# Influence of fatigue and aggressive exposure on GFRP girder to SFRSCC deck all-adhesive connection

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#### Abstract

To assess the influence of fatigue loading and environmental conditions on the bond behavior between glass fiber reinforced polymer (GFRP) systems and steel fiber reinforced self-compacting concrete (SFRSCC) that are adhesively bonded, an experimental program composed of push-out tests was carried out. The following three scenarios were selected for the environmental conditions: natural conditions; wet-dry cycles; and temperature cycles. Half of the specimens were submitted to monotonic loading up to failure, and the other half were submitted to a fatigue load configuration of 1-million cycles and then subjected to a monotonic loading up to failure. The results have shown that for the investigated environmental conditions the GFRP-SFRSCC push-out specimens never failed up to 1 million cycles. However, temperature cycles caused a considerable reduction on the stiffness and load carrying capacity in the specimens submitted to fatigue loading, while wet-dry cycles did not modify significantly the maximum shear stress transfer in the investigated connection. This paper describes in detail the experimental program, presenting and discussing the relevant results.

# Keywords

Pedestrian bridge; Composites; GFRP pultruded profiles; Fiber reinforced self-compacting Concrete; Composite action; Aggressive exposure

#### 1 Introduction

There is always a need for innovative and durable structural load-bearing systems to accelerate construction, especially in bridges. The application of Fiber Reinforced Polymer (FRP) materials is gaining a strong momentum in structural engineering applications [1], particularly in bridges, which are typically exposed to harsh environments. Pultruded FRP shapes have the potential to replace steel sections for bridges and buildings [2]. For this purpose, Glass Fiber Reinforced Polymer (GFRP) is being dominantly used rather than other types of FRP materials, such as carbon or aramid FRPs, because of its reasonable cost. However the uses of composites in bridge construction are still not that common, and a majority of built bridges must consider to be prototypes, such as Hayes *et* al. [3], Keller [4] and Mendes *et* al. [5].

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When compared to traditional materials such as steel, concrete or timber, FRP materials present several benefits [6]: mainly superior corrosion resistance, stiffness-to-weight and strength-to-weight ratios, low thermal expansion, non-magnetic properties, damage tolerance and ease of transportation and handling.

FRP materials present higher unit price when compared with conventional materials. However, due to the lower costs of transportation, installation, maintenance, and life cycle, constructive systems based on FRP materials become very competitive [7]. Due to superior durability of FRP materials, in many cases composite structures can last much longer than the ones made of conventional materials, yielding lower life-cycle cost [8].

The research carried out in this paper is part of a three-year research project funded by ADI (Portuguese Innovation Agency) to develop permanent pedestrian bridge systems using Steel Fiber Reinforced Self-Compacting Concrete (SFRSCC) decks and Glass Fiber Reinforced Polymer (GFRP) girders (see Fig. 1). During the first phase of the project, the creep, as well as the structural behavior of the pedestrian bridge was investigated [5]. Two different connections configurations between the deck and the girder were evaluated. The former using steel anchors, glued to the concrete deck by an epoxy adhesive, then bolted to the GFRP profiles, with the deck surfaces also glued to the GFRP girders by an epoxy adhesive (see Fig. 2). The latter using adhesive alone to glue the deck and the girder. Numerical simulations were carried out to appraise the possibility of using an all-adhesive connection. The results showed that for the serviceability limit states and for the ultimate limit states, the load carrying capacity of the bridge structure is not affected by using an all-adhesive connection between the deck and the girders. However, it was mentioned that further investigation should be carried out to appraise the response of the connection under extreme conditions.

Therefore, the present research focuses on the adhesively bonded deck-to-girder connection behavior, under static and fatigue loading, when exposed to temperature and wetdry cycles.

# 2 Experimental program

## 2.1 Material properties

Table 1 shows the mixture composition of the developed SFRSCC, and Table 2 presents the relevant rheological and mechanical properties of this material [9, 10], respectively. 60 kg/m³ of hooked ends steel fibers of 35 mm length, 0.55 mm diameter and 1100 MPa tensile strength were used. The methodology followed to formulate the SFRSCC composition is mainly based on the following three steps [9]: (i) the proportions of the constituent materials of the binder paste are defined; (ii) the proportions of each aggregate on the final granular skeleton are determined; (iii) binder paste and granular skeleton are mixed in distinct proportions until self-compacting requirements in terms of spread ability, correct flow velocity, filling ability, blockage and segregation resistance are assured, allowing the determination of the optimum paste content in concrete.

