

24 1. INTRODUCTION

25 Repair and strengthening solutions are commonly adopted in existing reinforced concrete (RC) structures
26 as a way to preserve and rehabilitate them. The use of Fibre Reinforced Polymers (FRP) applied
27 according to the Near Surface Mounted (NSM) technique is one of the possibilities for strengthening
28 existing RC structures. NSM technique is based on the insertion of the reinforcing composite material in a
29 groove cut in the concrete cover of the element to be strengthened [1-4]. Typically, stiff and hardly
30 deformable epoxy adhesives are used to fix the FRP to concrete. This bonding agent plays a key role on
31 the composite action of the system, because it is mainly responsible for the stress transferring between the
32 FRP and the concrete substrate, in both service and ultimate limit states.

33 There are very few studies intended at assessing the influence of the adhesive stiffness and deformability
34 on the flexural behaviour of RC elements strengthened using the Carbon FRP (CFRP) laminate systems.
35 Only two investigations were found, both about the EBR (Externally Bonded Reinforcement) and none
36 about the NSM technique. Derkowski et al. [5] performed a research on the use of stiff and flexible
37 adhesive layers of different stiffness (Young's module ranging between 2 MPa and 13 000 MPa) and
38 deformability (ultimate strains ranging between 0.2% and 150%) for bonding CFRP laminates in the
39 flexural strengthening of RC beams. In this investigation, RC beams were strengthened with CFRP
40 laminates using a stiff epoxy adhesive and five types of polymer adhesives of different flexibility. The
41 beams were monotonically tested. Regarding the results, the authors reported the advantage of using
42 highly deformable (flexible) adhesives in external bonding (EB) of CFRP laminates to RC beams as
43 strengthening solution, such as:

- 44 i) more uniform distribution of CFRP strains along its length and smallest deflection (mainly in the
45 case of the middle hard flexible polymer);
- 46 ii) protection against higher stress concentration in the CFRP due to cracking, reducing the risk of
47 CFRP failure at cracks;
- 48 iii) higher load carrying capacity of these structural elements.

49 Kwiecień et al. [6] tested the flexural efficiency of an innovative solution for repair and strengthening of
50 RC structures, consisting of the simultaneous use of rigid and flexible adhesive layers with CFRP
51 laminates externally bonded at the bottom of a RC beam, previously ruptured in a fatigue test. The results
52 indicate that this new system increases significantly the ductility of the repaired RC beam in the post-peak

53 behaviour and eliminates the disadvantage of brittle and rapid failure (without warning) of the composite-
54 to-concrete-joint. Moreover, the FRP composite system with stiff epoxy adhesive, when strengthening the
55 cracked RC concrete beam, is characterised by a brittle behaviour and vulnerability to uneven
56 deformations generating a notch effect and stress concentrations, which does not happen with this flexible
57 alternative bonding. Kwiecień [7] studied the bond behaviour of EB systems applied in CFRP
58 strengthening of masonry, by using stiff (an epoxy resin) and flexible (five polyurethane polymers)
59 adhesives in single-lap shear tests. The author concluded that the tested flexible adhesive was more
60 effective than the stiff one. Using the flexible polyurethane adhesive PS, higher ultimate forces (of about
61 42%) and ultimate slips (of about 63%) were reached in comparison to the stiff adhesive. The loaded-end
62 slip attained with the flexible adhesive was about 40 times higher than with the stiff adhesive. Thus, the
63 shear stress in the adhesive layer was reduced by the adhesive flexibility and more evenly redistributed
64 along the bonding length. The author concluded also that the flexible polymers protected the brittle
65 substrate against the local shear stress concentrations at the loaded-end caused usually by stiff adhesives,
66 which is responsible for the activation of the rapid detachment process.

67 The existing guidelines do not explicitly consider the influence of the adhesive type on the provisions for
68 estimation of the flexural capacity of RC members strengthened with NSM FRP systems. D'Antino and
69 Pisani [8] assessed the accuracy of the models proposed in four guidelines, namely ACI 440.2R-08 [9],
70 TR 55 [10], CSA S806-12 [11] and CNR DT 200 R1/2013 [12] for the estimation of the flexural capacity
71 of RC beams strengthened with NSM CFRP composites. From the analysis performed the authors
72 concluded that the procedures considered in these guidelines are based on the assumption that there is no
73 relative slip between FRP reinforcement and concrete. Moreover, only the English TR 55 establishes the
74 limitation of the strain in the NSM reinforcement to prevent the failure of the adhesive layer.

75 Taking into consideration the advantages of using flexible adhesives referred by the existing literature,
76 and the lack of experience in using these adhesives adopting the NSM technique, in the present work the
77 following effects on the flexural behaviour of the RC slabs strengthened with the NSM CFRP technique
78 are studied:

- 79 i) the using of different types of adhesives (stiff and flexible);
- 80 ii) the existence or not of cracks in concrete before applying the composite strengthening.

81 In the following sections, the experimental program is detailed and the main results obtained are
82 presented and analysed. In the last part of this work, the numerical simulations carried out to simulate the
83 experimental responses of the tested slabs are introduced and discussed.

84 **2. EXPERIMENTAL PROGRAMME**

85 **2.1 Test programme**

86 The experimental program (see Table 1) consisted on the flexural testing of seven slab specimens. The
87 investigation involved the study of three adhesive types, namely:

- 88 i) adhesive 1 (ADH1);
- 89 ii) adhesive 2 (ADH2);
- 90 iii) adhesive 3 (ADH3).

91 Each adhesive type was applied in two slabs:

- 92 i) a slab without pre-cracking (U);
- 93 ii) a slab which was pre-cracked before applying the strengthening system (C).

94 The cross-section of CFRP laminate adopted was 1.4×20 [mm]. In this study, it was also included a
95 control slab (without the application of the strengthening system). The generic denomination adopted for
96 the slab specimens is SL_ADHX_Y where X represents the adhesive type (1, 2 or 3) while Y indicates
97 the absence or presence of pre-cracking (U - Uncracked and C - Cracked). The control slab was named as
98 SL_REF.

99 **2.2 Slab geometry and test configuration**

100 Figure 1 presents the geometry of the slab specimens, as well as the details of the strengthening system
101 and test configuration adopted. The slabs have a total length of 2600 mm with a rectangular cross-section
102 of 600×120 [mm]. The bottom steel reinforcement was composed of 4 steel bars of 8 mm diameter (4Ø8),
103 which corresponds to a longitudinal reinforcement ratio, ρ_l , equal to 0.35%, while the top steel
104 reinforcement was composed of 3Ø6. Steel stirrups of 6 mm diameter spaced of 300 mm were adopted
105 (Ø6@300). The concrete cover was equal to 20 mm at the slab top and both sides and to 25 mm at the
106 bottom. The strengthening solution is composed of 2 CFRP laminates of 1.4×20 [mm] installed in the
107 concrete cover according NSM technique. The main purpose of this strengthening solution was to double
108 the load carrying capacity of control slab (SL_REF). The corresponding equivalent longitudinal steel

109 reinforcement ratio ($\rho_{s,eq}$) is equal to 0.49%, according to Sena-Cruz et al. [2]. The grooves used for
110 installing the CFRP laminates have a constant cross-section of 5×25 [mm]. The CFRP laminates have a
111 total length of 2200 mm coinciding their mid length with the mid span of the slab span. The absence of
112 CFRP reinforcements at the extremities of the slab was adopted to avoid the confinement effect provided
113 by the supporting conditions to the reinforcing materials during the test.

