted force controlled cyclic tests, with a load gradient 8 of 200 N/s, and the loading procedure was defined in order to obtain four or five cyclic 9 load steps for each masonry panel. In the experimental campaign presented by Santa-10 Maria et al [22], the test procedure started with the first diagonal loading and unloading, 11 followed by the loading of the second diagonal and unloading. Two cycles were 12 performed at each load level, the load increments at each level were of 25 kN.

The diagonal tensile test is regarded as a simple and expedite procedure for the in-plane behaviour characterization. However, it is important to stress that the capability of this type of test to disclose all the failure modes of masonry walls is limited. For example, the rocking/crushing mechanism cannot be evaluated, [17]. For the evaluation of those mechanisms, other experimental procedures, such as the one used by Pinho et al, [23], or Vasconcelos et al, [24] must be carried out.

19 A summary of data available in literature referred to diagonal tensile tests on 20 unreinforced clay brick masonry strengthened with additional overlays or composite 21 materials is presented in Table 1. Different types of masonry and strengthening systems 22 are presented. It is possible to verify that most of the authors use a reduced amount of 23 specimens for the characterization of a specific layout. However, whenever possible the 24 coefficient of variation was calculated and the results obtained are in general quite low, 25 especially for strengthened panels. Therefore the use of a reduced number of specimens 26 seems to lead to sufficiently representative experimental results.

		Dimension of	Dime	ension o	f the			_			Canacity
Reference	Type of masonrv	blockst	h	panel /	t	Nº of panels	Type of reinforcement	F <sub>max</sub>	$S_s$	CoV	increment
	,	(mm)		(mm)				(kN)	(MPa)	(%)	(%)
Santa-		290x140x112	1060	1100	140	5		132		15	
						4	2 diagonal 150 mm-wide CFRP laminate	226		2	71
Maria et	HCBM					5	2 diagonal CFRP fabric	196		4	48
1 (20)	HOBIN					4	2 horizontal 150 mm-wide CFRP laminate	144		6	9
al. [22]						5	2 horizontal CFRP fabric	182		8	38
		210x100x50	870	840	100	2		234			
Gabor et						1	unidirectional glass fibre, RFV	332			42
al. [25]						1	unidirectional carbon fibre, RFC	361			54
						1	bidirectional glass fibre, RFW	384			64
	DCBM	87x84x56	350	350	70	2			0.95		
						2	CFRP, strips		1.98		108
	TOLIN					2	CFRP, full surface		2.25		137
Ozsavin			500	500	70	2			0.92		
et al [26]	PCBM +CM					2	CFRP, strips		1.29		40
						2	CFRP, full surface		2.38		159
	PCBM					2			0.6		
	+CLM					2	CFRP, strips		1.39		132
						2	CFRP, full surface		1.16		93
	SCBM one-	230x110x70	700	700	110	3		25		2	
	wythe				160	3	unidirectionally anchored welded wire mesh	164		7	561
Kadam et	thick+CM					3	bidirectionally anchored welded wire mesh	174		13	602
al. [2]	SCBM two-		750	750	230	3		46		7	
	wythe				280	3	unidirectionally anchored welded wire mesh	236		8	413
	thick+CM					3	bidirectionally anchored welded wire mesh	223		15	385
Dehohani		225x105x40	450	450	105	2		51			
et al. [10]	SCBM+CM					8	ECC, (ft=3.8; fc=47 MPa), LT:10 and 15 mm	70-142			37-178

# 1 Table 1 - Summary of previous diagonal tensile test results on strengthened URM specimens.

2 Where: PCBM= perforated clay brick masonry; HCBM= hollow clay brick masonry; SCBM= solid clay brick masonry; CM= cement mortar; CLM= cement and lime mortar;

3 LT= layer thickness.

1 The test setup adopted for the diagonal tensile tests is represented in Figure 1 and complies with the experimental procedure described by ASTM-E519-02, [12]. The 2 3 dimensions of the panels were set to 0.99x0.99 m considering equipment constrains, 4 symmetry conditions and the dimensions adopted by other authors using similar brick 5 dimensions and panel configurations. As discussed by Brignola et al. [16] and Calderini et al., [27], and evaluated by Frocht [28] the elastic solution considering homogeneous 6 7 isotropic continuum providing the following stress state at the centre of the panel when only a diagonal compressive load is applied, q=0:  $\sigma_x=\sigma_y\cong-0.56P/A$ ;  $\tau_{xy}=$ 8 1.05P/A;  $\sigma_I \cong 0.5P/A$ ;  $\sigma_{II} \cong -1.62P/A$ . Where A is the transversal area of the panel, P 9 10 is assumed as positive and the loading direction angle always being 45°. The Mohr's 11 representation of the stress state in this region is depicted in Figure 2.

12



13

Figure 1 – Diagonal tensile test: a) general set-up

15



Figure 2 – Mohr's representation of the stress state assumed in the central area of the
specimen.

5 The values of the shear stress, shear strain and modulus of shear stiffness in the centre 6 of the panel are also obtained according the procedures indicated in ASTM-E519-02, 7 [12], in this case the stress state is characterized by a  $\tau_{xy} = \sigma_I = \sigma_{II}$ , [16]. The shear 8 stress is obtained by the equation (1):

$$S_s = \frac{0.707 \times P}{A_n} \tag{1}$$

9

10 where *P* is the applied load,  $A_n$  is the net area of the specimen's cross-section calculated 11 according to equation (2):

$$A_n = \left(\frac{w+h}{2}\right) \times t \times n^* \tag{2}$$

12

where w, h and t are, respectively, the width, the height, and the total thickness of the specimen, and  $n^*$  is the percentage of the gross area of the unit that is solid, expressed as a decimal.

16 The shear strain is computed as shown in equation (3):

$$\gamma = \frac{u_v + u_h}{g} \tag{3}$$

1 where  $\gamma$  is the shearing strain,  $u_{\nu}$  is the vertical shortening,  $u_{h}$  is the horizontal elongation

2 and *g* is the vertical gage length.

