

Inconel 718–Copper bimetallic joints fabricated by 3D multi-material laser powder bed fusion for aerospace components

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ABSTRACT

Usually, aerospace components are subjected to high demanding operating conditions in terms of mechanical and thermal stresses. The structural integrity of these components is assured by the use of high strength and temperature-resistant materials such as Inconel alloys. However, these alloys are known to have low thermal conductivity, which makes it difficult to extract heat from inside the component. Copper-based alloys are widely used in the aerospace field due to their high thermal conductivity, high strength, good ductility and corrosion resistance.

A 3D multi-material Inconel 718 – Copper solution was produced by laser powder bed fusion to be applied on a rocket engine wall aiming to improve its heat extraction ability. This approach combines high strength Inconel 718 and high thermal conductivity Copper in a single part, produced at once. The individual Inconel 718 and Copper zones and interface transition zone features were assessed in terms of metallurgical bonding and mechanical behaviour. This 3D multi-material Inconel 718 – Copper solution seems to be a promising approach since the two materials have a well-defined interface with no substantial defects. Inconel 718 and Copper seems to be capable to maintain its most important individual properties, high strength and high thermal conductivity.

INTRODUCTION

Additive Manufacturing (AM) processes are a group of technologies that allows to produce high quality metal parts. Material extrusion, binder or material jetting, sheet lamination, directed energy deposition (DED) and powder bed fusion (PBF) are the most common AM processes reported in literature [1]–[3][4].

PBF technologies includes spreading of powder bed and the use a heat source to melt atomized powder particle together layer by layer until the part is completed [5]. The most common PBF technologies are Electron Beam Melting (EBM) and Laser Powder Bed Fusion (LPBF) and differ from each other in the heat source type – there are heated thermal print head, electron beam and laser, respectively [6]–[8].

LPBF technology have been reported as a promising process to obtain highly densified parts (with densification close to 100%) with satisfactory mechanical properties when compared with cast parts and close to those produced by forging. Contrarily to conventional manufacturing, this technology have a huge design freedom that allows to produce parts with complex geometries [8]–[10].

This additive manufacturing technology are deeply explored on single materials such as titanium, stainless steel or nickel alloys. However, the production of multi-material parts has several advantages over single material parts since the joining of two or more materials offers allows to benefit of a larger range of properties - such as thermal, mechanical, electrical, optical, and corrosion resistance - of the materials included [11], [12].

The development of 3D multi-material solution (material varies along the xy-axis while is built in the z-axis) by laser powder bed fusion has been rarely proposed since the powder fusion technology was first conceived. Some achievements and complications are not yet explored in 1D or 2D multi-material solutions, especially the replacing of the bed twice for each layer [16], [17]. Multi-material laser powder bed fusion technology (MMLPF) has been applied on the production of functionally graded materials (FGM) usually known as having arbitrary changes in composition to regulate final properties of a part but only presents material's changes along x-axis [13]–[15]. Significant challenges for multiple material fabrication including grading in both the vertical and horizontal directions, clearing of unwanted powder compositions from previous layers, local processing parameter changes, grading of materials with disparate melting temperature\ and demonstration of complex geometries with graded composition.

Copper alloys are usually employed in main combustion chamber in the regeneratively cooled rocket engines due to its high thermal conductivity. However, these ductile alloys are also known as having low elevated temperature strength [18]–[21]. The joining of two distinct alloys is very useful to benefit from different properties. Some works reported the production of multi-material IN718-Cu solutions through conventional manufacturing processes proved difficult, including formation of microcracks and splat boundaries. Marques et al [22] access both individual material zones and interface transition zone features in terms of metallurgical bonding and mechanical behaviour fabricated by a 3D multi-material laser powder bed fusion system. Onuiké et al [23] measure shear strength of Inconel 718/copper alloy (GRCop-84) bimetallic joint built via laser engineered net shaping (LENS™). Onuiké et al [24] studied the interfacial and thermal properties of Inconel 718 - GRCop-84 bimetallic structures were produced by laser engineering net shaping (LENS™).

This work presented the diffusion behaviour and interfacial characteristics of a 3D multi-material solution processed by laser powder bed fusion in which each layer is constituted by two distinct materials, Inconel 718 (IN718) and Copper (Cu). The joining of these two distinct materials is a significant challenge but also a promise in the improvement of heat extraction ability of an aerospace component. Additionally, a multi-material solution constituted by these two materials joined together can also be interesting for application in collection coins.