114 A four-point bending loading test configuration was adopted to perform the quasi-static monotonic tests
115 (see Figure 1a). The distance between supports (span length) was 2400 mm, being the shear span of
116 900 mm (i.e. 7.5 times the slab thickness). The slab's instrumentation included the measurement of the
117 applied load (F) using a load cell with the maximum capacity of 200 kN and a linear error of $\pm 0.05\%$. For
118 measuring the deflection along the longitudinal axis of the slab, 5 linear variable displacement transducers
119 (LVDT1 to LVDT5) were used: 3 LVDTs in the pure bending zone (range of ± 75 mm and linearity error
120 of $\pm 10\%$) and 2 LVDTs (range of ± 25 mm and linearity error of $\pm 10\%$), one in each side, at mid distance
121 between the bottom supports and the line loads. The strains in the materials composing the slabs were
122 also assessed using strain gauges glued (see Figure 1b):

- 123 i) on bottom steel bars at mid-span (SG1 and SG2);
- 124 ii) on concrete under compression stress state at the top fibre in mid-span (SG3);
- 125 iii) on the CFRP laminates - two strain gauges were placed at mid-span (SG4 and SG5), one at the
126 loading point (SG6) and two between the loading point and CFRP extremities (SG7 and SG8).

127 Two different types of strain gauges were used:

- 128 i) TML BFLA-5-3-3L for steel bars and CFRP laminates;
- 129 ii) TML PFL-30-11-3L for concrete.

130 The tests were conducted using a servo-controlled equipment under displacement control (controlled by
131 the internal displacement transducer of the actuator) at a rate of 20 $\mu\text{m/s}$. During the tests the crack width
132 evolution of 3 cracks in the pure bending zone was measured using a handheld USB microscope
133 (VEHO VMS-004 D microscope), which has a native resolution of 640×480 pixels and magnification
134 capacity up to 400 times. In the present study, the evolution of the crack width was monitored with a
135 magnification factor of 20 times up to a pre-defined load in order to assure the safety of the operator.

136 **2.3 Material characterization**

137 In this work, the mechanical properties of concrete, stiff adhesives and steel bars were experimentally
138 assessed while the mechanical properties of both the CFRP laminate strips and the flexible adhesive were
139 taken from other publication, since the material came from the same batch.

140 All slabs composing the experimental program and the specimens used for the assessment of the
141 concrete's mechanical properties were cast from a single batch. The following characteristics were
142 adopted for the concrete:

- 143 i) strength class: C30/37;
- 144 ii) exposure class: XC4;
- 145 iii) maximum aggregate size: 12.5 mm;
- 146 iv) slump class: S3;
- 147 v) cement: CEM II/A-L 42,5R.

148 The concrete composition included 907 kg/m³ of coarse aggregates, 915 kg/m³ of fine aggregates, 310
149 kg/m³ of cement, 4.96 kg/m³ of admixture and a water/cement (W/C) ratio equal to 0.58. The concrete's
150 modulus of elasticity (E_c) and compressive strength (f_c) were assessed using cylinders with diameter of
151 150 mm and height of 300 mm, at 28 and 110 days after casting (the latter date coincides with the date of
152 slab's testing). The modulus of elasticity and the compressive strength were assessed according to LNEC
153 E-397-1993:1993 [13] and NP EN 12390-3:2009 [14] recommendations, respectively. Table 2 presents
154 the obtained results (average values).

155 The used steel bars are of the class A400 NR SD (Eurocode 2 [15]) according to the NP EN ISO 6892-1
156 [16] and their mechanical properties were assessed through uniaxial tensile tests. Three specimens were
157 used for each diameter. Table 2 presents the average values obtained for the yield stress (f_y) and ultimate
158 strength (f_{su}).

159 The CFRP laminate strips used for strengthening the slabs are produced by S&P® Clever Reinforcement
160 Ibérica with the trademark CFK 150/2000. These CFRP laminates are composed of unidirectional carbon
161 fibres agglutinated through an epoxy vinyl ester resin matrix, presenting a smooth surface. The content in
162 fibres relative to matrix is about 70% (in volume). As mentioned above, the mechanical properties of the

163 CFRP laminate strips were assessed in other publication – see in [17]. Table 2 presents the average values
164 obtained for the modulus of elasticity (E_f), tensile strength (f_{fu}) and ultimate strain (ε_{fmax}).

165 The stiff adhesives ADH1 and ADH2 (epoxy resins), have commercial trademarks Sikadur-30 and S&P
166 Resin 220, respectively. The flexible adhesive, ADH3, with the commercial trademark of Sika PS, is a
167 polyurethane polymer. During its application, the stiff adhesives have shown high viscosity, while the
168 flexible adhesive exhibited low viscosity and high flexibility after curing. The adhesives are provided by
169 the supplier in the form of two components (Component A = resin and Component B = hardener), which
170 need to be mixed before application according the supplier's ratios A:B of 3:1, 4:1 and 9:1 for ADH1,
171 ADH2 and ADH3, respectively. The tensile mechanical properties of ADH1 and ADH2 were obtained by
172 performing tests according to the ISO 527-2:2012 [18], while the mechanical properties of ADH3 were
173 previously assessed by Kwiecień [7], also according to the ISO 527-2:2012. Table 2 presents the results
174 obtained for the elastic modulus (E_a), tensile strength (f_a) and ultimate strain (ε_{amax}) for ADH1 and ADH2
175 and the values collected for ADH3. It should be noted that both stiff adhesives have shown approximately
176 similar mechanical properties, while adhesive ADH3 has shown significantly lower modulus of elasticity
177 and tensile strength, but much higher ultimate strain.

178 **2.4 Specimen preparation**

179 The preparation of the strengthened slabs included several steps, namely:

- 180 i) casting;
- 181 ii) groove's opening using a saw-cut machine with a diamond disk;
- 182 iii) pre-cracking (series "_C" only);
- 183 iv) cleaning of the grooves and CFRP laminates with compressed air and acetone, respectively;
- 184 v) application of a special primer (chemically compatible) on the groove surface (for the case of slabs
185 using ADH3), as recommended by the supplier;
- 186 vi) application of the adhesive on turned upside down specimens: ADH1 and ADH2 were applied
187 with the assistance of a spatula while the ADH3, due to its low viscosity, was applied by gravity
188 (see Figure 2); and finally,
- 189 vii) introduction of the CFRP laminate in the groove and regularization of the surface.

190 The specimens were kept in laboratory environment for approximately one month and a half before being
191 tested.

192 In the series “_C”, the pre-cracking process was performed using the same test configuration used for the
193 tests up to failure described previously. The main difference was that the pre-cracking process was
194 performed under force control at a rate of 0.05 kN/s up to a force of 15 kN, which corresponds
195 approximately to 2/3 of the load carrying capacity of the control slab (SL_REF). When this load level was
196 achieved, this value of force was kept constant for 10 minutes to mark the existing visible cracks and
197 measure the width of the cracks. During the time while the force remained constant, there was an increase
198 on mid-span displacement due to creep. After this period, the (total) mid-span deflection was about
199 13 mm. Finished this task, the slabs were unloaded. Then, the slabs unstressed (but with residual
200 deformations) were strengthened with the CFRP laminates following the procedure described at the
201 beginning of this section.