3 Finally, the modulus of stiffness in shear is calculated as shown in equation (4):

$$G = \frac{S_s}{\gamma} \tag{4}$$

4

## 5 2. OBJECTIVES AND RESEARCH SIGNIFICANCE

6 This research discusses the relevance of diagonal tensile test for the characterization of 7 the "in-plane" monotonic and cyclic behaviour of masonry specimens. An expeditious 8 procedure is developed and applied to quantify the contribution of the reinforcement 9 system to improve the masonry behaviour. The identification and interpretation of the 10 damage mechanisms that develop during the diagonal tensile tests are presented, and 11 an extrapolation of these to practical situations is discussed. Finally, a review of 12 analytical calculation procedures to estimate the load carrying capacity and identify the 13 mechanisms leading to rupture of strengthened masonry is carried out, and the results 14 are presented and discussed.

15

#### 16 3. MATERIALS AND METHODS

### 17 3.1. Materials used in the experiments

18 The materials used in the preparation of the masonry elements, representative of infill 19 walls, and of the strengthening system were selected with the intention of representing 20 regional real cases as much as possible. Although lime mortars are also frequently used 21 in the structural rehabilitation of masonry walls, the system investigated was only based 22 on Portland cement as binder. The masonry specimens were assembled using ceramic 23 bricks (Length x Height x Thickness: 28.5x19.5x11.0 cm<sup>3</sup>) and the following cement 24 mortar for the joints: Portland cement 32.5N, medium graded river sand and water, in a 25 volume ratio of 1:5:2. After a layer of roughcast was applied: Portland cement 32.5N, 26 medium graded river sand and water, in a volume ratio of 1:4:2. Finally a render layer of 27 a cementitious mortar with a thickness of 1.5 cm was added on both faces of the masonry specimen, same composition as the mortar used in the joints. The composition and workability of the mortars were initially optimized, and then remained similar for all the assembled specimens. The main mechanical properties of the used materials were experimentally assessed, and the results obtained are summarized in Table 2.

5

## 6 Table 2 – Properties of the masonry components

#### Masonry components

Compressive strength of units, parallel to hollows, EN 772-1 [29] ***	6.55 (8%) N/mm <sup>2</sup>
Compressive strength of units, normal to hollows, EN 772-1 [29] *,**	2.21 (9%) N/mm <sup>2</sup>
Compressive strength, mortar joints and render, EN 1015-11 [30] $^{\star}$	8.58 (7%) N/mm <sup>2</sup>
Flexural strength, mortar joints and render, EN 1015-11 [30] $^{\star}$	2.11 (3%) N/mm <sup>2</sup>
Adhesion strength to render layer to brick, EN 1015-12 [31]*	0.51 (22%) N/mm <sup>2</sup>

7 \*The values in parenthesis represent the coefficient of variation.

8 <sup>\*\*</sup>Values obtained experimentally in dry air condition state.

9

10 A commercially available system, herein designated FRMCom, was used for the 11 strengthening of the masonry specimens. The FRMCom combines a carbon fibre mesh 12 (CFM) with a cementitious mortar matrix reinforced with polypropylene fibres (PFRM). 13 The polypropylene fibres have the main function of preventing cracking shrinkage of the 14 mortar, while the CFM aims to assure strengthening attributes to the FRMCom system. 15 The experimental results obtained from the characterization of the main physical and 16 mechanical properties of the PFRM in the hardened state are summarized in Table 3, 17 and the corresponding followed standards are indicated. The mechanical tests were 18 conducted in all cases 28 days after casting, in agreement with the EN 1015-11, [30]. 19 The flexural strength was obtained by averaging the results obtained in 3 specimens and 20 the compressive strength was obtained by averaging the results obtained in 6 21 specimens. All specimens were cured in laboratory ambient conditions, at an average 22 temperature of 19+/-2°C and a relative humidity of about 55% +/- 10%. The bending tests 23 were performed by applying a constant load increment of 35 N/s and the compression

- 1 tests were executed by applying an increasing force at a constant loading rate of 400
- 2 N/s until failure.
- 3
- 4 Table 3 Properties of the *PFRM*

2050 kg/m <sup>3</sup>
4% by weight
10000 N/mm <sup>2</sup>
4 (6%) N/mm²
8.27 (6%) N/mm²
.48 (20%) N/mm²
2

5 \* Values in parenthesis represent the coefficient of variation.

6

7 The adhesion strength was evaluated by means of pull-off tests conducted according the 8 EN 1015-12, [31]. The specimens with 30x20x13.5 cm<sup>3</sup> were built following the same 9 process used for the specimens of diagonal tensile tests and cured at constant 10 temperature and relative humidity of 20°C and 90%, respectively, [31]. The preparation 11 of the samples for testing implied the execution of a circular slot with a depth of 27 mm 12 and a diameter of 50 mm, using a core drilling machine and water for easier cutting and 13 for avoiding excessive vibrations. After cleaning the surface, a metallic plate was bonded 14 to the test area. The metallic plate was later attached to the pull-off machine, after the 15 initial levelling of the equipment, an increasing traction force was applied at a constant 16 loading rate of 40 N/s. The maximum force recorded corresponds to the adhesion force 17 that after has been divided by the area of the core providing the adhesion strength. The 18 pull-off tests were carried out 28 days after casting.

In Table 4 the properties of the CFM are presented, according to the data provided bythe supplier, [32].

21

1 Table 4 – Properties of the *CFM*, [32]

CFM	
Carbon fibres in both directions	50 threads/m
Elastic modulus	≥240 kN/mm²
Tensile strength	≥4300 N/mm²
Elongation at rupture	1.75 %
Ultimate tensile force	185 kN/m

#### 3 3.2. Preparation of the specimens for the diagonal tensile tests

The procedure adopted in the production of the masonry walls involved the following steps: soaking of the bricks until saturation; placement of guides to keep a constant joint thickness of 1.5 cm; levelling of the bedding mortar; placement and levelling of a row of bricks; placement of vertical mortar joints with a thickness of 1.5 cm; verification of the thickness of the joints; check the plumb; repeat the previous steps until the wall was finished. Six walls were produced for the diagonal tensile tests, three for reference specimens and three for subsequent strengthening.