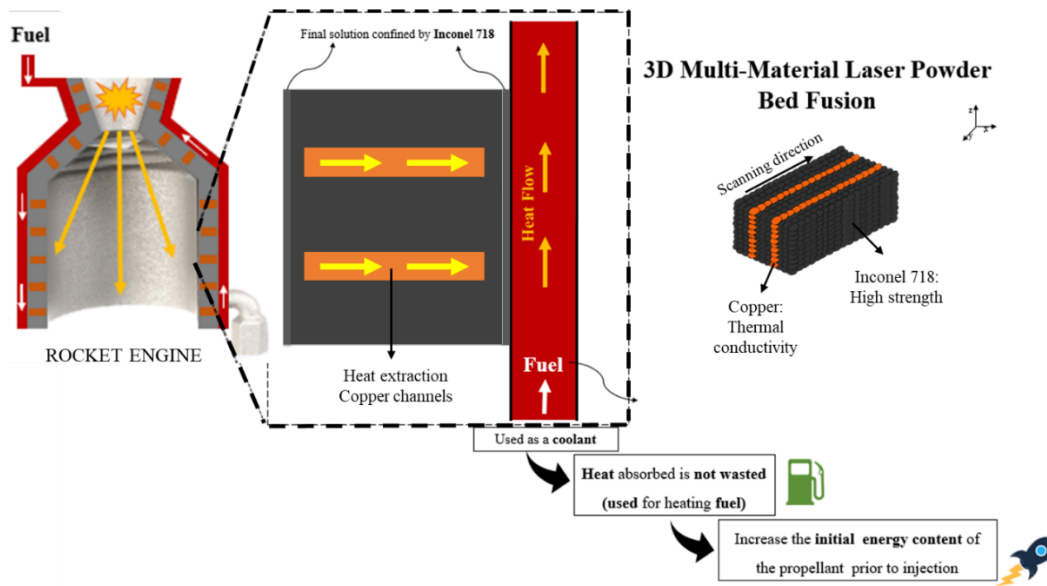


Figure 1. Design concept of the 3D multi-material IN718 – Cu solution for a rocket engine.

MATERIALS AND METHODS

Materials

IN718 and Cu powders used in this study were purchased from Carpenter Additive and TLS Technik LPW Technology suppliers, respectively, with a given particle size ranging between 15 and 45 μm . The Cu powder with 99.7% purity presented a spherical shape (Figure 2a) and IN718 powders showed some satellites (Figure 2b).

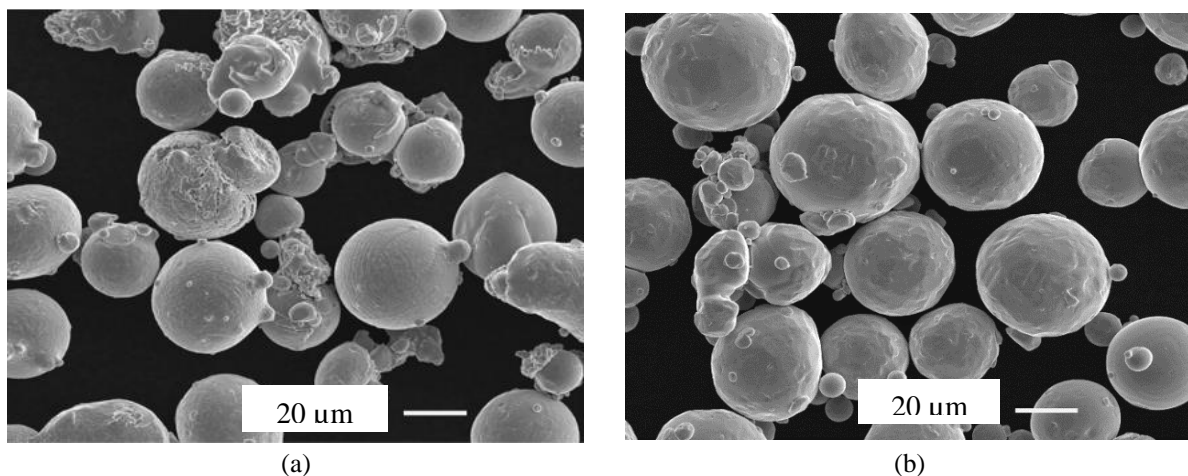


Figure 2. (a) IN718 and (b) Cu powder's morphology.

The IN718 alloys constituted by 50.0 – 55.0 % Ni and 21.0 % Cr, 2.8 – 3.3 % that is responsible to prevent oxidation at high temperatures and of molybdenum (Mo) that avoid the pitting corrosion. This alloy have also as titanium, aluminum, cobalt and iron [25][26] (Table 1).

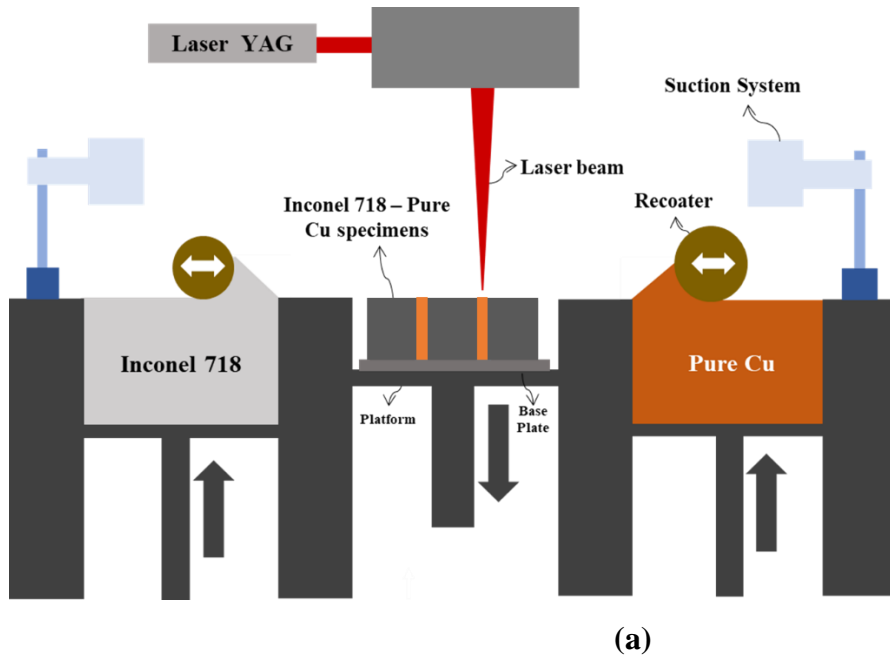
Table 1. Chemical composition of IN718 powder (Carpenter Additive Lda) [27].