202 The response of the slabs “_C” was very similar to the one obtained in the control slab. During the
203 unloading process, it could be observed the recovery of the elastic deformation, with a remaining residual
204 mid-span deflection of about 6 mm (44% of the observed maximum displacement) and the steel residual
205 stain of about 0.1%.

206 **3. EXPERIMENTAL RESULTS AND DISCUSSION**

207 **3.1 Main results**

208 Table 3 presents the main results obtained. In this table, K_I , K_{II} and K_{III} represent the flexural stiffness in
209 each of the three main representative stages, respectively: i) elastic phase, ii) cracked phase, and iii) post
210 steel yielding phase. These parameters were determined by computing the slope of the corresponding
211 branch using two representative points; F_{cr} , F_y and F_{max} correspond to the force at the cracking initiation,
212 bottom steel yielding and maximum force, respectively, and δ_{cr} , δ_y and δ_{max} correspond to the mid-span
213 displacements at F_{cr} , F_y and F_{max} , respectively; ε_{fmax} is the maximum strain attained in the CFRP laminate
214 at F_{max} . The values between brackets represent the increase of load carrying capacity compared to the
215 control slab. The ductility of each slab was also assessed through the parameter δ_{max}/δ_y . The ratio between
216 the residual force (at the end of the test) and the corresponding maximum force, F_r/F_{max} is also included.
217 Finally, the last column includes the observed failure modes.

218 Figure 3 presents the applied force *versus* mid-span displacement relationships obtained for the tested
219 slabs. These relationships present the typical behaviour observed in RC slabs strengthened in flexure with
220 NSM-CFRP systems. It is observed an increase in load carrying capacity due to the strengthening

221 application. In the case where no pre-cracking was applied to the slabs, three main phases can be
222 observed:

- 223 i) the elastic phase (I), from the beginning of the test up to the crack initiation without significant
224 change in stiffness when compared with SL_REF, due to the reduced amount of the CFRP
225 reinforcement utilized;
- 226 ii) the cracked phase (II), from the crack initiation up to the steel yielding, where the contribution of
227 the CFRP reinforcement starts playing an important role;
- 228 iii) the post yielding phase (III), from the steel yielding up to the maximum load carrying capacity,
229 where the contribution of the CFRP reinforcement is responsible for carrying the additional
230 increments of load.

231 As expected, the elastic phase does not exist on the pre-cracked slabs since they were pre-cracked before
232 the application of strengthening (see Figure 3b). It is also observed a decrease of the flexural stiffness
233 along the test due to the increase of damage in the composing materials, as well as cracking and the
234 degradation of the bond properties between materials (steel/concrete and CFRP/adhesive/concrete).

235 As stated before, similar responses were obtained during the elastic phase for all slabs, including SL_REF
236 due to the low amount of strengthening reinforcement applied. In the cracked phase, all the strengthened
237 slabs exhibited very similar behaviour. However, at the yielding point important differences can be
238 observed for the different types of adhesive used: SL_ADH1 and SL_ADH2 present a higher yielding
239 load than the SL_ADH3, regardless of the existence or not of pre-cracking. This behaviour is due to the
240 level of slip between the CFRP and concrete occurred at this load level. This slippage is directly
241 controlled by the stiffness of the adhesive. After yielding, slabs SL_ADH1 and SL_ADH2 exhibited an
242 almost linear elastic behaviour. Since stiff ADH1 and ADH2 provided higher level of bond between the
243 CFRP and concrete linked with low levels of slip, the tensile failure of CFRP was achieved. As expected,
244 after failure of the CFRP of these slabs (SL_ADH1 and SL_ADH2) the flexural response resembles the
245 one observed in the SL_REF. It should be noted that on slabs SL_ADH3, the third branch is different
246 from the one observed on SL_ADH1 and SL_ADH2, with a pronounced non-linear relationship between
247 the applied force and the deflection at mid-span. This behaviour is mainly governed by the significant
248 amount of slip between the CFRP laminates and the concrete. Due to that, the CFRP failure did not occur.
249 After reaching the maximum load, those deformation coincides with the deformation achieved in the slabs

250 SL_ADH1 and SL_ADH2 at the maximum load, a softening branch with a gradual decrease of strength is
251 observed, with a significant residual strength (for 120 mm of deflection was about 77% of F_{max}). This
252 more deformable response observed in SL_ADH3 can be explained by the initial higher shear
253 deformation of the flexible adhesive and then the progressive loss of bond between the CFRP laminate
254 and concrete, and by the progressive increasing cohesive failure at the adhesive, which decreases the
255 contribution of the CFRP laminates for flexural capacity of these slabs after reaching the maximum load.

256 3.2 Strains

257 3.2.1 Force *versus* mid-span CFRP strain

258 Figure 4 presents the force *versus* mid-span CFRP strain relationships obtained for the strengthened slabs.
259 As in force *versus* mid-span displacement relationships, three phases can be observed for uncracked series
260 and two phases can be observed for cracked series. In general, the level of mobilization of the CFRP is
261 higher with stiff adhesives than with flexible one, which prove the higher capacity for stress transfer of
262 the stiff adhesives. Contrarily to SL_ADH1 and SL_ADH2, the CFRP did not fail suddenly after F_{max} on
263 SL_ADH3, but the slow decreasing of CFRP strain can be observed after the peak load was reached.
264 Most likely this is the result of the CFRP gradual slippage, as a consequence of the loss of bond at
265 laminate adhesive interface. Finally, it should be noted that a smaller level of CFRP strain was attained at
266 the initial phase of the test of the uncracked series, perhaps due to the contribution of the uncracked
267 concrete under tension, as opposed to what was observed for the cracked series. After the crack initiation,
268 the process of stresses transfer from the concrete under tension to the CFRP laminate results in a sudden
269 increase of the CFRP strain when stiff adhesives are used. In contrast, in the slab where flexible adhesive
270 was used this increase was not observed, but an almost monotonic strain increase was obtained. This may
271 reveal a reduction of CFRP's stress concentration at the cracks locations, and the stress redistribution
272 along the CFRP laminate provided by the flexible adhesive, leading to a smoother mobilization of the
273 mechanical properties of the CFRP laminate as discussed in [5]. Flexible adhesive (SL_ADH3) protected
274 the CFRP laminate against the notch effect at the crack locations, which was responsible for the CFRP
275 laminate failure when stiff adhesives were used (SL_ADH1 and SL_ADH2).