All masonry specimens (a) were sprinkled with a cement mortar (1:4:2) two days after (b), then a cement mortar layer with a thickness of 1.5 cm (volume ratio 1:5:2) was applied on both faces (c), as shown in Figure 3a) and 3b).

14 The specimens were strengthened using the FRMCom system, as shown in Figure 3a) 15 and 3b), by carrying out the following steps: initial spraying of the wall with water; 16 application of the first layer of PFRM with an approximate thickness of 1.2 cm (d); 17 placement of the CFM on top of the PFRM layer (e); application of the second layer of 18 PFRM with an approximate thickness of 1.2 cm (f); use of a ruler and a trowel to level 19 and smoothen the mortar layer surfaces; spraying of the wall surfaces with water 15 min 20 after finished, to avoid shrinkage. The strengthening operations were performed 14 days 21 after the application of the render layer (c).



Figure 3 - General layout of specimens: a) Procedure adopted in the production of the
specimens; b) Section of the specimens and detail of the FRMCom system. (Legend: a
- masonry; b - roughcast mortar, mortar 1:4:2; c - render layer 1.5 cm thick, mortar 1:5:2;
d - PFRM 1.2 cm; e - CFM; f - PFRM 1.2 cm. Dimensions in cm)

#### 6 3.3. Set up used for the diagonal tensile tests

7 Diagonal tensile tests were performed to assess the contribution of the strengthening 8 system for increasing the load carrying capacity, the deformability and the energy 9 dissipation performance before failure of the masonry elements when subjected to a 10 loading scheme that resembles the in-plane shear loading conditions. The specimens 11 had a square geometry with approximately 990 mm side, as shown in Figure 4. The 12 number of clay blocks used in the vertical and horizontal directions was 5 and 3.3, and 13 the symmetry of blocks and joints both in the horizontal and in the vertical directions was 14 guaranteed. The thickness of the reference specimens was 140 mm and the thickness 15 of the strengthened specimens was 190 mm, as shown in Figure 3.



Figure 4 - Test set-up of the direct tensile test: a) Geometry of specimens and position
of LVDTs (dimensions in mm); b) Detail of the positioning of the specimen and of the test
set-up.

5 The set-up included a testing frame, an actuator with a 500kN load cell, and a servo-6 hydraulic closed loop controlled system, a data acquisition system and a monitoring 7 system composed by 5 linear variable displacement transducers (LVDT's). The vertical 8 and horizontal deformability of both specimen surfaces were determined in the central 9 area, between ¼ and ¾ of the diagonal of the specimen, by using 2 LVDT's in each 10 direction, as shown in Figure 4a).

The dimensions of the steel shoes adopted a "V" shape with 152 mm in each side and 320 mm long, according to ASTM E-519-2, [12]. In the case of the strengthened specimens, the local crushing and splitting of the external strengthening layer at both loading edges were observed, as observed for specimen FRMCom\_01 subsequently described. In order to avoid this premature local failure mechanism, in the subsequent specimens two steel plates were placed near the supports (150x150x30 mm<sup>3</sup>) to provide additional confinement to the material in this region. These plates were transversely
connected with 16 mm diameter steel rods crossing the specimen between the opposite
faces in each support, as shown in Figure 4b).

4 Each test was performed using displacement control of the actuator cross-head, by 5 measuring the displacement of the cross-head with an external LVDT. The applied 6 displacement rate was kept constant at 0.01 mm/s, for both monotonic and cyclic tests. 7 In the case of the cyclic tests, the displacement amplitude was gradually increased until 8 the last cycle was reached. The amplitude of the cycles was specified considering the 9 displacement of the actuator cross-head at the peak of the monotonic test, the increase 10 in each cycle was determined as 1/5 of that value. A maximum number of 7 cycles were 11 imposed, and an additional final loading cycle was imposed by applying a monotonically 12 increasing displacement until failure was reached (Figure 5). In the case of the 13 strengthened specimens, the experimental procedure was interrupted after the 7th cycle, 14 due to the loss of contact between the actuator and the loading shoe. The 8th cycle was 15 started after correcting this residual vertical deformation, by lowering 5 mm the reference 16 initial position of the actuator cross-head, as shown in Figure 5.





19

## 1 4. DIAGONAL TENSILE TEST RESULTS

The main results obtained for the monotonic diagonal tensile tests, which include the peak load and the corresponding horizontal and vertical average displacements, are presented in Table 5. The results obtained for the cyclic tests are presented in Table 6. For each cycle the peak load and the corresponding displacements, horizontal and vertical, are indicated in this table. For the specimens ref\_03 and FRMCom\_03 only the results up to the 6th and 7th cycles, respectively, are presented due to the occurrence of their premature failure at the referred cycles.

9

	ref_02			FRMCom_01	
Load	u <sub>h</sub>	$u_v$	Load	<i>u<sub>h</sub></i>	u <sub>v</sub>
(kN)	(mm)	(mm)	(kN)	(mm)	(mm)
97.34	0.12	-0.26	409.54	1.47	-0.68

10 Table 5 – Monotonic diagonal tensile test results.

11

12 Table 6 – Cyclic diagonal compression test results.

		ref_01			ref_03		FRI	MCom_	02	FRI	MCom_	03
	Load	$u_h$	$u_v$	Load	$u_h$	$u_v$	Load	$u_h$	$u_v$	Load	$u_h$	$u_v$
Cycle	(kN)	(mm)	(mm)	(kN)	(mm)	(mm)	(kN)	(mm)	(mm)	(kN)	(mm)	(mm)
Cycle	(1.1.1)	()	()	()	()	()	(	()	()	(	()	()
1st	14.51	0.00	-0.02	26.15	0.02	-0.04	115.68	0.02	-0.07	82.83	0.01	-0.05
2nd	42.50	0.01	-0.06	54.79	0.03	-0.09	219.93	0.06	-0.15	198.38	0.03	-0.12
3th	72.47	0.03	-0.11	84.66	0.05	-0.14	339.56	0.22	-0.29	310.40	0.11	-0.23
4th	100.48	0.04	-0.16	110.27	0.07	-0.18	401.35	0.78	-0.53	386.43	0.76	-0.44
5th	115.88	0.06	-0.19	136 54	0.10	-0.24	418 32	1 41	-0.66	430.81	1 37	-0.71
our	110.00	0.00	0.10	100.04	0.10	0.24	410.02	1.41	0.00	400.01	1.07	0.7 1
6th	98.44	0.13	-0.28	146.17	0.24	-0.32	383.04	1.90	-0.79	394.21	1.63	-0.74
7th	80.36	0.38	-0.32				312.35	2.59	-0.75	315.35	2.48	-0.46
8th	50.89	1.33	-0.42				298.16	3.07	-1.02			