The GFRP profiles were produced by the pultrusion process and are composed by 65% in volume of E-glass fibers embedded in a polyester matrix. The fiber systems is composed of 97% of E-glass roving type and 3% of E-glass continuous filament mat (CFM). The production speed of these GFRP profiles is 0.3 m/minute at a temperature and humidity of 180 °C and 55% in the die, respectively. The transition temperature of the polyester matrix is 87 °C. From an extensive experimental program for the characterization of the relevant

properties of the type of GFRP profile used in the present research, the values indicated in Table 3 were determined [12]. The "S&P Resin 220" epoxy adhesive used to glue the GFRP profiles to the SFRSCC panels was applied cured at room temperature. According to the supplier this epoxy adhesive presents a bending tensile strength of 30 MPa, a compressive strength of 90 MPa and bulk shear strength of 3 MPa. By executing tensile tests according to the ISO 527-2:1993 standard [13] Sena *et* al. [14] obtained for the average tensile strength and the average Young's modulus the value of 21 MPa and 7.7 GPa, respectively. The glass transition temperature (Tg) of this adhesive is 55 °C [11]. The long-term properties of this epoxy adhesive were assessed experimentally by Costa and Barros [15].

# 2.2 Test specimens, setup and instrumentation

Fig. 3 shows the push-out test specimens used in the present research. The specimens consist of two concrete panels (400×250×40 mm³) adhesively bonded to each flange of the GFRP profile. The GFRP profiles were bonded to the SFRSCC panels when the SFRSCC age was 21 days. The cross-section of the GFRP profile is a I-shape with 200 mm height, 100 mm wide and a constant thickness of 10 mm. The adhesive volume in the bond between the flange of the GFRP and the SFRSCC panel is 350×100×2.0 mm³. A needle scaler was used to ensure that the concrete laitance was removed before bonding the GFRP profiles to concrete. Then, a spatula trowel was used to apply the adhesive on the concrete surface. The surfaces of the GFRP profiles were treated by using sandpaper to improve the adherence between epoxy adhesive and GFRP profile. After the fabrication process, the specimens were left in a room at an ambient temperature of 22 °C for 7 days, allowing a complete cure of the adhesive.

To ensure minimum friction between the concrete panels and the supporting steel frame (see Fig. 4), rectangular plates of neoprene were used. Additionally, to prevent any suddenly failure of the specimen, a passive confinement system was used, as shown in Fig. 4.

A total of 18 specimens were tested. The specimens were divided into three series according to the type of exposure used: reference exposure, temperature cycles, and wet-dry cycles. The first character in the specimen ID indicates the series number (R for reference, T for temperature cycles and WD for wet-dry cycles), the middle character specifies the loading type, such as S for static, F for fatigue and PF for post-fatigue, and the last character gives the specimen number. As an example, R\_S1 stands for the reference specimen number one submitted to static loading.

The push-out specimens were tested with a vertical pushing load, as shown in Fig. 4. The monotonic load and the fatigue load were applied using a 500 kN servo-hydraulic actuator incorporating a load cell of 500 kN.

To measure the slip between the flanges of the GFRP profile and SFRSCC, five LVDTs (LVDT 1 to 5) with a stroke of 5 mm were placed with a regular pattern in four different locations along the bond length (see Fig. 3). The supports used to hold these LVDTs are perfectly aligned in the horizontal plan. Two LVDTs of 10 mm stroke were used (LVDT 6 and 7) to measure the relative horizontal movement of the concrete panels (see Fig. 3).

## 2.3 Exposure environments

The connection system proposed in the present paper is intended to be used in a full-scale composite pedestrian bridge to be installed in the north of the Portuguese territory. Thus, to ensure a long service live and guarantee its proper functioning, with minimal maintenance,

is essential to estimate the long-term behavior of such connection under different environmental conditions. Therefore, the exposure conditions chosen for this research were based on the characteristics of the actual environment to which the bridge structure will be exposed.

The north of the Portuguese territory features the Mediterranean climate, with warm, dry summers and mild rainy winters. Summers are typically sunny with average temperatures between 14 °C and 28 °C, but can rise to as high as 40 °C during occasional heat waves. During such heat waves the humidity can reach 95 %; however, during summer, the average relative humidity is 80 %. Winter temperatures typically range between 4 °C during morning and 14 °C in the afternoon but rarely drop below 0 °C at night [16].