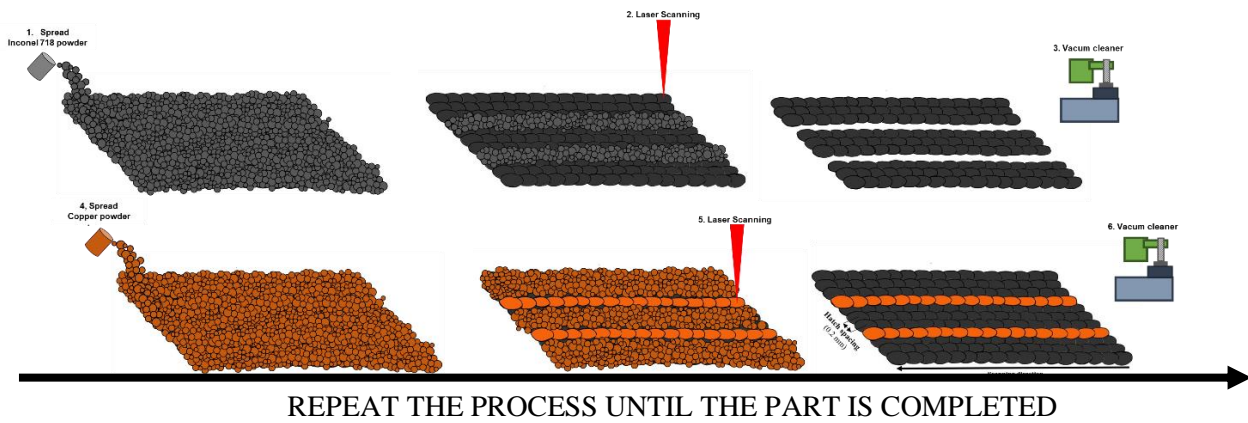
Elements	Al	Cr	Co	Cu	Fe	Mn	Mo	Ni	Nb+Ta	Si	Ti
wt. %	0.70	21	1	0.3	Balance	0.35	2.80-3.30	50-55	4.75-5.50	0.15	1.15

Equipment

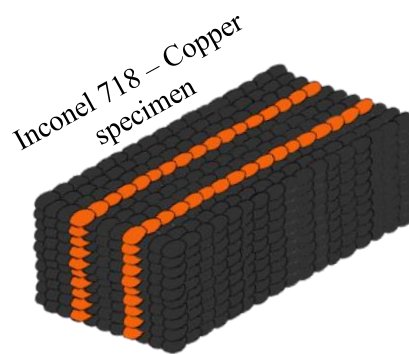
The 3D Multi-material IN718 – Cu specimens was produced using home-made Multi-Material Laser Powder Bed Fusion system (MMLPBF) developed at Center for Microelectromechanical Systems (CMEMS) at University of Minho which is equipped with an Nd: YAG laser as the heat source (maximum power of 80 W and laser wavelength (λ) of 1064 nm) represented in Figure 3a.

The 3DMMLPBF system includes two simultaneous and independent powder deposits (IN718 and Cu), two redesigned recoaters which are used to deposit two feedstock materials alternately and two vacuum suction systems to clean and replace the powder materials (Figure 3b). Both IN718 and Cu feedstock powders were deposited and processed on an IN718 build plate. The 3D multi-material specimens (Figure 3c) were produced under an argon atmosphere (gas pressure of 2.5 bar) in order to control the level of oxygen once this later is one of the main problem of processing copper alloys by this technology [28]. The IN718 and Cu powder were alternately deposited into the platform, allowing to fabricate 3D multi-material parts, layer-by-layer and in each layer can have different materials.





(b)



(c)

Figure 3. (a) Multi-Material LPBF system equipped with two powder deposits, for Inconel 718 and Pure Copper, two suction systems and laser YAG (b) Main steps on the production of a single layer (c) Inconel 718 – Copper multi-material specimen.

The 3D multi-material specimen was designed with fifteen IN718 single scan tracks intercepted by two of Cu and each single track is separated by $200\ \mu\text{m}$ (scan spacing). The processing parameters are presented Table 2. The power laser for Cu (12 W) was higher than for IN718 powder (10 W), given that the Cu has a significantly higher thermal conductivity ($\sim 400\ \text{W/mK}$ at room temperature [29]) than IN718 ($\sim 11\ \text{W/mK}$) [30]. As reported by Wessel et al [12], if the processing parameters were optimized for Cu, the confined liquid melt pool temperatures of the IN718 would probably be higher than the suitable for processing.