13

The load vs displacement responses of the reference and strengthened specimens in terms of the averaged vertical and horizontal LVDT measurements are shown in Figure 6a) and Figure 6b). In general, the strengthened specimens have reached considerably

1 higher strengths than the reference specimens. The post-peak behaviour of the 2 reference specimens showed a smooth load decay, with the exception of specimen 3 ref\_03\_cic, which was supposed to undergo a cyclic loading sequence but failed 4 suddenly after reaching a considerably higher peak load than the remaining reference 5 specimens. The responses registered for the strengthened specimens were, in general, 6 similar, showing a pre-peak stage of significant load carrying capacity increase, and the 7 peak load was reached for substantially higher displacements than the ones observed 8 for the reference specimens. After peak load the strengthened specimens presented a 9 relatively smooth load decay with the increase of the horizontal deformation.



Figure 6 - Load vs displacement responses of the strengthened specimens: a) global
 response; b) detail of the response until peak.

The typical crack patterns observed at the surface of the reference and strengthened specimens after testing are presented in Figure 7a) and Figure 7b), respectively. The reference specimen presented a vertical crack that developed in a straight fashion from the lower to the upper support, very few and minor additional cracks presenting a very small tip opening developed after that main crack. The strengthened specimens presented, in a first phase, the same type of cracking as the reference specimens. However, at higher load levels multiple cracks developed and the failure was reached when the strengthening overlay started to detach from the masonry substrate (Figure 7c). This detachment occurred at the interface between the render layer (c in Figure 3) and the ceramic masonry brick surface, as shown in Figure 7c).



a)

b)

c)

Figure 7 - Image of the specimens after testing: a) Crack pattern in reference specimen,
ref-02 side B; b) Crack pattern in strengthened specimen, FRMCom-03 side B; c)
Detachment between the strengthening overlay and substrate, FRMCom-03.

9

## 10 5. DISCUSSION OF RESULTS

11

## 12 5.1. Peak values and capacity ratios

The responses presented in Figure 8 allow the clear distinguishing of the contribution of the strengthening system to the increase of the load carrying capacity of the tested masonry specimens. The shear stress and the shear strain were evaluated according to equation (1) and (3), respectively, where for *g* a value of 700 mm was considered. The shear strength of strengthened specimens was approximately two times higher than that of the reference walls. Considering the shear strain values obtained, the scatter of the results registered in the case of the reference specimens was greater, essentially due to the absence of the effect provided by the strengthening system for stabilising the fracture process. However, the strengthened specimens also showed a considerable scatter of results in terms of ultimate shear strain, in this case probably due to the influence of brittle failure modes such as the formation of the typical diagonal tensile crack and the detachment of the PFRM layers.

7 The maximum shear stress values obtained for cyclic testing were slightly higher than 8 the values obtained in monotonic tests. This is explained by the possible accommodation 9 of the loading shoes during the initial loading/ unloading cycles, providing a more 10 homogeneous diagonal load transmission to the specimen and therefore a slight 11 increase of load capacity.







14

The collapse of the reference specimen ref\_03\_cic was reached shortly after the peak load was reached, as soon as the diagonal crack was formed. In contrast, the other two reference specimens have reached considerably higher deformation levels and the collapse took place after a gradual load decay. In general, all reference specimens have

1 shown elastic responses almost up to the peak load. In contrast, the strengthened 2 specimens exhibited a substantially increased post-cracking load carrying capacity. The 3 non-linear portions of the responses were much more significant than in the case of the 4 reference specimens. The energy dissipation capacity was assessed for the monotonic 5 tests by computing the area under the load vs displacement response, where the 6 displacement represented the distance difference between the two opposite loaded 7 edges of the specimen during testing (see Figure 4). When comparing reference and 8 strengthened panels the increment of 1390% of the dissipated energy was obtained. 9 Considering that the monotonic load-displacement responses approximate well the 10 envelope of the cyclic tests, similar results of the dissipated energy increments are 11 expected for the cyclic tests.

12 The average values of the shear stress, the shear strain and the shear modulus are 13 computed in Table 7 and 8. The ratios between the limit values of the shear stresses and 14 the shear strains in the cases of the elastic and of the non-linear branches of the 15 responses observed in the strengthened and in the reference specimens were computed 16 as well. The elastic branch was considered to develop up to 33% of the peak load in the 17 case of the specimens tested with monotonic loading, and up to the peak of the first cycle 18 in the case of cyclic loading. These results show that the strengthening system has 19 provided a shear strength increase of approximately 2.3.

1 Table 7- Average of the limit values of the shear stress, shearing strain and shear

Response range	Type of specimen	Shear Stress <i>S</i> s (MPa)	Shearing Strain γ (%)	Modulus of shear stiffness <i>G</i> (MPa)
Electic bronch	Reference	0.185	0.007	2716.2
Elastic branch	Strengthened	0.532	0.012	4457.8
Poak	Reference	0.871	0.054	1696.1
I CAN	Strengthened	2.002	0.299	650.6

2 modulus obtained for the reference and the strengthened specimens.

3

4 Table 8– Increase of capacity in terms of shear stress, shearing strain and increment of

Response range	Shear Stress $S_s$	Shearing Strain $\gamma$	Modulus of shear stiffness <i>G</i>
Elastic branch	2.1	1.1	1.6
Peak	2.3	5.5	0.4

5 shear modulus obtained after strengthening.