Based on such conditions, two different types of exposure were considered for the present research: alternate wet-dry cycles and temperature variation cycles. In addition, one group of specimens was kept in the laboratory as control specimens. Details of specimens tested under various exposure regimes are shown in Table 4. Both exposure conditions were applied after the epoxy adhesive has been cured during at least 7 days.

In the alternate wet-dry exposure, each cycle consisted of complete immersion of the specimens in 3 % NaCl solution for 12 hours followed by drying in air another 12 hours. The NaCl solution was used to simulate the worst-case scenario, where the structure is located near the shore. For such cycles two separate containers were used, each one equipped with a water pump programed to pump the water for one container to another in each 12 hours. The water was kept at 20 °C using a thermostat during immersion, and the air was kept approximately at 25 °C during the drying period using a thermo ventilator. Two small pumps were installed in each container to circle the water, avoiding the sedimentation of the salt as well as the water impurities.

A total of 100 cycles were repeated with the described conditions. The amount of salt in the water was decided from the stipulated guidelines provided by ASTM B 117-85 for salt spray (fog) testing. The pH of the solution was found between 7.5 and 8.0.

In the temperature cycle exposure, a climatic chamber was used with temperature and humidity regulation capability. A typical cycle of temperature variation with time used in the tests is shown in Fig. 5, i.e. six cycles a day with maximum and minimum temperatures of 60 °C and -10 °C, respectively. The specimens were exposed to a total of 100 cycles, with a RH of 80 % during positive temperatures, and approximately 10 % during negative temperatures.

#### 2.4 Test procedures

To develop an all-adhesive connection for use in pedestrian bridge systems, it is essential to analyze the bond behavior under different exposure and loading conditions. Thus, three different load scenarios were used: static load, fatigue load and post-fatigue static load.

The static load allows the determination of the bond-slip relationship, as well as the ultimate bond strength. The fatigue loading allows the assessment of the connection behavior at long-term. And lastly, the post-fatigue static load allowed the evaluation of the changes in the stiffness and bonding capabilities after fatigue loading.

The static load and the post-fatigue static load tests were carried under displacement control mode. The static load was applied monotonically until failure. The displacement rate 0.2 mm/min was adopted to control the test with the internal LVDT of the actuator. The

fatigue load was determined based on previous tests and numerical simulations of the bridge structure [5], which allowed the determination of the maximum shear stress in the GFRP-concrete connection for the serviceability limit states. Therefore, to determine the load to obtain the equivalent shear stress, one reference specimen was monitored with strain gauges under static load. The strain gauges were positioned at the level of a horizontal plane, one crossing LVDT 1, and the other passing LVDT 5, and were bonded to the concrete and to the flange's GFRP profile. The obtained results allowed the determination of the fatigue load range. Hence, the fatigue tests were carried out at maximum and minimum stress ratios of  $S_{max}(F_{max}/F_{ult}) = 0.48$  and  $S_{min}(F_{min}/F_{ult}) = 0.14$ . All tests ran at a load ratio R  $(F_{min}/F_{max}) = 0.28$ . Therefore, the specimens were firstly loaded statically up to 90 kN, and then the fatigue load was applied between 40 kN and 140 kN, corresponding to a variation of  $\pm 50$  kN. All specimens were subjected to 1 million sinusoidal load cycles at a frequency of 2 Hz, and the sinusoidal loading was force controlled. The cycle readings were taken at the following rate: every 1000 cycles 10 cycles were recorded. Note that in the subsequent sections the results showed are for the representative push-out specimen from the three specimens tested.

## 3 Test results and discussion

#### 3.1 Monotonic tests

Table 5 shows the results and statistical data for the monotonic and post-fatigue monotonic behavior series. Note that in the subsequent sections the analysis of the results is carried out in specimens assumed to be representative of the push-out tests of the three environmental conditions. Thus, for the monotonic series the R\_S2, T\_S3 and WD\_S1 were adopted, while for the post fatigue series the R\_PF3, T\_PF1 and WD\_PF3 were selected for this purpose.

Fig. 6 shows the typical load versus loaded end slip (LVDT1, Fig. 3) curves for the three different environmental exposures (see also Table 5). The control exposure (R series) exhibited the highest peak load (at the failure of the specimen), reaching approximately 315 kN, with a loaded end slip of 69  $\mu$ m. The results from temperature cycles exposure (T series) reveal that, when compared to the ones of the control exposure, the peak load and the loaded end slip decreased significantly, reaching 183 kN and 42  $\mu$ m, respectively, representing a reduction of 41 % in terms of load carrying capacity. This behavior was a consequence of the stress cycles imposed by the temperature cycles. Also, the temperature cycles exceeded the  $T_g$  (55 °C), which, according to Moussa et al. [17], has a negative impact on the bond strength and stiffness.