Table 2. LPBF process parameters for the alloys and their resultant relative densities.

Processing Parameters	Inconel 718	Pure Copper
Laser Power (W)	10	10 – 15
Laser scan speed (mm/s)	120	120
Layer thickness (μm)	30	30
Hatch spacing (mm)	0.2	0.2
Remelting	No	No
Uni-directional length (mm)	4	4
Laser energy density (mm/s)	0.15	0.08 – 0.13

Characterization Techniques

Both melt pool and polished surfaces (in top and cross section sides) were analyzed through an analytical scanning electron microscope (SEM) JEOL JSM-6010LV) equipped with Electron Dispersive X-ray Spectroscopy (EDS) for further assessment of the chemical composition at the IN718, interface and Cu zones. X-ray diffraction (XRD) analysis in the interface IN718 – Cu specimens was executed by using a Bruker D8 Discover diffractometer with a classical θ - 2θ analysis (Bragg-Brentano geometry utilizing $\text{CuK}\alpha$ radiation) in which were considered a 2θ ranging between 30° and 100° , a 4 s/step of step-time and 0.04° of step-size.

The Vickers' hardness results were assessed through a microindenter equipment from EmCoTest – DuraScan using a diamond indenter. The measurements were performed considering three load values (100, 50 and 10 g) due to ductility discrepancy between IN718 and Cu and also based on literature methodology [31], [32].

RESULTS AND DISCUSSION

Morphological characterization

Processing parameters optimization

The metallurgical bonding of IN718 and Cu in 3DMMLPBF technology was achieved through a local fusion of the powders and later consolidation (Figure 4). The IN718 processing parameters were previously optimized, and a laser power of 15 W was fixed as optimum for obtaining a consolidated IN718 part. Since literature have reported undesirable results on pure Cu solidification [33], [34], this study performed an optimization of the processing parameters for sintering Cu powder. The Cu powder was deposited and melted over the IN718 processed powder. Following this, the interface region was

melted by using the same laser power as for Cu (10, 12, 14, 14.5 and 15 W). The use of low amount of energy (10 W) proved to be insufficient to reach high densification levels of Cu. A high number of pores with irregular shapes (with tens of micrometres) were observed on top surface due to lack of fusion. The increase on the laser power to 12 W led to an increase in the number of pores found on the Cu surface, but the Cu line is still not well defined geometrically. However, when applied at 14 W laser power, the porosity was significantly reduced and the Cu line is an approximated continuous line. As the laser power, in terms of densification, seems to be close to the optimum values, the laser power was increased 0.5 W to 14, 14.5 and 15 W. The use of 15 W of laser power led a well-defined Cu continuous line without significant number of pores.

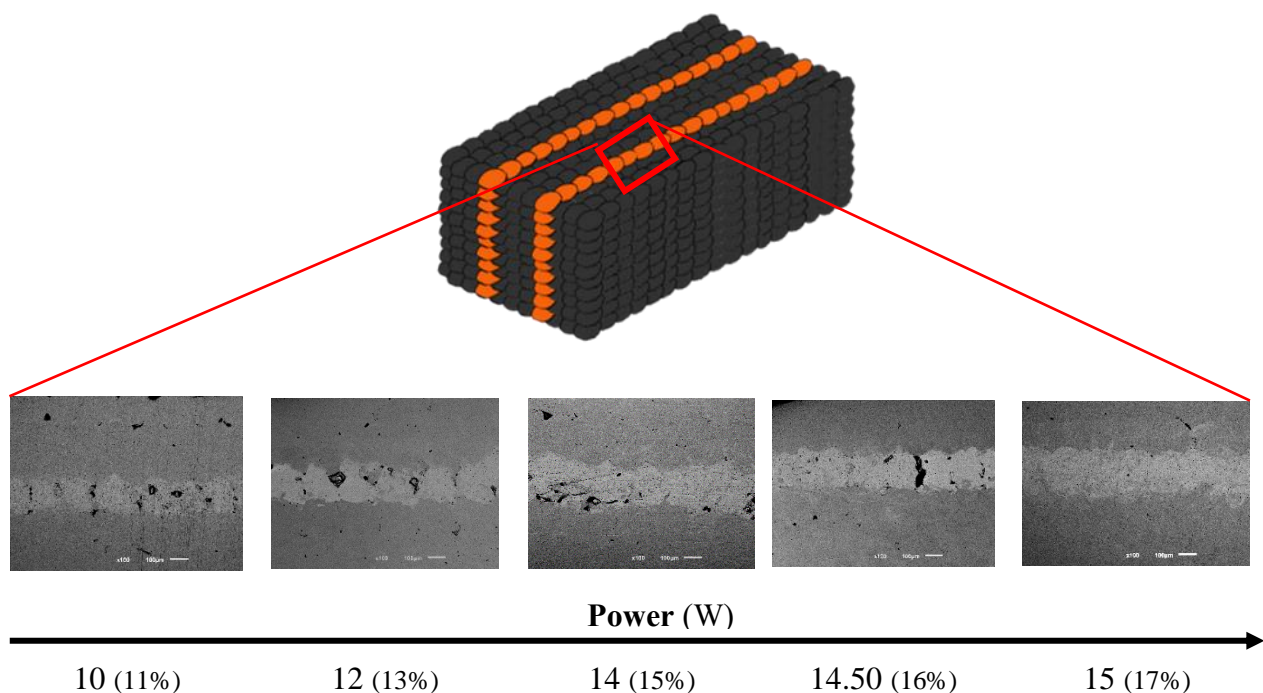


Figure 4. Top surface SEM images and EDS line across a Cu line.