6

7 The evolution of the damage due to the cyclic loading is shown in Table 9, where G<sub>cycle</sub> 8 represents the modulus of shear stiffness in each cycle, considering the linear part of the 9 curve shear stress vs. shear strain during the loading sequence, and G<sub>E</sub> represents the 10 modulus of shear stiffness in the elastic branch of the first cycle. The specimens 11 strengthened with the FRMCom system presented similar ratios G<sub>cycle</sub>/G<sub>E</sub> in the final 12 cycles. In the case of the reference specimens the damage occurred suddenly in the 13 case of the specimen ref 03, where the computed ratio was always 1.0 until failure. This 14 result indicates that the specimen failed in a brittle manner. In the case of ref\_01 the 15 response obtained showed a gradual decay of the G<sub>cycle</sub>/G<sub>E</sub> ratio, in a similar fashion to 16 the observed in the case of the strengthened specimens.

	re	f_01	rei	f_03	FRM	Com_02	FRMCom_03		
	Gcycle	ratio	Gcycle	ratio	G <sub>cycle</sub>	ratio	Gcycle	ratio	
Cycle	(MPa)	$G_{\text{cycle}}/G_{\text{E}}$	(MPa)	$G_{\text{cycle}}/G_{\text{E}}$	(MPa)	$G_{\text{cycle}}/G_{\text{E}}$	(MPa)	$G_{\text{cycle}}/G_{\text{E}}$	
Elastic	3326.6		2347.0		4427.2		4447.4		
1st	3326.6	1.0	2347.0	1.0	4427.2	1.0	4447.4	1.0	
2nd	2799.6	0.8	2297.7	1.0	4430.7	1.0	4467.3	1.0	
3rd	2706.9	0.8	2358.8	1.0	4210.0	1.0	4513.1	1.0	
4th	2526.8	0.8	2367.2	1.0	2627.8	0.6	3251.9	0.7	
5th	2477.7	0.7	2436.7	1.0	1379.2	0.3	1325.5	0.3	
6th	2083.7	0.6	2249.3	1.0	1044.5	0.2	975.7	0.2	
7th	1036.0	0.3			913.9	0.2	847.3	0.2	
8th	421.8	0.1			873.4	0.2			

### 1 Table 9 – Variation of the Shear modulus during cyclic tests.

2

3 The effectiveness of different strengthening systems and the scatter of results obtained 4 can be compared with the ones presented previously in Table 1. The mean values of the 5 maximum load and coeficient of variation obtained in this research for reference 6 specimens were 120 kN and 17%, which is expectable considering the brittle failure 7 mechanism obtained. Meanwhile, for strengthened specimes the mean values of the 8 maximum load and coeficient of variation obtained were 420 kN and 2%, representing a 9 increase in the capacity of 250%. The number of specimens tested, although limited, 10 agrees well with the number of specimens typically used by other authors for similar 11 experimental characterizations and the coeficient of variation obtained is also 12 comparable to the ones obtained by other researchers, as shown in Table 1.

13

14 5.1.1. Failure modes and crack patterns

During the diagonal tensile tests several types of mechanisms of structural degradationwere observed (see Figure 9):

1 Type A. Crushing of the masonry units next to the supports, toe crushing, 2 development of horizontal cracks, (ref\_01, FRMCom\_01). 3 Type B. Failure due to diagonal tensile stresses imposed during the tests, 4 development of vertical cracks in the central area of the specimens (ref\_01\_cic, 5 ref\_02, ref\_03\_cic). Type C. Delamination of the render and strengthening overlay from the masonry, 6 7 development of vertical cracks between the masonry and the rendering layers. Delamination starts in the two free corners of the specimen (FRMCom\_01, 8 9 FRMCom 02 cic, FRMCom 03 cic).

Type D. Cracking and failure of webs and shells of the ceramic bricks
(FRMCom\_01, FRMCom\_03\_cic).



Figure 9 – Schematics of the different mechanisms leading to structural degradation
during the diagonal tensile tests.

14

The failure of the specimens was reached after the development of one or more of these mechanisms. In the case of the reference specimens the mechanisms developed were mainly of the types B and D. In the case of strengthened specimens, a sequence of the mechanisms type B, C and D occurred before failure. The mechanism type A occurred only when the first strengthened specimen, FRMCom\_01, was tested. In the subsequent tests this failure mode did not occur, since the hollows of the brick in this zone were filled with a high compressive strength mortar, and metallic plates were applied to provideadditional confinement to the loading areas.

3 The instant at which the different mechanisms were initiated can be identified in the 4 Figure 10, where the letters identify the stage at which the respective type of mechanism 5 tends to occur. For both types of specimens, it was possible to trace the degradation of 6 the stiffness of the load vs displacement response. The stiffness of the load vs horizontal 7 displacement response is represented by the slope of the loading-unloading branches K<sub>d,h</sub>, while the stiffness of the load vs vertical displacement response is represented by 8 9 the slope of the loading-unloading branches,  $K_{d,v}$ . After the limit of the elastic branch was 10 reached (point b in Figure 10), both slopes, K<sub>d,h</sub> and K<sub>d,v</sub>, decrease faster for reference 11 specimens, indicating that a higher degradation was obtained in each cycle. For the 12 strengthened specimens the referred slopes show a lower variation per cycle, revealing 13 that the contribution of the strengthening solution to the shear capacity of the specimens 14 results in a reduction of the stiffness degradation. The variation of the stiffness of the 15 load vs vertical displacement responses, K<sub>d,h</sub> and K<sub>d,v</sub>, and of the ratios between the 16 stiffness of the first and of the consecutive cycles are presented in Table 10.