276 3.2.2 Force *versus* mid-span steel strain

277 Figure 5 presents the force *versus* mid-span steel strain relationships for the bottom steel reinforcement.
278 First, it should be mentioned that these results are dependent on the position of the strain gauge in relation

279 to the concrete cracks: if a strain gauge is located in the point where a crack opens, higher values of strain
280 are measured than if a strain gauge is placed in a zone between two cracks. The results show that:

- 281 i) for uncracked series, up to the crack initiation, the strains on steel are very small; after the crack
282 initiation, a huge increase of the strains was observed;
- 283 ii) for the cracked series, the yielding strains are lower than the ones observed for uncracked series,
284 probably due to the residual strain present in steel resulting from the pre-cracking process;
- 285 iii) on cracked series, the mobilization of the steel reinforcement since the beginning of the test is
286 higher than in the uncracked series, as a result of the cracked state.

287 3.2.3 Force *versus* mid-span concrete strain

288 Figure 6 presents force *versus* mid-span concrete strain relationships. The results show that:

- 289 i) in general, SL_ADH1 and SL_ADH2 slabs presented concrete strains lower than SL_ADH3 slabs;
290 ii) in the transition between the elastic and cracked phase, and after steel yielding, there was a huge
291 increase in the concrete strains, higher on SL_ADH3 than on SL_ADH1 and SL_ADH2;
292 iii) higher level of strains was achieved in the cracked series.

293 3.3 Failure modes

294 Figure 7 shows images of the CFRP reinforcement obtained after failure. Two types of failure modes,
295 related to the mechanical properties of the adhesives, were observed in this study:

- 296 i) slabs SL_ADH1 and SL_ADH2 failed by rupture of the CFRP laminate at mid-span (see Figure
297 7a). For the SL_ADH1 and SL_ADH2 slabs, in some zones of the strengthening, it was possible to
298 observe cracks on the adhesive-concrete interface;
- 299 ii) slabs SL_ADH3 failed by debonding of the CFRP laminate at laminate-adhesive interface, mainly
300 at mid-span (see Figure 7b), and by cohesive failure of the adhesive on other parts along the
301 strengthening, mainly at the ends (see Figure 7c and Figure 7d). This cohesive failure of the
302 adhesive at the ends took place only in one laminate of each slab SL_ADH3.

303 3.4 Crack width, crack pattern and average crack distance

304 As stated in Section 2.2, the crack width of target cracks was monitored using a handheld USB
305 microscope with a magnification factor of 20 times. To perform this, three cracks were selected on each
306 slab in the pure bending zone: one crack as close as possible to the mid-span and the other two close to

307 the loading points. Then, during the tests, images selected from each crack were periodically captured.
308 For each crack image, three measurements were performed in order to obtain the average crack width.
309 The increase in crack width due to the increase of the imposed vertical displacement was measured up to
310 a threshold of about 30 kN of load, due to safety reasons.

311 Figure 8 presents the evolution of the crack width against the increase of the applied force approximately
312 up to the steel yielding. From this stage onwards the $F-w$ relationship stopped following the linearity
313 observed up to this phase. The results show that, for the same load level, there is a tendency for higher
314 crack openings on the slab SL_ADH3 than in slabs SL_ADH1 and SL_ADH2. Comparing uncracked
315 series with the cracked series, some findings can be pointed out: the crack width measurements started
316 earlier on the cracked series, since at the beginning of the test there were already cracks, as opposed to the
317 uncracked series.

318 The final crack pattern obtained in each slab was evaluated after the test. Figure 9 presents the results
319 obtained for both lateral and bottom surfaces of each slab. Figure 10 presents the values obtained for the
320 average crack spacing. These results were obtained by measuring the distance between cracks at the edge
321 between the lateral surface exposed during the test and the bottom surface.

322 Analysing the results, in general, strengthening leads to an:

- 323 i) increase on the number of cracks;
- 324 ii) increase on the crack band zone, which tends to extend from the pure bending zone toward the
325 ends of the slabs (see Figure 9);
- 326 iii) and, reduction of the average crack spacing, as already observed by Correia et al. [19].

327 When SL_REF is used as a comparison, at the uncracked series, the reduction on the average crack
328 spacing was about 20%, 29% e 16% respectively for the slabs SL_ADH1_U, SL_ADH2_U and
329 SL_ADH3_U. On cracked series, the values of spacing reduction were 34%, 23% e 12%, respectively for
330 the slabs SL_ADH1_C, SL_ADH2_C and SL_ADH3_C. The results indicate that the stiff adhesives
331 provide higher reduction of crack spacing than the flexible one.

332 The crack pattern obtained for SL_ADH3 slabs was approximately similar to the one obtained for
333 SL_REF, with higher average crack spacing than the one in slabs SL_ADH1 and SL_ADH2, as well as
334 less increase in the number of cracks and on crack width (see Figure 9). This behaviour can be explained

335 by the less efficient flexible adhesive on CFRP mobilisation during the test. In fact, according to the
336 literature, the increase on the amount of the reinforcement leads to a decrease on the necessary distance
337 for appearance of a new crack between two existing cracks. Then, once ADH3 is less efficient, the
338 distance necessary for the formation of a new crack is higher. Thus, the crack pattern is significantly
339 influenced by the adhesive type. The typical "fish-spine" crack pattern observed for stiff adhesives (e.g.
340 Oehlers et al. [20]) on the concrete close to the groove of the strengthening system was also not visible in
341 the flexible adhesive's slabs.

342 Comparing the uncracked and cracked series, there was a higher average crack distance on cracked series,
343 except for SL_ADH1_C. Finally, the crack width is slightly higher in cracked series for slabs
344 strengthened with stiff adhesive and essentially similar for slabs where the flexible adhesive was used.
345 The number of cracks tends to be lower in cracked series.

346 **3.5 Influence of adhesive type and pre-cracking on the flexural behaviour of the slabs**

347 In this section, the results previously presented are analysed considering i) the force achieved at crack
348 initiation, at steel yielding and at the maximum force, ii) maximum CFRP strain and iii) the ductility and
349 the residual force observed in each slab.

350 Figure 11a presents the force recorded at crack initiation and its respective increase comparing to
351 SL_REF. The presence of the strengthening leads to an increase in force at crack initiation in comparison
352 to SL_REF, being observed an average increase of 42% comparing to SL_REF (see Table 3 and Figure
353 11). Thus, it can be concluded that at this stage, the flexural behaviour probably is not dependent of the
354 adhesive type since they may have similar mechanical behaviour.

355 Figure 11b presents the force achieved at yielding phase for each slab and its respective increase
356 compared to SL_REF. According to the results, the corresponding force was very similar on the slabs
357 SL_ADH1 and SL_ADH2 (cracked or not), as expected. However, the slab SL_ADH1 presented a
358 slightly higher value, perhaps due to the slightly higher mechanical properties of this adhesive. In the case
359 of SL_ADH3, the increase of the load at this stage was smaller than with the application of the stiff
360 adhesives. According to the values shown in Table 3 and Figure 11b, the presence of pre-cracking
361 resulted in a decrease of 1.1%, 2.1% and 10.0% of yielding force with respect to the uncracked slabs,
362 respectively for SL_ADH1, SL_ADH2 and SL_ADH3. From these results it can be concluded that, with
363 the stiff adhesives, the presence of pre-cracking had little influence on the level of force reached at the

364 yielding phase. However, when considering the flexible adhesive, this decrease was more significant. The
365 small decrease in the yielding force observed with the stiff adhesives may be the result of a residual
366 deflection related with internal residual strains on the bottom steel reinforcement after the pre-cracking
367 process, which leads to slightly reduced efficiency of the strengthening.