The main balance on the optimization of the processing parameters for sintering Cu was to obtain a continuous and well-defined Cu line, low porosity and non-significant diffusion at the interface. It is known that the interfacial diffusion is relevant for a proper metallurgical bonding between two materials. However, the objective of this multi-material solution is to guarantee the individual properties of each material, allocated in distinct regions of the same part. In this sense, after found a laser power range (14 and 15 W) in which the Cu line are well-defined with low porosity levels, an EDS line scans from IN718 to Cu with 170 and 230 μm . Ni, Cr, Fe and Cu were the main chemical elements discussed since IN718 alloy has a limit of 0.1% wt Cu and about 50 – 55 % wt Ni [35] and the Cu, from pure Cu, having 99.7% wt Cu and 0% wt Ni. Figure 5 shows that an abrupt transition at low laser power values (12 and 14 W) and a Cu region with almost 100% of copper which means that a non-significant diffusion region are

observed at the interface. At 14.5 W, the IN718 region showed an increase on Cu element (20 – 40 % wt) and the Cu region presented some valleys coincident with peaks of Ni, Cr and Fe. The diffusion of two materials at 15 W is more evident with %wt Cu in Copper region ranging between 60 and 80 %.

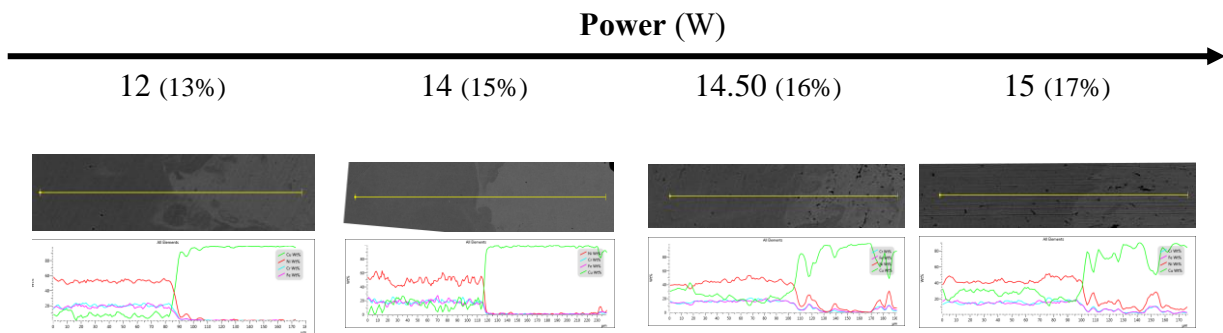


Figure 5. SEM images and EDS line across IN718 and Cu region.

Interfacial Characterization

After adjusting 3DMMLPBF processing parameters (to raise the laser energy density), the Cu powder has been reasonably solidified and bonded to the IN718 melted powder as well IN718 substrate (with a porosity of about 0.356 ± 0.05 %). Although the processing of Cu is still a challenge, the processing parameters used in the present study allowed to obtain well-defined Cu heat conductive channels with approximately 200 μm width. Residual irregular pores (with tens of micrometres) were found in Cu melted powder (as can be observed in Figure 6). Morphological characterization of the top and cross surface of the 3D multi-material IN718 – Cu specimens.) due to lack of fusion. Circular pores observed in Cu melted powder were attributed to the high cooling rate of the 3DMMLPBF technology since the gases have not time to escape, being entrapped into the solidified metal.

The Cu powder was deposited and melted over the IN718 processed powder to melt the interface region by using the laser power value of 12 W. The use of higher energy density for processing the interface leads to undergo of the IN718 processed powder and mixing with the as-deposited copper. Wits et al. [12] have reported the development of graded structures in LPBF and considered that it is advantageous to stretch the diffusion zone, creating a gradual material transition. The cracks formation also can be originated by the amount of copper at the interface during the processing process. Liu et al [36] studied the metallurgical diffusion between 316L stainless steel and C18400 copper alloy and reported that as higher the amount of copper the interface higher the possibility of creation dilute solution with iron, which strongly improves the stress at the fusion zone and so the crack formation

The IN718 inner zone is almost completely consolidated and free of defects as it is visible at the top and cross-section of the SEM images (Figure 4). Some occasionally circular shaped pores are still perceptible, which may have originated by gas entrapping during the processing [37]. These results

allowed us to conclude that the applied laser energy density was enough to melt the IN718 material, without lack of fusion.

The use of high energy the processing parameters used in the present study allowed to obtain well-defined Cu heat conductive channels with approximately 200 μm width. Circular pores observed in Cu melted powder were attributed to the high cooling rate of the 3DMMLPBF technology since the gases have not time to escape, being entrapped into the solidified metal.