Considering the wet-dry cycles (WD series), the results show an increase of the specimen stiffness up to a load of about 150 kN, then the stiffness remained identical to the control exposure up to a load of about 250 kN, followed by a decrease in the bond stiffness. The peak load also decreased slightly, reaching 296 kN with a loaded end slip of 48 µm, corresponding to a reduction of approximately 6 % and 30 %, respectively, when compared to the control exposure. The increase of the initial stiffness for the WD series could be explained by the less aggressive and best curing conditions provided by the WD exposure, i.e., the temperature ranged between 20 °C to 25 °C, which lead to a gradual curing of the adhesive. Due to this stiffer behavior of the connection system, it is natural that when damage initiates, its effect is more pronounced, not only in terms of stiffness but also in terms of ultimate load, due to a more abrupt release of the energy accumulated in the constituent materials and bond systems, which justifies the behavior of the last phase of the WD series.

Fig. 7 presents the failure modes of the specimens after the monotonic tests. The failure occurred only in one of the bonded flanges of the GFRP profile. Although all specimens failed by debonding, the concrete layer peeled off varied with the type of exposure, and thus affecting both the bond stiffness and strength. It is noticeable that at the loaded-end, in the T and WD series, small portions of adhesive remained bonded to the concrete, meaning that cohesive failure occurred. This small portions of cohesive failure denoted that, under such environmental conditions, the adhesive tends to be the weakest component of the system.

In the control specimens the debonding was uniform and the concrete layer peeled off was thicker than the other series (Fig. 7a). This figure also shows the presence of some aggregates in the concrete layer peeled off, denoting the stiffness of the connection.

# 3.2 Fatigue behavior

In Fig. 8, the loaded end slip *versus* number of cycles at both the maximum and the minimum load levels are plotted for each exposure condition. The shape of the curves is generally similar among the different exposure conditions. Initially, a significant increase of slip is observed up to the 100,000th cycle, which is then followed by a mild growth period where the slip increased slowly until the last cycle. This clearly indicates the continuous degradation of the bond interface during the fatigue load cycles.

Fig. 9 evidences that the amplitude of the loaded-end slip was almost constant during the fatigue loading process, which indicates that the bond connection between GFRP profile and SFRSCC plates was essentially governed by the elastic properties of the adhesive. In the R\_F, T\_F and WD\_F series this amplitude was 14 to 16.5 µm, 26.5 to 30 µm and 16.5 to 20 µm, respectively. Therefore, the larger amplitude of loaded-end slip was registered in the T\_F series (86% higher than in R series), which is a consequence of the decrease of the elasticity modulus of the adhesive caused by the exposure conditions that in this series includes a temperature higher than the Tg of the adhesive. The amplitude of the loaded-slip was 20% higher in the WD\_F series than in R\_F series, which was not expected taking into account the stiffest response of the WD series registered in Fig. 6. The relatively small difference between these two series in terms of loaded-end slip amplitude (it becomes closest during the fatigue loading process) can be justified by the difficulties of assuring equal geometric and material properties for the applied adhesive layer.

# 3.3 Post-fatigue monotonic behavior

After the fatigue tests, all specimens were loaded monotonically up to failure, being the results presented in Fig. 10 and Table 5. Due to the fatigue loading effect, the specimens showed a significant increase in the bond stiffness up to a load range between 50 and 100 kN. This phenomenon can be explained by the densest and more compact connection resulted from the fatigue load, promoting a more homogeneous microstructure of all the system (SFRSCC, GFRP and adhesive). Subsequently the specimen starts to reach inexperienced load levels (fatigue loading between 40 and 140 kN), causing a reduction in the bond stiffness, which remains identical to the results from the monotonic tests.

Thus, the control exposure (R series) reached a peak load of approximately 310 kN, with a loaded end slip of 32 µm, presenting a reduction of 1 % and 53 %, respectively, when

compared to the corresponding results registered in the homologous specimens subjected to control exposure and tested in monotonic load conditions. Hence, the fatigue load reduced the ductility of the bond, since the strength remained identical for half of the bond slip.

The specimens subjected to temperature cycles (T series), after the initial "elastic" phase (that ended at about 90 kN) presented a softer response than the specimens of the R series. The lower bond stiffness is in agreement with the results from the fatigue behavior for the same specimen, where the slip amplitude was also larger when compared to other series.