368 Figure 12a shows the maximum force achieved in each slab during testing and its respective increase
369 compared to SL_REF. Similar values of load increase were obtained for the slabs strengthened using stiff
370 adhesives. In contrast, for the slabs where the flexible adhesive was used, the load carrying capacity was
371 18% smaller, for both uncracked and cracked series, than the average value reached for the slabs
372 strengthened using stiff adhesives. Thus, a better performance was obtained with stiff adhesives, which
373 can take better advantage of the CFRP tensile strength. On the contrary, the tensile strength was not
374 achieved with the flexible adhesive and the CFRP slippage was observed instead of CFRP failure (see
375 Figure 7). Comparing both series, a slight decrease of the maximum force was observed between slabs of
376 the same adhesive. Thus, it can be concluded that the presence of cracking does not affect significantly
377 the performance of these slabs.

378 Figure 12b presents the maximum CFRP strain. Higher values were obtained for slabs SL_ADH1 and
379 SL_ADH2 than for slabs SL_ADH3 (in average, 32% higher). Regarding the influence of the pre-
380 cracking, no significant changes in strain were observed, with decreases on SL_ADH1, SL_ADH2 and
381 SL_ADH3 of 5.7%, 4.3% and 3.8%, respectively. Similar values of maximum strain on CFRP laminate
382 were observed, independently of the presence of pre-cracking. This is due to the fact that, at the moment
383 of the failure of the slab, the initial existing residual deformation/strain state have minor influence on the
384 loading carrying capacity of the slabs, since the involved materials (concrete and steel) at the pre-cracking
385 phase were submitted to relatively low levels of stresses. This fact can also explain the similar values
386 obtained for the maximum force. Similar results are found in the literature, in studies where the lower
387 influence of pre-cracking was observed (e.g. Juvandes et al. [21]). It should be highlighted that the
388 damage applied to the slabs (pre-cracking) does not fully represent the typical conditions on existing
389 structures, since they usually are also stressed, at least due to the self-weight. Thus, the linking of this
390 study with a real applications should be carefully judged (e.g. the yielding of internal steel reinforcement
391 may occur at loads lower than the ones observed in the present case).

392 The ductility was assessed in this study by computing the ratio δ_{\max}/δ_y (see Table 3). For the uncracked
393 series, the values obtained were quite similar, showing that the influence of the adhesive type is not
394 significant. Comparing the two series, ductility increases on cracked series for slabs SL_ADH1,
395 SL_ADH2 and SL_ADH3 of 7.1%, 19.5% and 43.5%, respectively, were obtained. Thus, a trend for
396 ductility increase with pre-cracking was observed, being more pronounced for slabs SL_ADH3. Using the
397 ratio F_r/F_{\max} it is possible to show that the residual force developed by the SL_ADH3 slabs after the
398 maximum load was significantly higher than he one for SL_ADH1 and SL_ADH2 slabs. This comparison
399 is more significant when residual forces were compared with the maximum force of SL_REF. Using the
400 ratio F_r/F_{\max_REF} it is possible to define additional post-failure safety factor of residual load, which
401 determines increase of post-failure slabs strength in comparison to the offered one by the steel
402 reinforcement. For the uncracked series these increases for slabs SL_ADH1, SL_ADH2 and SL_ADH3
403 were of 4.0%, 8.5% and 40.2%, respectively. For the cracked series these increases for slabs SL_ADH1,
404 SL_ADH2 and SL_ADH3 were of 12.8%, 5.6% and 44.0%, respectively. The post-failure residual load
405 carrying capacity of the NSM CFRP system with flexible adhesive (ADH3) is pronounced, when an
406 emergency action on a strengthened structure is considered, to safe human life and property.

407 **4. NUMERICAL MODELLING**

408 This section presents the numerical simulations of the experimentally tested slabs. The parameters of the
409 model were calibrated on the basis of the experiments described in this paper and the work described in
410 [22], where the bond-slip laws for concrete-to-laminate interface for stiff and flexible adhesives are
411 discussed. The results obtained from the numerical simulations provide additional information on the
412 behaviour of the structure (at a local and global level) and are a valuable supplement to the experimental
413 tests. Additionally, the properties calibrated in the numerical model are important to support engineering
414 practice and can be used to design and analyse NSM CFRP strengthening systems in RC existing
415 structures.

416 The slabs were modelled using the DIANA FEA software [23], using material models available in the
417 software's library.

418 **4.1 Finite element model**

419 The finite element mesh topology adopted in the calculations is presented in Figure 13a. The geometry of
420 all specimens as well as loading configuration is symmetrical along the mid-span axis. Therefore, only

421 half-span with the proper boundary conditions was modelled. The finite element mesh of the concrete
422 matrix consists of two-dimensional quadrilateral, eight-node, isoparametric plane stress elements
423 (CQ16M) with thickness of 600 mm in the direction perpendicular to the plane of structure. The structural
424 type of FEM mesh was applied with the maximum dimension of each finite element of 10 mm. A three-
425 node, two-dimensional beam element (CL9BE) with the quadratic interpolation of displacement fields
426 was used for modelling CFRP laminate. The total cross section of the CFRP laminates used as the
427 strengthening system was 2.8 mm in width and 20 mm in height as the two CFRP laminate of 1.4 mm
428 width were used in the experiments. For the implemented finite element model the components of
429 stiffness matrix and internal forces are numerically integrated over the height of its cross-section. Thus,
430 the nonlinear effects can be modelled using this type of finite element but only in the direction parallel to
431 the element axis (nonlinear effects are captured for stresses normal to the cross-section). This means that
432 the model is unable to simulate the shear effects for CFRP tape as well as the effects related to partially
433 loaded areas in the direction perpendicular to the laminate fibres, for example in the vicinity of a crack (a
434 notch effect). However, for the NSM strengthening technique the above mentioned effects have moderate
435 or even negligible influence on the load-displacement behaviour and the failure of the slabs. The effects
436 of relative displacements between the laminate and the concrete matrix were modelled using zero-
437 thickness, six-node interface elements (CL12I). The nodes of the interface, CFRP laminate and concrete
438 elements shared the same locations. The contact perimeter of this interface element considered equal
439 82.8 mm, results from the two CFRP cross-section perimeters (without considering the bottom edges).

440 The configuration of steel reinforcement is presented in Figure 13b. The upper bars and stirrups were
441 modelled using the concept of embedded reinforcement. This means that the reinforcement does not have
442 its own degrees of freedom. This type of element only changes the stiffness matrix of the mother element
443 so the uniaxial strain in the reinforcement element is compatible with the so-called mother element (i.e.
444 an element in which reinforcement is embedded) strain fields in the direction of a bar element. The strain
445 and stress in the embedded reinforcement are therefore calculated from the mother element strain fields.