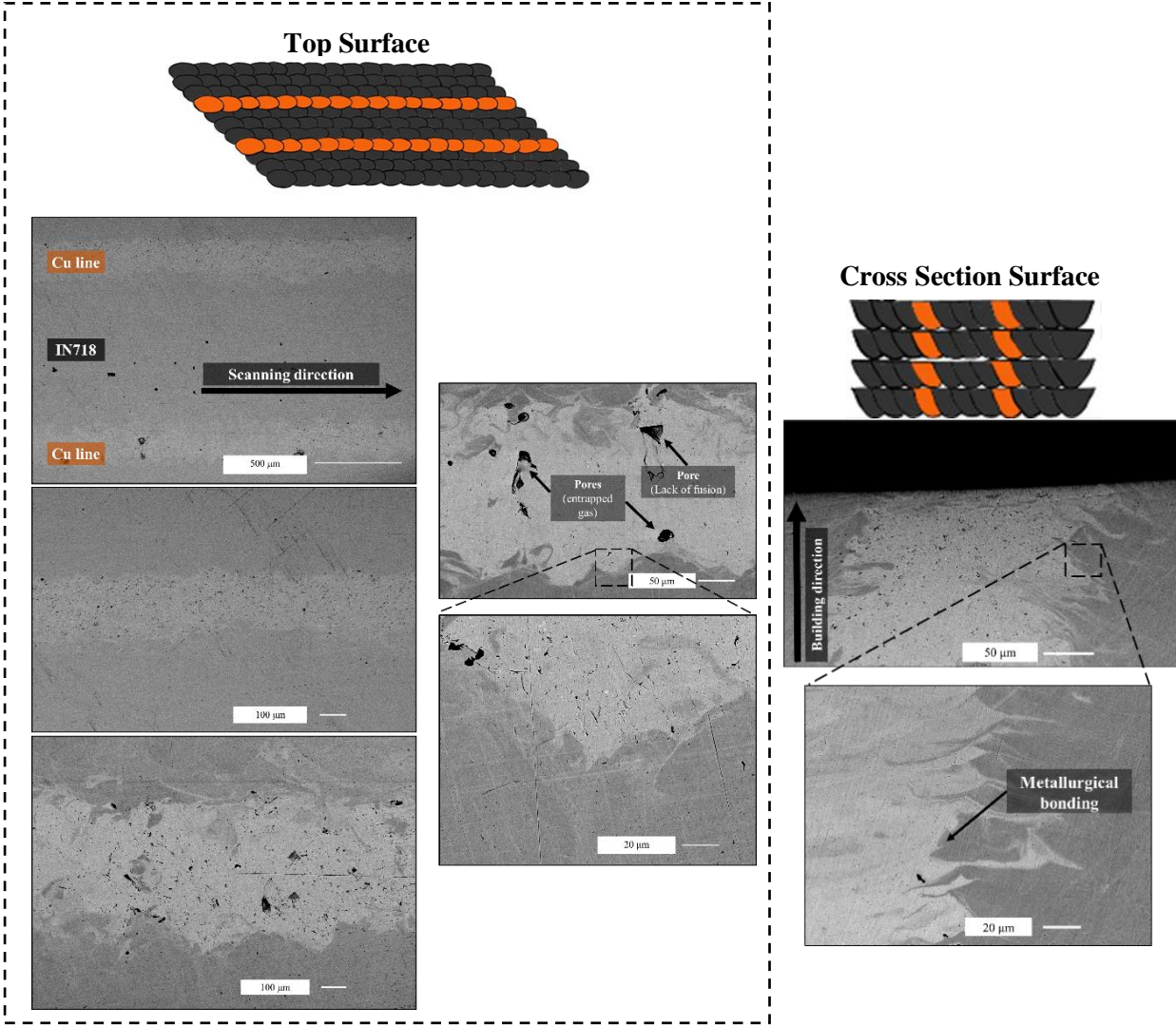


Figure 6. Morphological characterization of the top and cross surface of the 3D multi-material IN718 – Cu specimens.

EDS elemental spectra of Ni, Fe and Cr (main chemical elements of IN718) and Cu are shown in **Error! Reference source not found.** The IN718 is mainly made of Ni (represented in red), with approximately 50 - 60 % and the Cu zone wt.% is quite high ranging from 80 to 100 (represented in orange). The interface region is represented by a gradual transition between IN718 and Cu with a significant amount of Ni and Cu element diffusion (about 25 μm thickness) at the interface region suggested a metallurgical bonding (confirmed in). Small jumps in Ni and Cu detection can be observed at 25 μm and 75 μm

distance, suggesting that there is an insignificant intermixed powders zone close to the interface may be due to an incomplete suction of the powders.

These results indicate strong evidence that this new 3DMMLPBF concept is successful mainly due the presence of a well processed Cu line. The high cooling right of the this technology indicate a fast solidification, avoiding the generation of large diffusion areas in the interface zones and thus, preventing the existence of fragile zones and undesirable intermetallic phases, as reported elsewhere [13].