The specimens of the WD series, after the initial "elastic" phase (that ended at about 50 kN), developed a stiffer behavior than the other series, which is in agreement with the results of Fig. 6. The peak load reached 315 kN, at a loaded end slip of 30 µm, which correspond to an increase of 6 % and a reduction of 37% when compared to the results registered in the specimens subjected to wet-dry cycles and tested under monotonic loading conditions. This behavior is also in agreement with the results obtained in the fatigue tests, where the slip amplitude was smaller than the one determined in the specimen of T series, which justifies the stiffer bond response of the specimens submitted to WD conditions.

All PF specimens registered a stiffer peak behavior when compared to the homologous specimens submitted to static loading. This behavior is consequence of the test setup adopted, which in the case of the present monotonic tests was slightly different. Thus, in the present test setup the confinement system (see Fig. 4) was kept tighter, to prevent any brittle failure due to the fatigue history of the specimen, leaving only 1 mm of clearance between the steel member and the concrete plate. Thus, when the passive confinement starts to become active, the system becomes stiffer, leading to a stiffest peak behavior.

Fig. 11 presents the failure modes observed in the specimens that after have been subjected to fatigue loading, were tested under monotonic loading up to failure. Once again, the failure occurred by debonding at the concrete / GFRP flange interface. Moreover, these failure modes were very similar to those registered in the monotonic tests, which means that the failure mode was not affected by the fatigue loading.

#### 4 Numerical simulations

A three-dimensional finite element analysis was conducted to predict the bond behavior of the GFRP-concrete adhesive system using the FEMIX 4.0, a software based on the Finite Element Method [18], which includes several types of finite elements and constitutive models for the linear and material nonlinear analysis of cement-based and polymer-based composite materials. The numerical simulation focused only in the control exposure specimens. The following summarizes the numerical approach.

# 4.1 Brief description of the Model

The adopted model, which is described in detail elsewhere [18], is based on the strain decomposition concept for smeared cracked concrete. Herein only a brief description of the model is given in order to provide the fundamental information for an understanding for the data supplied for modeling the behavior of the intervening materials. Based on the above strain decomposition concept, the constitutive law for the cracked concrete has de following format:

$$\Delta \underline{\sigma} = \left( \underline{D}^{co} - \underline{D}^{co} \left[ \underline{T}^{cr} \right]^{T} \left( \underline{D}^{cr} + \underline{T}^{cr} \underline{D}^{co} \left[ \underline{T}^{cr} \right]^{T} \right)^{-1} \underline{T}^{cr} \underline{D}^{co} \right) \Delta \underline{\varepsilon}$$
 (1)

where  $\underline{D}^{co}$  is the elasticity matrix for concrete between cracks that depends on the Young's modulus (E) and the Poisson's ratio (v) of this material,  $\underline{T}^{cr}$  is the transformation matrix that transforms the stress components from the global coordinate system to the local crack coordinate system  $(n, t_1 \text{ and } t_2 \text{ axes, where } n \text{ is the vector orthogonal to the crack plane, while } t_1 \text{ and } t_2 \text{ are in crack plane)}$ . In Eq. (1)  $\underline{D}^{cr}$  is the crack constitutive matrix:

$$\underline{D}^{cr} = \begin{bmatrix} D_n^{cr} & 0 & 0 \\ 0 & D_{t_1}^{cr} & 0 \\ 0 & 0 & D_{t_2}^{cr} \end{bmatrix}$$
 (2)

where  $D_n^{cr}$ ,  $D_{t_1}^{cr}$  and  $D_{t_2}^{cr}$  represent, respectively, the fracture mode I softening modulus, and the sliding softening modulus in the  $\hat{t}_1$  and  $\hat{t}_2$  direction. The fracture mode I modulus,  $D_n^{cr}$ , is defined by the tensile-softening diagram represented in Fig. 12, while the sliding fracture mode modulus,  $D_{t_1}^{cr}$  or  $D_{t_2}^{cr}$ , can be obtained with,

$$D_{t_1}^{cr} = D_{t_2}^{cr} = \frac{\beta}{1 - \beta} G_c \tag{3}$$

where  $G_c$  is the concrete elastic shear modulus and  $\beta$  is the shear retention factor:

$$\beta = \left(1 - \frac{\varepsilon_n^{cr}}{\varepsilon_{n,u}^{cr}}\right)^{p_1} \tag{4}$$

where  $\varepsilon_n^{cr}$  is the current crack normal strain and  $\varepsilon_{n,u}^{cr}$  is the ultimate crack normal strain. Alternatively, to model the crack shear stress transfer in  $\hat{t}_1$  and  $\hat{t}_2$  directions, and to improve the accuracy of the simulations of structures failing in shear,  $D_{t_1}^{cr}$  or  $D_{t_2}^{cr}$  can be obtained with an independent shear crack softening diagram [18].