	ref_01				ref_03				FRMCom_02				FRMCom_03			
	K <sub>dv</sub>	ratio	$K_{dh}$	ratio	K <sub>dv</sub>	ratio	$K_{dh}$	ratio	K <sub>dv</sub>	ratio	$K_{dh}$	ratio	K <sub>dv</sub>	ratio	$K_{dh}$	ratio
Cycle	(kN/mm)	$\mathbf{K}_{dv}$	(kN/mm)	$K_{dh}$	(kN/mm)	$K_{dv}$	(kN/mm)	$K_{dh}$	(kN/mm)	$K_{dv}$	(kN/mm)	$\mathbf{K}_{dh}$	(kN/mm)	$K_{dv}$	(kN/mm)	$K_{dh}$
1st	-672	1.00	4266	1.00	-704	1.00	1714	1.00	-1671	1.00	6211	1.00	-1815	1.00	13252	1.00
2nd	-669	1.00	3118	0.73	-641	0.91	1876	1.09	-1620	0.97	5819	0.94	-1694	0.93	8146	0.61
3rd	-660	0.98	2787	0.65	-617	0.88	1841	1.07	-1431	0.86	3674	0.59	-1686	0.93	7242	0.55
4th	-637	0.95	2254	0.53	-615	0.87	1736	1.01	-1096	0.66	1261	0.20	-1353	0.75	2309	0.17
5th	-618	0.92	2081	0.49	-609	0.86	1605	0.94	-683	0.41	422	0.07	-816	0.45	439	0.03
6th	-413	0.62	1480	0.35	-555	0.79	1318	0.77	-557	0.33	229	0.04	-517	0.28	254	0.02
7th	-241	0.36	234	0.05					-426	0.26	120	0.02	-632	0.35	124	0.01
8th	-114	0.17	32	0.01					-345	0.21	98	0.02				

1 Table 10 – Evolution of the stiffness of the load vs horizontal displacement response from reference specimens under cyclic loading.

1 The crack patterns observed after the diagonal tensile tests are shown in Figure 11 and 2 Figure 12. In general, the failure modes for the unreinforced masonry reference 3 specimens were the typical ones expected for this type of test. In the case of the 4 strengthened specimens the crack patterns observed, and failure mechanisms obtained, 5 were influenced by the contribution of the strengthening system. The failure mode observed in this case was mostly characterized by a first phase where diagonal tensile 6 7 cracks were developing, followed by the delamination of the render layer plus 8 strengthening mortar, and a final stage of failure determined by the internal crushing of 9 the units. In the case of the FRMCom 02 and FRMCom 03 specimens several cracks 10 were formed, particularly next to the supports of the specimen, as shown in Figure 12. 11 The onset of the damage mechanism type D (Figure 9) is clearly revealed by the sudden 12 increase of the vertical displacement when compared to the horizontal ones (Figure 6), as identified in Figure 10 by the letter "d". The vertical displacement increments become, 13 14 at this stage, much larger than the horizontal ones due to the crushing of the shells and 15 webs of the ceramic bricks.

16



a) ref\_01\_side A

b) ref\_02\_side A

c) ref\_03\_side A

17 Figure 11 - Crack patterns of reference specimens after testing.



a) FRMCom\_01\_side A b) FRMCom\_02\_side A c) FRMCom\_03\_side A Figure 12 - Crack patterns of strengthened specimens after testing.

2

1

## 3 5.2. Analytical description of the experimental results

4 According to the theory of elasticity, as previously discussed in the introductory section 5 it can be assumed that during the diagonal tensile test the central area of a square 6 masonry specimen is approximately subjected to a stress state where  $\sigma_x = \sigma_y \cong$ -0.56P/A (compression) and  $\tau_{xy} = 1.05P/A$ . The principal stresses in this case are  $\sigma_I \cong$ 7 8 0.5P/A and  $\sigma_{II} \simeq -1.62P/A$ . The contribution of FRCM systems to the in-plane shear 9 capacity can be calculated following the analytical methodology presented in ACI 10 549.4R-13, [33]. According to this standard the nominal shear capacity,  $V_n$ , is computed 11 as shown in equation (5):

12

$$V_n = V_m + V_f \tag{5}$$

13 where  $V_m$  represents the contribution of the masonry wall, and  $V_f$  represents the 14 contribution of the FRM system.

The contribution of the masonry wall can be obtained according to the methodology presented by Li et al., [34], and applied to diagonal tensile tests by Babaeidarabad et al., [35]. Four types of failure mechanisms can occur, and the shear capacity for each type can be computed through the following equations:

Shear capacity due to shear sliding failure, V<sub>ss</sub>, calculated according to equation
(6):

$$V_{ss} = \frac{\tau_0}{1 - \mu_0 \times \tan\theta} \times A_n \tag{6}$$

where  $\tau_0$  is the shear bond strength of the mortar joint (MPa),  $\mu_0$  is the coefficient of internal shear friction in mortar joints,  $\theta$  is the angle between horizontal and main diagonal of the wall (degrees), and  $A_n$  is the net area of the cross section of the masonry wall (mm<sup>2</sup>).

6

Shear capacity due to shear friction failure, V<sub>sf</sub>, calculated according to equation
(7):

$$V_{sf} = \frac{\tau_{0,m}}{1 - \mu_m \times \tan \theta} \times A_n \tag{7a}$$

$$\tau_{0,m} = \frac{1}{1 + 1.5 \times \mu_0 \times \frac{h}{w}} \tag{7b}$$

$$\mu_m = \frac{\mu_0}{1 + 1.5 \times \mu_0 \times \frac{h}{w}} \tag{7c}$$

9

10 where  $\tau_{0,m}$  is the modified shear bond strength in mortar joint (MPa), with *h* and *w* being 11 the height and length of the bricks (mm), respectively, and  $\mu_m$  is the modified coefficient 12 of internal shear friction in mortar joint.

Shear capacity due to the diagonal tension failure, V<sub>dt</sub>, calculated according to
equation (8):

$$V_{dt} = \frac{\tan\theta + \sqrt{21.16 + \tan\theta^2}}{10.58} \times f'_t \times A_n \tag{8}$$

15

16 where  $f'_t$  is the tensile strength of masonry,  $f'_t = 0.67 \times \sqrt[2]{f'_m}$ ,(MPa), with  $f'_m$  being 17 the compressive strength of the masonry (MPa)

Shear capacity due to toe crushing failure at the loading end, V<sub>c</sub>, calculated
 according to equation (9):

$$V_c = \frac{2 \times w \times f'_m}{3 \times h + 2 \times w \times \tan \theta} \times A_m \tag{9}$$

20

21 where  $A_m$  (mm<sup>2</sup>) is de interface loading area between the steel shoe and the wall.