446 In the case of the bottom reinforcement, local slips in the vicinity of the flexural cracks affect crack
447 spacing, what should be reflected by the numerical model. For this reason, the bottom reinforcement is
448 modelled using three-node truss elements connected with the concrete by special interface elements. This
449 type of connection is able to model relative displacements (slips) between the concrete matrix and
450 reinforcing bars in the direction tangential to the reinforcement. Similarly to the concept of embedded

451 reinforcement, this type of slipping reinforcement can be modelled independently of the connectivity of
 452 concrete elements. In Figure 13b each line represents three $\varnothing 6$ bars for the top reinforcement, two $\varnothing 6$
 453 bars for the stirrups and four $\varnothing 8$ bars for the bottom reinforcement.

454 The effect of lack of compatibility of displacements in the horizontal direction between top and bottom
 455 surfaces of the slabs and the steel loading plate or supporting plate is reflected by the interface elements
 456 (CL12I).

457 **4.2 Material constitutive relationships**

458 4.2.1 Concrete model

459 The constitutive model of concrete applied in the numerical simulations is based on the smeared crack
 460 approach with fixed crack direction [23], [24], [25]. In this approach the nonlinear, uniaxial stress-strain
 461 relationships (Figure 14) are evaluated in the directions of the principal strains. The stiffness matrix is
 462 calculated on the basis of the secant elastic modulus \bar{E}_{comp} and \bar{E}_{tens} , see Figure 14. Following the
 463 fixed crack concept the local directions for evaluating stress-strain relationships are "frozen" after the
 464 appearance of the first crack. In the fixed coordinate system the shear strains and stresses appear. The
 465 shear stiffness is reduced multiplying the shear modulus by a shear retention factor, $\beta < 1.0$. A secondary
 466 crack may appear only in the direction perpendicular to the first crack.

467 In plane stress conditions the constitutive relationship based on the secant stiffness matrix is described by
 468 the following equation [24], [25]:

$$469 \quad \boldsymbol{\sigma} = \mathbf{D}_{sec} \boldsymbol{\varepsilon} \quad (1)$$

470 where: $\boldsymbol{\sigma} = [\sigma_n \quad \sigma_t \quad \tau_{nt}]^T$ is the vector of stresses, $\boldsymbol{\varepsilon} = [\varepsilon_n \quad \varepsilon_t \quad \gamma_{nt}]^T$ is the vector of mechanical
 471 strains, n and t are the directions perpendicular and tangent to the first crack, respectively. Strains $\boldsymbol{\varepsilon}$ are
 472 decomposed in the total strains (total means here mechanical strains and the ones induced by shrinkage)
 473 in the following way:

$$474 \quad \boldsymbol{\varepsilon} = \boldsymbol{\varepsilon}_{tot} - \boldsymbol{\varepsilon}_{sh}, \quad \boldsymbol{\varepsilon}_{sh} = -|\varepsilon_{cs}(t)| \cdot [1 \quad 1 \quad 0]^T \quad (2)$$

475 where $\varepsilon_{cs}(t)$ is the evolution in time t of the mean shrinkage strain due to cement hydration and concrete
 476 drying.

477 In Equation (1), the secant stiffness matrix has the form:

$$478 \quad \mathbf{D}_{sec} = \begin{bmatrix} \bar{E}_n & 0 & 0 \\ 0 & \bar{E}_t & 0 \\ 0 & 0 & \beta \bar{G} \end{bmatrix} \quad (3)$$

479 where \bar{E}_n and \bar{E}_t are the secant elastic modulus in the normal and tangent directions to the first crack,
480 respectively, \bar{G} is the shear modulus. The secant values of the stiffness matrix are calculated from the
481 uniaxial stress-strain relationships. In tension this relationship is assumed as in [26], [27] – see Figure 14:

$$482 \quad \sigma_t = \begin{cases} E_c \cdot \varepsilon & 0 \leq \varepsilon \leq \varepsilon_{cr} \\ f_t \left(\left(1 + \left(c_1 \frac{\varepsilon - \varepsilon_{cr}}{\varepsilon_{cr}^{ult}} \right)^3 \right) \cdot \exp \left(-c_2 \frac{\varepsilon - \varepsilon_{cr}}{\varepsilon_{cr}^{ult}} \right) - \frac{\varepsilon - \varepsilon_{cr}}{\varepsilon_{cr}^{ult}} (1 + c_1^3) \cdot \exp(-c_2) \right) & \varepsilon_{cr} < \varepsilon \leq \varepsilon_{cr}^{ult} \\ 0 & \varepsilon > \varepsilon_{cr}^{ult} \end{cases} \quad (4)$$

483 where E_c is the mean concrete elastic modulus, the same in the elastic part in tension and compression,

484 f_t is the concrete tensile strength, $\varepsilon_{cr} = \frac{f_t}{E_c}$ and the constants $c_1 = 3.0$, $c_2 = 6.93$ are taken from [27].

485 The mesh objectivity of the numerical solution is provided by keeping constant the fracture energy G_{ft}

486 for a given area of a cracked element [28]. Thus, the ultimate strain ε_{cr}^{ult} is calculated as

487 $\varepsilon_{cr}^{ult} = \varepsilon_{cr} + 5.136 \cdot \frac{G_{ft}}{h \cdot f_t}$, where h is the crack bandwidth. For the applied type of finite element,

488 $h = \sqrt{A_{FE}}$ [29], where A_{FE} is the area of an individual finite element.

489 The fracture parameters (tensile strength f_t , fracture energy G_{ft}) were not directly measured in the

490 experimental tests. These material parameters were calculated using correlation formulas according to

491 [30] and [31] on the basis of the experimentally obtained compressive strength of concrete. The adopted

492 values of material parameters of concrete are shown in Table 4.

493 The uniaxial stress-strain curve for concrete in compression is shown in Figure 14. The formula is defined

494 by Equation (5), in compliance with [32]:

$$\sigma_c = \begin{cases} -f_c \frac{1}{3} \frac{\varepsilon}{\varepsilon_{ce}} & \varepsilon_{ce} < \varepsilon \leq 0 \\ -f_c \frac{1}{3} \left(1 + 4 \frac{\varepsilon - \varepsilon_{ce}}{\varepsilon_{c1} - \varepsilon_{ce}} - 2 \left(\frac{\varepsilon - \varepsilon_{ce}}{\varepsilon_{c1} - \varepsilon_{ce}} \right)^2 \right) & \varepsilon_{c1} < \varepsilon \leq \varepsilon_{ce} \\ -f_c \left(1 - \left(\frac{\varepsilon - \varepsilon_{c1}}{\varepsilon_c^{ult} - \varepsilon_{c1}} \right)^2 \right) & \varepsilon_c^{ult} < \varepsilon \leq \varepsilon_{c1} \\ 0 & \varepsilon \leq \varepsilon_c^{ult} \end{cases} \quad (5)$$

496 where $\varepsilon_{ce} = -\frac{1}{3} \frac{f_c}{E_c}$, $\varepsilon_{c1} = 5\varepsilon_{ce}$, f_c is the concrete compressive strength. The values of the compressive

497 strength and elastic modulus of concrete were taken directly from the experimental tests. Similarly to the

498 post-peak behaviour of concrete in tension, compression deformations after the peak stress show a

499 tendency to localize to a certain zone ([33], [34], [35]). This means that the descending part of the stress-

500 strain relationship is size dependent, and stress-displacement description is more suitable in this case than

501 stress-strain relationship. However, it is very convenient in FEM to have the constitutive material model

502 defined by stress-strain relationship (or by their increments), as presented in Equation (5). In order to gain

503 the objectivity of the post-peak behaviour of concrete in compression independently of FE mesh, the

504 ultimate compressive strain is introduced to Equation (5) in the form [32]: $\varepsilon_c^{ult} = \varepsilon_{c1} - \frac{3}{2} \frac{G_{fc}}{h \cdot f_c}$, where

505 G_{fc} is the compressive fracture energy, h is the characteristic length of a finite element, assumed the

506 same as for tension. The G_{fc} value is an additional material property and can be calculated from the

507 post-peak stress-displacement diagram. The values for this quantity available in literature range from

508 10.0 to 25.0 N/mm [33], [36]. The value of G_{fc} adopted in the calculations is shown in Table 4.