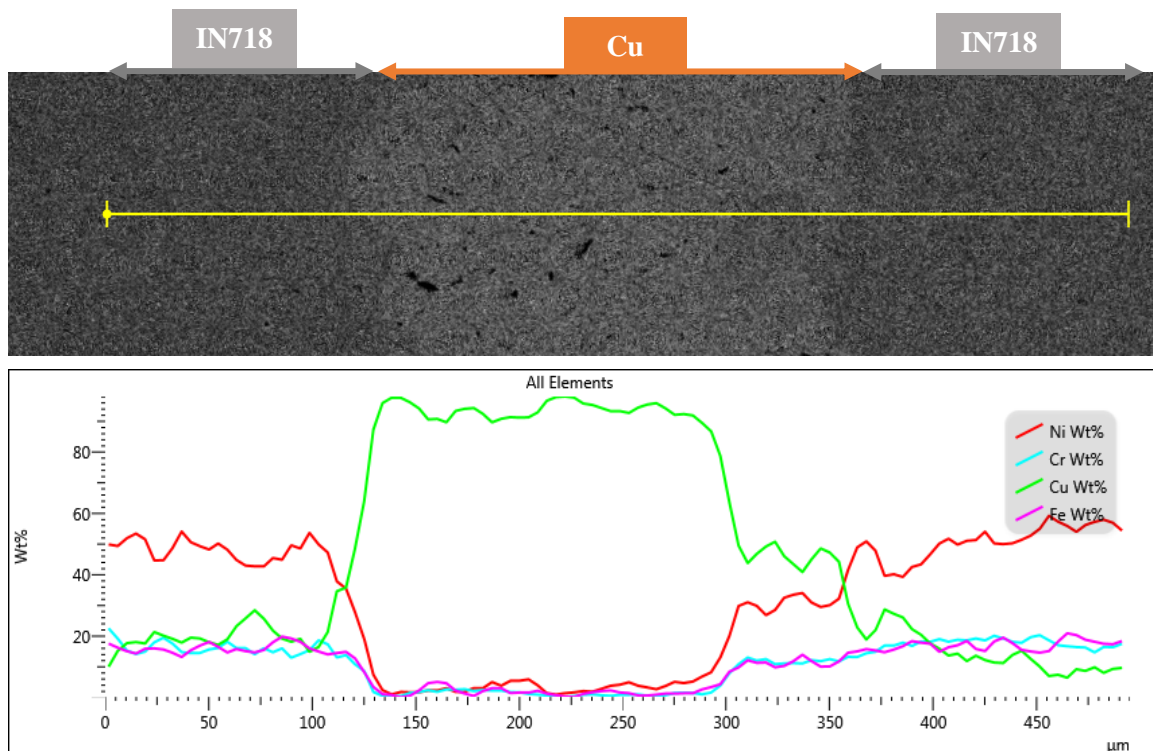


Figure 7. EDS mapping line crossing IN718, interface and Cu zones.

The XRD pattern is coincidental to the solid solution of austenite (γ) constituted by a single phased structure with Ni and Fe (the most important elements of IN718) [38]. Figure 8 showed the XRD patterns at the interface IN718 – Cu with peaks of IN718 and Cu at different angles of diffraction that are overlaid with each other. This indicate that only a single-phase solid-solution were formatted at the interface as can be confirmed in the binary equilibrium phase diagram Ni – Cu. Therefore, the absence of two-phase solid solution is beneficial since it avoids the formation of brittle intermetallic at the interface. On the other hand, the formation of a single phase at the interface induces a suitable metallurgical bonding as well high strength at the interface.

There should be underlined that the alignment of XRD peaks from different materials (IN718 and Cu), observed in Figure 8, indicate the interfacial zone carried the features of both IN718 and Cu single-phase solid solutions.

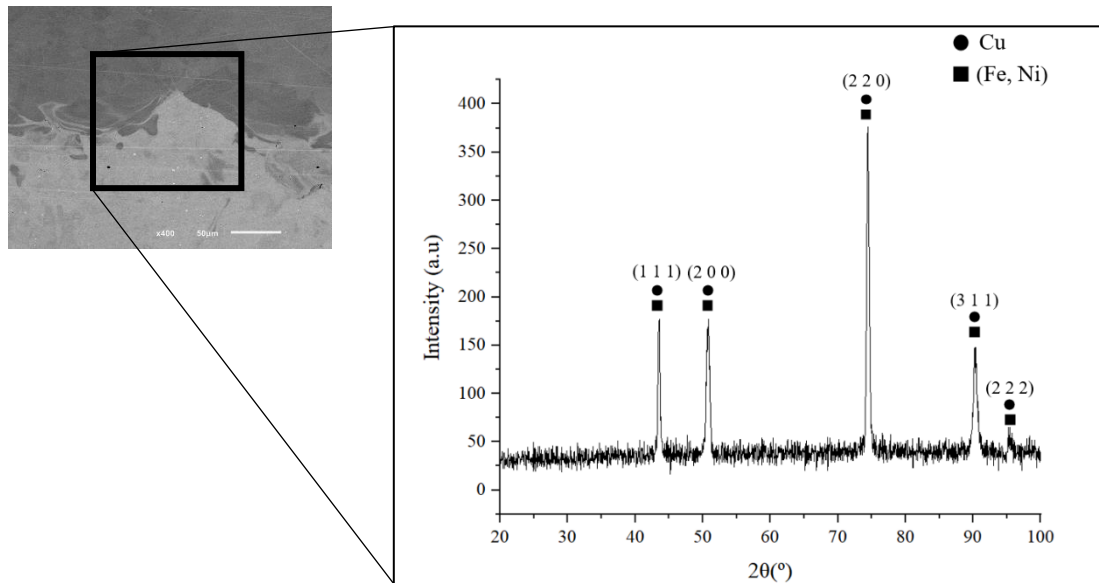


Figure 8. XRD analysis across IN718 – Cu interface.