1 Finally the shear capacity of the unreinforced or plain wall,  $V_m$ , is the minimum value of 2 previous expressions, see equation (10):

$$V_m = \min\{V_{ss}; V_{sf}; V_{dt}; V_c\}$$
(10)

3

$$V_m = \min\{V_{ss}; V_{sf}; V_{dt}; V_c\}$$
(10)

4 The behaviour of a near surface mounted FRP bars (NSM) based strengthening system 5 is presented by Li et al., [34]. The developed theoretical formulation takes into account 6 two different mechanisms, which are the debonding of the bar and the failure of the bar 7 due to the tensile stresses. In contrast the FRMCom system uses a carbon mesh layer 8 to resist the tensile stresses, and in this case only the failure of the fibres by tensile 9 stresses is considered. The contribution of the FRM system for the shear capacity,  $V_f$ , is 10 obtained according to equation (11):

$$V_f = 2 \times n \times A_f \times L \times f_{fv} \tag{11}$$

12 where n is the number of fabric layers,  $A_f$  is the area of mesh reinforcement by unit width (mm<sup>2</sup>/mm), L is the length of the wall in the direction of the shear force (mm),  $f_{fv}$  is the 13 design tensile strength of FRM shear reinforcement,  $f_{fv} = E_f \times \varepsilon_{fv}$ , (MPa),  $E_f$  is the 14 tensile modulus of elasticity of the cracked FRM specimen, (MPa), and  $\varepsilon_{fv}$  is the design 15 16 tensile strain of FRM shear reinforcement (mm/mm).

17 The shear capacity of the masonry specimens was calculated according to the analytical 18 model presented previously using the values from Table 11. The compressive strength 19 of the masonry was estimated according to EC6, since it was not possible to assess 20 these parameters experimentally. However, previous studies conducted by Pereira et. al 21 [36] indicate that these estimations, although slightly conservative, approximate well the 22 experimental results. The characterization of the initial shear strength and the shear 23 friction angle of mortar joints was carried by Capozucca [36], the values obtained are 24 used in the analytical model as a lower bound value. The tensile behaviour of the FRCM 25 composite system utilized was not experimentally characterized at this stage, since the 26 failure is expected to be governed by the failure of the fibre mesh reinforcement. The

- 1 mechanical characteristics of the FRCM system single components (matrix and fibre
- 2 mesh) were previously presented in Table 3 and 4.
- 3
- 4 Table 11 Values used in the analytical model to obtain the shear capacity of the
- 5 specimens.

Description:	Value
Shear bond strength of the mortar joint, according to Capouzzuca [37]	$\tau_0 = 0.31 \text{ MPa}$
Coefficient of internal shear friction in mortar joint according to Capouzzuca [37]	$\mu_0=0.60$
Angle between horizontal and main diagonal of the wall	$\theta = 45$ °
Net area of the masonry wall cross section	$A_n = 116600 \text{ mm}^2$
Height of the brick units	h = 195  mm
Length of brick units	<i>w</i> = 285 mm
Compressive strength of the masonry (according to EC6 [13] )	<i>f'</i> <sub><i>m</i></sub> = 1.14 MPa
Interface loading area between the steel shoe and the wall specimen	$A_m = 33440 \text{mm}^2$
Number of fabric layers	n = 2
Area of mesh reinforcement by unit width, CFM [32]	$A_f = 0.044 \text{ mm}^2$
Tensile modulus of elasticity of the cracked FRM specimen [32]	$E_f = 240000 \text{ MPa}$
Design value of the tensile strain of FRM reinforcement (according to ACI 549.R-13 [33])	$\varepsilon_{fv} = 0.004 \text{ mm/mm}$
Ultimate tensile strain of FRM reinforcement according to the supplier [32]	$\varepsilon_{fv,ult} = 0.0175 \text{ mm/mm}$
Tensile strength of the render mortar	$f_{t,r} = 2.11 \text{ MPa}$
Net area of the render cross section	$A_{n,r} = 31800 \text{ mm}^2$
Tensile strength of the PFRM	$f_{t,Ac} = 3.27 \text{ MPa}$
Net area of the PFRM cross section	$A_{n,Ac} = 53000 \text{ mm}^2$

7 The contribution of the rendering and PFRM layers, considering diagonal tensile 8 failure,  $V_r$  and  $V_{PFRM}$ , are obtained according to equation (8) by using the net area of the 9 cross section and the tensile strength of the respective materials. 10 Considering the equations (6) to (10) the result is a shear capacity of the unreinforced 11 masonry wall,  $V_m = 33.2 \ kN$ , and of the rendering layers,  $V_r = 33.8 \ kN$ , which have to be 1 summed in order to obtain the total capacity of the masonry specimen before 2 strengthening,  $V_{ref} = 67.0 \ kN$ .

3 With regards to the capacity of the strengthening system, according ACI 549.4R-13 the 4 maximum allowable extension of the fibre reinforcement is 0.004 mm/mm. Considering 5 this value and the equation (11), the maximum shear capacity of the strengthening 6 system is  $V_f = 118.3 \ kN$ . On the other hand, when considering the ultimate extension of 7 the reinforcing fibres provided by the supplier,  $\varepsilon_{fv,ult} = 0.0175$  mm/mm, [32], the 8 maximum shear capacity is  $V_{f,ult} = 517.5 kN$ . Finally, considering the diagonal tensile 9 failure according to equation (8), the shear capacity of the PFRM layers is  $V_{PFRM}$  = 10 87.3 kN.