509 4.2.2 Bond-slip model between concrete and laminate, model for reinforcing steel, CFRP laminate and

510 concrete-to-bar bond-slip behaviour

511 The behaviour of the concrete-to-laminate interface is described in the work [22]. In the current analysis

512 the average values of material parameters were used for slabs strengthened using the stiff types of the

513 adhesives – Table 4 in [22]. In the case of the flexible adhesive the average bond-slip properties lead to an

514 overestimation of the load bearing capacity of slabs SL_ADH3_U, SL_ADH3_C. This is caused by

515 rheological effects – the flexible adhesive is very sensitive to the loading rate [37], as discussed in the

516 next section of this paper. The higher the strain rate, the stiffer behaviour of polyurethane material, and

517 also the higher the strength and ultimate strain. For lower loading rates, the opposite occurs, due to creep
518 effect. Therefore, in the simulation of the slabs strengthened with the flexible adhesive the effective bond-
519 slip law was employed. The adopted mechanical parameters for the bond-slip law of the flexible adhesive
520 are shown in Table 5. Figure 15 shows the comparison between the bond-slip law obtained from direct
521 pullout tests (DPT) for a loading rate of 5 $\mu\text{m/s}$, as presented in [22], and the effective law assumed for
522 the flexible adhesive for the rate of slip between concrete and laminate that was realized during testing of
523 the strengthened slabs.

524 The bond-slip behaviour of the tensile (bottom) reinforcement is described according to [38] and
525 presented in Figure 16a. The bond stress t_t and the relative bar - concrete slip \bar{u}_t is governed by the
526 following exponential equation:

$$527 \quad t_t = a_{nb} \cdot t_{t \max} \left(1 - \exp \left(\frac{-40 \cdot \bar{u}_t}{\varnothing} \right)^{0.6} \right) \quad (6)$$

528 where $t_{t \max} = 0.9 f_c^{2/3} \text{MPa}$, \varnothing is the diameter of the tensile reinforcement bar and a_{nb} is equal to the
529 number of tensile bars, $a_{nb} = 4$. The mechanical parameters for this bond-slip behaviour were calculated
530 on the basis of the concrete compressive strength given in Table 4. High penalty stiffness was assumed in
531 the direction normal to the bar.

532 The constitutive model for steel is unambiguously defined by the uniaxial stress-strain relationship. The
533 elastic-plastic with linear kinematic hardening model was assumed – Figure 16b. The mechanical
534 parameters adopted for steel were determined on the basis of experimental tests and are summarised in
535 Table 2.

536 The linear elastic behaviour of CFRP laminate was characterized by the elastic modulus, ultimate tensile
537 strength and corresponding ultimate strain according to Table 2. After reaching the ultimate strength the
538 short plateau is assumed (0.5%) followed by sharp drop to zero for the next 0.5% – Figure 16c. It should
539 be noticed that the adopted post-peak behaviour of the laminate does not represent the experimental
540 observations where usually, after reaching the maximum stress, the sudden rupture takes place. The short
541 plateau followed by descending branch were assumed in order to stabilise the numerical solution.

542 **4.3 Strategy for numerical simulations**

543 4.3.1 Loading

544 Three types of loading were tested and simulated numerically. These types of loading include: the dead
545 weight, shrinkage of concrete and external loading in the form of concentrated forces. The loading
546 schemes were applied in the sequences that followed the experimental loading program described in the
547 previous sections.

548 The self-weight was modelled as mass forces imposed at each node of the finite element mesh. The
549 external concentrated load was simulated in two different ways. For the pre-loaded pre-cracked slabs this
550 load was modelled as imposed forces at the node located in the middle of the steel plates used to transmit
551 the load to the slab. For the phase after strengthening the loading process was realized in two steps: first
552 external forces up to approximately 80% of the ultimate load of the slabs were imposed, then the loading
553 was switched to displacement control. The node that controlled the increments of the vertical
554 displacement was located in the middle of the steel plate. The kinetics of shrinkage strains of concrete
555 was assumed according to [30]. Considering that the autogenous shrinkage effects take place when the
556 elastic properties of concrete are not fully developed, only mean drying shrinkage was accounted for. The
557 non-uniform shrinkage distributions associated to humidity gradients over the cross-section of the slabs
558 were neglected as the tests were performed over three months after casting. It was assumed that after this
559 time the moisture fields were uniform, and that self-balanced stresses due to non-uniform shrinkage
560 strains were negligible, considering the reduced thickness of the slabs.

561 4.3.2 Phased analysis and numerical procedure

562 The laboratory experiments entailed several stages that had to be precisely reflected in the numerical
563 simulations. Therefore, a phased analysis with the incremental-iterative procedure was employed in the
564 calculations. The following phases were considered:

- 565 • Phase I – the RC slab without strengthening. The model components that were activated in this phase
566 consist of the concrete elements, steel reinforcement, interface between the bottom rebar elements
567 and concrete elements and kinematic boundary conditions. The initial loads were applied
568 sequentially: shrinkage, dead weight and pre-loading (only for pre-cracked slabs);

- 569 • Phase II – strengthening and loading to approximately 80% of the ultimate load. For this phase the
570 CFRP laminate and concrete-to-laminate interface elements were added to the components of the
571 model from Phase I;
- 572 • Phase III – loading until the failure. In this phase the additional kinematic support was activated on
573 the vertical direction located in the middle of the steel plate.

574 All loadings were applied incrementally. For each load increment, the equilibrium between internal and
575 external forces was verified using the Newton-Raphson procedure. For the last phase the secant BFGS
576 (Broyden–Fletcher–Goldfarb–Shanno) iteration algorithm was applied. The Euclidian norm of
577 displacements and residual forces vectors were used as the convergence criteria. The tolerances were
578 referred to the initial (i.e. at the beginning of each increment) displacements and residual forces vector
579 [23]. The tolerances were 0.01 and 0.02 for displacement norm and unbalanced forces norm, respectively.

580 **4.4 Results of numerical simulations and validation of the model**

581 This section includes the discussion of the results obtained for the above described numerical modelling.
582 The main mechanical quantities measured in the experimental campaign are compared with the numerical
583 predictions. This analysis is summarised in Table 6 and in Figure 17 to Figure 20.