## 4.2 Mesh, loading, boundary conditions and constitutive models

Fig. 13 shows the FE mesh geometry adopted in the numerical simulations. As the specimens used are symmetric along the x1 and x2 axis, only one fourth of the specimen was simulated. A relative mesh fine refinement was used with a maximum element size of 25 mm. The boundary conditions were applied to simulate the test conditions (see Fig. 4), by constraining necessary nodes in translation at supports of the specimens. The load was applied uniformly along the upper edge of the GFRP profile in agreement with the test procedure.

Four different constitutive models were used in the simulation of the different components composing the system. All three materials, concrete, adhesive and GFRP, were modeled using 20-node solid finite elements with  $2\times2\times2$  Gauss-Legendre integration scheme. Only one layer of solid elements was used in the adhesive. Perfect bond was assumed

between all the elements. The GFRP profiles were modeled assuming linear and isotropic behavior material, characterized by a Young's modulus of 34 GPa and a Poison's ratio of 0.18 (IV, Fig. 13). The concrete was separated in two layers, one assuming linear material behavior (I) and the other assuming nonlinear material behavior (II). The thickness of the nonlinear material behavior layer was determined from concrete layer peeled off from the experimental results. Hence, the thickness used was 5 mm, which is greater than those from the experimental results. The linear material behavior for concrete is characterized by a Young's modulus of 32 GPa and a Poisson's ratio of 0.20 (I, Fig. 13). The 3D multi-fixed smeared crack model briefly described in the previous section was adopted to simulate the nonlinear material simulation of concrete and adhesive [18]. Table 6 shows the values of the model parameters used in the simulations. The trilinear tensile diagram represented in Fig. 12 was used for modeling the fracture mode I in both the concrete and adhesive. However, the concrete was assumed a tensile strain softening material, while adhesive a tensile strain hardening material (see the values in Table 6 that define the trilinear diagram). A maximum of 2 cracks per integration point was adopted, being the crack bandwidth the square root of the integration point. In the present simulations a value of 2 was assumed for the  $p_1$  parameter in Eq. (4). The parameters of the concrete and adhesive were obtained in previous experimental programs dedicated to the characterization of the SFRSCC and epoxy adhesive [15, 19]. For modeling the tensile nonlinear behavior of the adhesive observed in experimental tensile tests [14], the 3D smeared crack model described in Section 4.1 was used, by considering that the inelastic tensile deformation is caused by the formation of diffuse micro-cracks in its micro-structure, thereby being considered as a strain-hardening material governed by the fracture properties indicated in Table 6. This approach was already adopted with success by Costa and Barros [15]. The smaller elasticity modulus adopted in the numerical simulation for the adhesive simulates the higher deformability that the adhesive presents in the SFRSCC-GFRP connection (when compared to the deformability determined in samples prepared in laboratory and tested in idealized tensile conditions) due to the higher susceptibility of introducing voids and defects in the adhesive layer during its application process. These values were obtained by inverse analysis, by fitting as closest as possible the experimental results.

# 4.3 Results

# 4.3.1 Relative displacements

Fig. 14 shows the points where the measurements were made, in both numerical and experimental procedures. Regarding the relative displacement between the concrete and the GFRP profile, these measurements were made in the experimental tests along the bond length in four different points, as mentioned in Section 2.2. These points (A to D) are equally spaced by 87.5 mm (LVDT's 1 to 5 in Fig. 3). The results presented in this section were compared with a control specimen (R series) under monotonic load.

Fig. 15 shows the results, for the relative displacement between the GFRP flange and the concrete, comparing both the numerical and experimental results, in the four different points (A to D). The trends of the numerical results compared well with those from the experimental results. Thus, as expected, the results show a decrease in the relative displacement as the distance from the loaded-end increases. Hence, in point D the relative displacement reaches only 16 % of the loaded-end slip (point A). The brittle failure of the bond can be explained by its high stiffness, as shown by the very small relative displacements measured.