It is known that performing hardness tests can provide relevant inputs regarding the plastic deformation of a material. Moreover, the resistance to wear (by friction or erosion by steam, oil or water) of aerospace components under harsh environments can also be indicated by hardness results [39].

Hardness Characterization

Vickers' microhardness was measured considering three load values (100, 50 and 10 g) since the two materials exhibit a very different ductility. Moreover, these measurements were taken in the left, middle and right side of the specimen in order to study how the strategy influences the hardness results along the specimen (see Figure 9). Regarding IN718 zone, the hardness measurements considered a load of 100 and 50 g and the results proved that there was no significant hardness discrepancy between distinct regions. Moreover, the hardness range values (from 292 ± 11 to 344 ± 12 HV) are coherent with those reported in literature concerning the as-built LPBFed IN718 specimens (275 - 350 HV) [40]–[63]. In agreement with the multi-material IN718 – Cu solution chemical composition (which transit from 100% IN718 to almost 100% Cu), the hardness values jumped from 344 to 126 HV respectively.

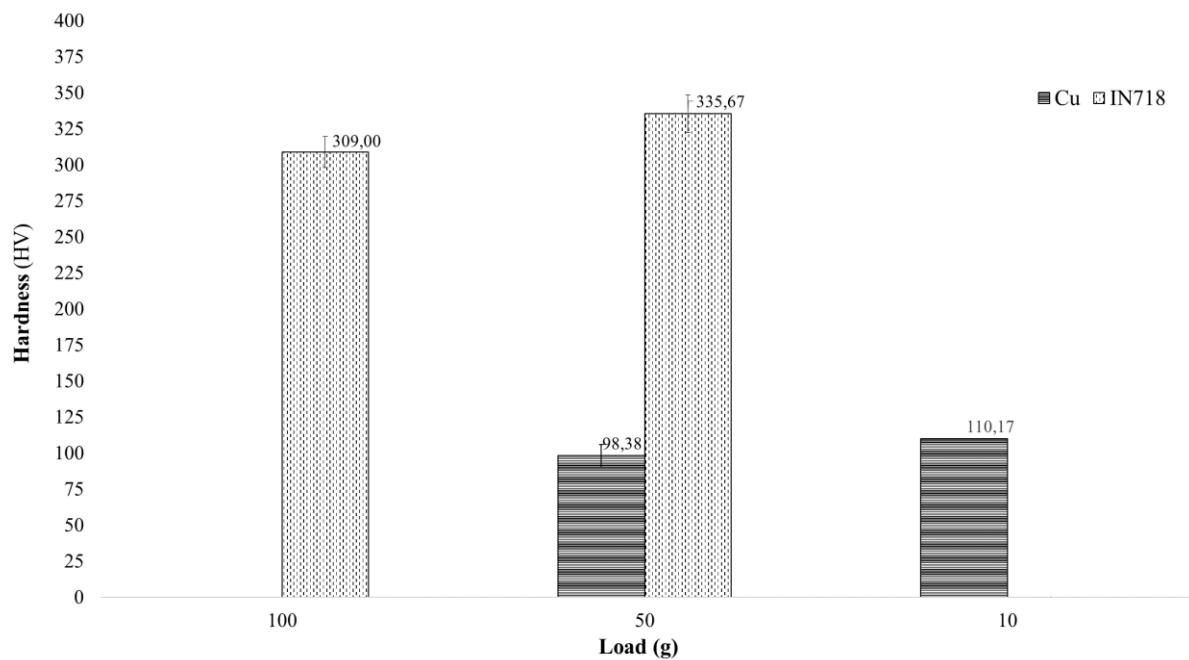


Figure 9. Vickers's microhardness results considering 100, 50 and 10 g at IN718 and Cu zones.

CONCLUSIONS

This first approach on producing 3D multi-material IN718 - Cu specimens by Laser Powder Bed Fusion here presented was considered successful. The IN718 zones are almost free of defects and show hardness values ranging the values reported in literature. On the other hand, the Cu lines present a well-defined geometry with about 200 μm width and occasional defects. A thin diffusion zone of about 25 μm is present in the transition between IN718 and Cu. In this sense, the samples produced by this new 3DMMLPBF system presented a metallurgical bonding at the most critical zone (interface) as well as a desirable metallurgical bonding between these two distinct materials. Also, there are no evidence of the formation of intermetallic compounds. This is the first work indicating the hardness values for Cu powder processed by LPBF (~126 HV). Future works should focus on studying the metallurgical features, mechanical and thermal analysis under rocket engines operation conditions.

FUNDING RESOURCES

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