11 The shear capacity values obtained experimentally are compared to the ones determined 12 analytically in Table 10. Different experimental/analytical shear capacity ratios were 13 computed, by following different sets of assumptions. For the reference case, the 14 maximum shear capacity was estimated considering the contribution of the masonry,  $V_m$ , 15 and rendering layer,  $V_r$ . For the "ACI" case, following ACI 549.4R-13, [33], the maximum 16 shear capacity,  $V_{ACI}$ , was estimated considering the contribution of the masonry,  $V_m$ , of 17 the rendering layer,  $V_r$ , and of the carbon strengthening mesh,  $V_f$ , by assuming 0.004 for 18 the maximum allowable extension of the carbon fibres, as prescribed by ACI 549.4R-13, 19 [33]. For the "ACI+PFRM" case, the maximum shear capacity, V<sub>ACI+PFRM</sub>, was estimated 20 considering the contribution of all previous factors and the additional contribution of the 21 PFRM layers,  $V_{PFRM}$ . Finally for the "Sup" case,  $V_{Sup}$ , it was consider the contribution of 22 all the elements and the maximum allowable extension of the carbon fibres prescribed 23 by the supplier,  $V_{f,ult}$ .

While comparing the results obtained analytically to the experimental ones, see Table 12, it is possible to confirm that the procedure proposed by ACI 549.4R-13, [33], leads to conservative results. This procedure leads to a safety factor of 2.3, which may be considered somewhat high considering the low scatter of the experimental results

1 obtained. This value compares with a safety factor of 1.33 proposed by ACI 549.4R-13, 2 [33] for similar types of strengthening systems. However one may consider that the 3 strengthening technique is still, to a certain extent, novel and therefore still encompasses 4 significant uncertainties, most regarding durability and time dependent behaviour. When 5 the contribution of the layers of the strengthening overlay is added to the analytical 6 estimation, V<sub>ACI+PFRM</sub>, the experimental vs analytical ratio decreases to 1.5. On the other 7 hand, it is also possible to observe that, when considering the ultimate extension of the 8 fibres, V<sup>an</sup><sub>Sup</sub>, a ratio of 0.6 is obtained, meaning that the failure of the masonry model 9 occurred before the ultimate extension in the fibres was reached. This was confirmed 10 during the experimental program, since the failure was determined by the delamination 11 of the strengthening system and crushing of the masonry, and not by the failure of the carbon mesh. Finally, if the contribution of the analytically estimated values of  $V_m$  and  $V_r$ 12 13 is replaced by the experimentally obtained mean value of the diagonal tensile strength 14 results in the reference specimens, the safety factors 1.8, 1.3 and 0.6 are obtained for  $V_{ACI}^{an}$  ,  $V_{ACI+PFRM}^{an}$  and  $V_{Sup}^{an}$ . 15

- 16
- 17

•	Table 12 – S	Shear capa	icity obta	ined for	reference	and stren	gthened	specimens.
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		Reference	ce Strengthened		
Experimental (kN)		<i>V</i> <sup><i>exp</i></sup> 119.8		<i>V<sub>str</sub></i> 419.5	
Analytical		$V_{ref}^{an}$	V <sup>an</sup> <sub>ACI</sub>	$V_{ACI+PFRM}^{an}$	V <sup>an</sup> Sup
(kN)	$V_m = 33.2$	+	+	+	+
	$V_r = 33.8$	+	+	+	+
	$V_f = 118.3$		+	+	
	$V_{f,ult} = 517.5$				+
	$V_{PFRM} = 87.3$			+	+
		67.0	185.3	272.6	671.8
	Ratio: Experimental/Analytical	1.8	2.3	1.5	0.6

#### 1 6. CONCLUSIONS

2 The potentialities of a commercial system formed by a mortar reinforced with carbon fibre 3 mesh (FRMCom system) for the strengthening of masonry walls were assessed in the 4 present work by carrying out in-plane diagonal tensile tests. This strengthening system 5 provided an increase of about 2.3 times of the shear strength of the reference specimens. 6 The test procedure adopted allowed to evaluate the evolution of the damage in the 7 specimens while subjected to cyclic loading, which was in general similar for all 8 strengthened specimens. In the case of reference specimens, the damage developed in 9 a more sudden manner.

10 In general, the failure modes of the reference specimens were the typical ones and 11 expected for diagonal tensile testing. In contrast, the failure modes and crack patterns 12 obtained for the strengthened specimens were characterized by a first phase at which 13 the normal diagonal tensile cracks were developing gradually, followed by the 14 delamination of the strengthening mortar immediately before failure. Additionally, in the 15 case of the strengthened specimens several cracks were formed, particularly next to the 16 supports of the specimen.

17 The adoption of a cyclic loading procedure on the diagonal tensile test, for the 18 experimental characterization of the masonry, allowed the assessment of the stiffness 19 degradation during each cycle, as well as the deterioration of the strengthening 20 mechanisms. Additionally, it was possible to verify the adequacy of this test to assess 21 the relevant in-plane failure mechanisms for typical masonry walls.

Although the experimental programme was limited the dispersion of the results allowed to assess and quantify the contribution by the strengthening system to the increase of the load carrying capacity with a reasonable accuracy. The FRMCom system strengthening technique provided a significant increment of strength and energy dissipation ability to this type of construction system, as well as higher shear strain combined with lower scatter of the obtained results. The delamination of the strengthening overlay may suggest that, in some cases, the adoption of transverse

connecting systems between the masonry opposite faces may become essential to fully
 explore the contribution of the strengthening overlay to the load carrying capacity
 increment.

4 The analytical model described by ACI 549.4R-13 led to results that can be considered 5 conservative, with a safety factor of 2.3. This is in part explained by the neglecting of the 6 contribution of the layers of the strengthening mortar and by the limitation of the tensile 7 strain of the carbon fibres to a value of 0.004 mm/mm. When the contribution of the 8 strengthening mortar is considered, the safety factor decreases to 1.5. Finally, when the 9 carbon fibres are allowed to reach the ultimate tensile strain of 0.0175 mm/mm, a safety 10 factor of 0.6 is obtained indicating the premature failure of the strengthening system. This 11 result was corroborated by the experimental results, where the premature failure of the 12 strengthened specimens was observed before the ultimate tensile strain was reached on 13 the carbon fibre reinforcement.

14

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