#### 4.3.2 Stresses and failure modes

Fig. 16 shows the relationship between the applied force and stresses in both materials, concrete and GFRP (x<sub>3</sub> direction). In the experimental tests, the strain was measured using strain gages. Then, the Hooke's law was used to convert the strains in stresses. This assumption for the concrete is acceptable since the level of stress state in the analyzed region was quite low when compared with the compressive strength of the concrete. On the other hand, GFRP material is assumed to be linear elastic up to the failure.

The compressive stress measured in concrete (point E) shows a nonlinear behavior up to 240 kN. This behavior is explained by the degradation of the bond stiffness. As the micro cracking grows, the load transfer between the adhesive and the concrete starts to decrease, causing a slightly reduction in the concrete stress in point E. Then, before failure, the micro cracking growths causing macro-cracks in concrete, resulting in the debonding of the adhesive. This local debonding causes a great reduction in the load transfer between the adhesive and the concrete, resulting in a reduction of stresses in concrete from 240 kN up to failure. In point F, near the supports zone, the compressive stress is higher and present a linear response, as expected, reaching 9 MPa.

The compressive stress measured in the GFRP profile in point G (see Fig. 16) also exhibits a linear behavior up to 240 kN. After the 240 kN load, the stress increases significantly. This increase is caused by the reduction in the load transfer between the adhesive and the concrete, i.e., as the connection starts to debond, the load starts to be transferred to the profile, causing an increase in the compressive stress in the GFRP, and a reduction in the compressive stress in the concrete.

The crack pattern presented in Fig. 17, regarding the cracks in the concrete, illustrates the phenomena described previously. It is noticeable that at a load of 150 kN a significant number of cracks occurred in the loaded-end zone, that significantly increase up to a load of 240 kN, causing an increase in the compressive stress in the GFRP profile (point G) along with a reduction in the compressive stress in the concrete (point E). At a load of 200 kN the cracks start to appear along the bond length, increasing up to the load of 240 kN. Thus, the bond surface starts to be filled with cracks, weakening the connection and leading to the failure of the system. Regarding the adhesive, only few micro cracks occurred, which did not contributed to the system failure.

#### 5 Conclusions

The performance under static, fatigue and aggressive exposure of the proposed adhesive joints was investigated experimentally. Based on the results of the investigations presented in this paper, the following conclusions can be pointed out:

- The results from the monotonic tests evidenced that when the connection is exposed to the temperature cycles (T series) a reduction of 41 % in the peak load was obtained, when compared to the control exposure (R series), indicating that the detrimental effect of the temperature cycles should be considered on the design evaluation of the long term performance of this type of connection.
- The failure modes for the monotonic tests showed that all specimens failed by debonding. However, the area of the concrete layer peeled off has varied with the type of exposure, and thus affecting the bond stiffness and strength of the connection. For

the T and WD series, in some portions of the bonded area, cohesive failure of the adhesive occurred due to the aggressive exposure, being more pronounced in the WD series.

- The fatigue behavior showed a slightly continuous degradation of the bond interface during the fatigue load cycles. These results also revealed that during the fatigue loading the specimens from the WD series presented a less stiff response, since the slip amplitude was larger than those of the reference specimens. Moreover, from the maximum and the minimum slip curves, it is noticeable that the stiffness remained constant during the fatigue loading for the three series. After 1 million cycles, the specimens did not display any visible damage.
- The post-fatigue monotonic behavior was affected by the fatigue loading, i.e., in terms of initial stiffness the fatigue loading had a favorable effect of increasing significantly bond stiffness up to a load interval between 50 and 100 kN. Then, when the monotonic load level applied to the specimen exceeded the maximum load level the specimen has experienced in the fatigue loading process, a significant reduction in the bond stiffness has occurred, becoming identical to the stiffness registered in the monotonic tests.
- By simulating a representative push-out test of a specimen of the R series subjected to
  monotonic loading, it was verified that by using a 3D smeared crack model for
  modeling the material nonlinear behavior of SFRSCC and adhesive, and assuming the
  GFRP profile as formed by a linear and elastic material, the relevant results obtained
  experimentally were predicted accurately.

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Table 1 - SFRSCC mix proportion per m<sup>3</sup>

Cement [kg]	Limestone filler [kg]	Water [kg]	Superplasticizer [kg]	Fine sand [kg]	River sand [kg]	Crushed stone [kg]	Fibers [kg]
380.54	353.00	140.00	7.83	237.00	710.00	590.00	60.00

Table 2 – (a) Rheological and (b) mechanical properties of the SFRSCC.

(a)

Parameter	Value			
Slump flow	Diameter =80 cm			
Stump now	T50=4 s			
V-funnel	T=9 s			
L-box	H2/H1=0.8			

*(b)* 

С	ompressive strength [MPa]	Young's modulus [GPa]	Flexural tensile strength [MPa]	f <sub>R,1</sub> [MPa]	f <sub>R,2</sub> [MPa]	f <sub>R,3</sub> [MPa]	f <sub>R,4</sub> [MPa]	f <sub>eq,2</sub> [MPa]	f <sub>eq,3</sub> [MPa]
	47.00	32.00	10.20	9.15	11.34	9.08	7.54	9.06	9.85

Table 3 - Main mechanical properties of the GFRP profile [12]

GFRP	Tension (longitudinal		Compression (longitudinal		Compression (transverse	
profile	direction)		direction)		direction)	
	$\sigma_{\mathrm{tu,L}}  [\mathrm{MPa}]$	E <sub>t,L</sub> [GPa]	$\sigma_{\mathrm{cu,L}}$ [MPa]	E <sub>c,L</sub> [GPa]	σ <sub>cu,T</sub> [MPa]	E <sub>c,T</sub> [GPa]
Webs	385.3±23.7	33.0±1.2	460.4±64.4	26.7±1.2	106.1±28.5	7.67±2.0
Flange	414.2±21.0	35.9±2.1	388.6±59.2	$33.9 \pm 3.4$	70.4±17.4	5.9±0.6

Notes:  $\sigma_{tu,L}$  = ultimate tensile strength in the longitudinal direction;  $E_{t,L}$  = tensile Young's modulus in the longitudinal direction;  $\sigma_{cu,L}$  = ultimate compression strength in the longitudinal direction;  $E_{c,L}$  = compression Young's modulus in the longitudinal direction;  $\sigma_{cu,T}$  = ultimate compression strength in the transverse direction;  $E_{c,T}$  = compression Young's modulus in the transverse direction.

Table 4 – Specimens and test program

Exposure regime	Duration	No. of specimens
Wet-dry	100 cycles (one cycle = 1 day)	6
Temperature cycle	100 cycles (one cycle = 4 hours)	6
Control	3 months	6

Table 5 – Monotonic and post-fatigue test results.

Specimen	Peak load	Average (CoV %)	Strength	Max slip	Average (CoV %)	Slip ratio
ID	(kN)		ratio (%)	(µm)		(%)
R_S1	220.5			64.0		
R_S2	314.7	284.0 (19.3)	100	69.2	60.7 (17.2)	100
R_S3	317.0			49.0		
T_S1	208.0		,	65.7		
T_S2	192.0	194.3 (6.5)	68.4	30.7	46.1 (38.8)	76.0
T_S3	183.0			41.8		
WD_S1	296.2		,	48.3		
WD_S2	294.0	283.3 (7.2)	99.8	52.7	51.9 (6.2)	85.5
WD_S3	259.6			54.6		
R_PF1	333.0			10.0		
R_PF2	333.0	325.3 (4.1)	115.0	16.3	19.5 (59.0)	32.1
R_PF3	310.0			32.4		
T_PF1	241.0			43.8		
T_PF2	298.0	257.3 (13.8)	90.6	14.0	24.3 (69.7)	40.0
T_PF3	233.0			15.0		
WD_PF1	222.0			10.0		
WD_PF2	226.0	254.3 (20.8)	89.5	28.0	23.0 (49.4)	37.9
WD_PF3	315.0			31.0		

Table 6 – Concrete and adhesive properties used in the numerical simulation (see [18]).

Parameter	Unit	Value			
i didiletei	Oille	Concrete (II)	Adhesive (III)		
Young's modulus	GPa	32.0	4.5		
Compressive strength	MPa	47.0	70.0		
Tensile strength	MPa	2.3	6.4		
Fracture energy $(G^I_f)$	N/m	2.9	0.2		
$\alpha_I^{\ *}$	-	0.5	1.5		
$a_2$ *	-	0.2	2.0		
ξ1*	-	0.05	0.06		
ξ <sub>2</sub> *	-	0.5	0.14		
Threshold angle	0	30	30		

 $*\left(\alpha_{1} = \sigma_{n,2}^{cr} / \sigma_{n,1}^{cr}, \alpha_{2} = \sigma_{n,3}^{cr} / \sigma_{n,1}^{cr}, \xi_{1} = \varepsilon_{n,2}^{cr} / \varepsilon_{n,u}^{cr}, \xi_{2} = \varepsilon_{n,3}^{cr} / \varepsilon_{n,u}^{cr}\right)$ 

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