584 Figure 17 presents the results in terms of load – mid-span displacements for slabs strengthened using the
585 stiff adhesives. In this figure also the results for the reference, unstrengthened specimen are shown.
586 Generally, the numerical model precisely follows experimental responses for all loading stages, i.e. before
587 cracking, during the crack formation and after cracking has stabilized, as well as after yielding of the
588 bottom reinforcement. Some minor discrepancies can be noticed for the pre-cracked slabs at the initial
589 phase after strengthening and for the reference slab at the post-yielding stage. The model slightly
590 overestimates the ultimate loads (see Table 6) as well as the mid-span displacement. This is most likely
591 due to the effect of the local action of the vertical crack edge on the CFRP laminate, which causes the
592 earlier rupture of the laminate (a notch effect). Due to the fact that the laminate was modelled using a
593 beam element, such local effects cannot be correctly simulated. It should be noted, however, that the
594 discrepancies caused by these effects are less than 4% and are not important from the point of view of this
595 numerical study.

596 The load-displacement responses for the slabs strengthened with the flexible adhesive are shown in
597 Figure 18. The numerical simulations were performed for two concrete-to-laminate bond-slip models: the

598 average obtained from the DPT tests [22] and the effective model (see Figure 15). It is demonstrated that
599 for the average concrete-to-laminate bond-slip law the numerical model overestimates stiffness after
600 yielding of the bottom reinforcement and the ultimate load both for the uncracked and pre-cracked slabs.
601 This is the result of the different slip rates between the concrete and laminate applied in the DPT tests,
602 and for this reason the average bond-slip model was adopted using the slip rate obtained in the tests on
603 slabs SL_ADH3_U and SL_ADH3_C. The slip rate between the laminate and concrete for these two tests
604 are compared in Figure 19. This figure shows the averaged slip rates over the total length of the CFRP
605 laminate as a function of the mid-span displacement. It can be noticed that the slip rates are
606 approximately one order of magnitude lower than the slip rate adopted in the DPT tests for the flexible
607 adhesive. As mentioned in the preceding paragraph, the constitutive relationships for the flexible adhesive
608 are very sensitive to the load rate [37]. These rheological effects were not directly taken into account
609 mainly due to the lack of experimental data on the rheological behaviour of the flexible adhesives in the
610 DPT tests. Therefore, the simplified approach was applied in the calculations based on the effective bond-
611 slip concrete-to-laminate model. This model was worked out by a trial-and-error procedure and is
612 presented in Figure 15. The results of the simulations for effective bond-slip model are in very good
613 agreement with the experimental measurements up to the ultimate load – see Figure 18. After reaching the
614 maximum value of the external load the simulations became unstable and did not converge to the
615 solution. Therefore, the post critical behaviour of SL_ADH3_U and SL_ADH3_C slabs is not reflected in
616 Figure 18.

617 Figure 20 presents the experimentally and numerically obtained crack pattern for SL_ADH2_U specimen.
618 The cracks in Figure 20b are presented in the form of short lines perpendicular to the direction of the
619 strain ε_n , and these are perceived as numerical cracks. For each finite element only one line section may
620 be formed, representing the maximum value of ε_n from all integration points in the finite element. It
621 should be noted that the line sections in Figure 20b are visible if the cracks have the width greater than
622 0.05 mm, i.e. if $\varepsilon_n \cdot h > 0.05 \text{ mm}$. The first numerical cracks appeared near the stirrups. After cracking
623 stabilisation two to four new cracks appeared between stirrups, which were also observed during the
624 experiments. Therefore, it can be concluded that the numerical model correctly reproduced the spacing
625 between cracks observed experimentally, what is especially important for the simulation of bonding of the
626 laminate between the adjacent cracks.

627 The beams strengthened using the stiff and flexible adhesives showed different slips and bond stress
628 distributions along the laminate – Figure 21. For the specimens strengthened with the stiff adhesive the
629 local anchorage of the laminate near each crack is observed. Due to the high bond strength this form of
630 slip and bond stress development remained until the rupture of the laminate. In the case of the flexible
631 concrete-to-laminate connection the slip and bond stress are smoother over the length of the tape.
632 Therefore, despite the low maximum bonding strength, the laminate is able to carry stresses in the vicinity
633 of the critical cross-sections and contribute to the stiffness of the element after yielding of the bottom
634 reinforcement.

635 **5. CONCLUSIONS**

636 This paper presents an experimental and numerical research on the flexural behaviour of strengthened
637 slabs using the NSM CFRP system and considering the following variables: (i) type of adhesive to fix the
638 CFRP laminate to concrete (stiff and flexible adhesives) and (ii) presence or absence of cracking due to
639 pre-loading. From this study, the following conclusions can be pointed out:

- 640 • The application of the strengthening increases the load carrying capacity of the slabs;
- 641 • For the uncracked series, the cracking load is not influenced by the adhesive type;
- 642 • During yielding, the load is dependent of the adhesive type, being 16% higher for stiff adhesives in
643 the uncracked series and 28% in the cracked ones, when compared to the flexible adhesive;
- 644 • The two stiff adhesives resulted in similar load carrying capacities, which were 23% higher than the
645 one obtained when using the flexible adhesive;
- 646 • The responses of slabs with stiff and flexible adhesives obtained in both uncracked and cracked
647 series are similar, except during the elastic phase since cracked series does not show an elastic phase;
- 648 • Two distinct failure modes were observed depending on the adhesive type. When using stiff
649 adhesives the slabs failed by the CFRP rupture, while when using the flexible adhesive the slabs
650 failed by debonding of the CFRP;
- 651 • When using the flexible adhesive, after the maximum load the CFRP continues to contribute to the
652 load carrying capacity, no rupture occurs and the adhesive continues to provide resistance - post
653 failure residual load carrying capacity, about 40% higher than the residual load carrying capacity
654 assured by steel reinforcement (additional post failure safety);

- 655 • Wider cracks are observed when flexible adhesive is used (an increase of about 50% when compared
656 to the case of stiff adhesive);
- 657 • The crack width is shorter using the flexible adhesive (around 10%), the number of cracks observed
658 is smaller (around 29%) and the average crack spacing is higher (around 18%) than in the case of
659 using stiff adhesives;
- 660 • In general, the flexible adhesive provides slightly lower load carrying capacity values (around 19%
661 less, when compared with the case where stiff adhesives are used), but it can provide a more ductile
662 failure and a higher residual load capacity after failure (around 61% more);
- 663 • It should be noticed that numerical model precisely reflects all loading stages for the slabs
664 strengthened with the stiff adhesives. The maximum discrepancies between the experiments and
665 calculations are less than 4%;
- 666 • The good agreement between the model and the experimental results is also confirmed for the slabs
667 strengthened with the polyurethane adhesive. It was demonstrated that the slip rates between concrete
668 and laminate play a very important role in behaviour of this type of adhesive. The numerical
669 simulations performed for the average bond-slip law, obtained from the DPT tests that were carried
670 out for the slip rate of 5 $\mu\text{m/s}$, overestimate the load bearing capacity, because the derived average
671 slip rate between concrete and laminate during the real tests on slabs was one order of magnitude
672 lower (0.4 $\mu\text{m/s}$). This indicates that the rheological effects have to be considered in the modelling of
673 the bond-slip behaviour for polyurethane types of adhesives. In the presented simulations the
674 simplified approach with the effective bond-slip model was proposed. The calculations with this
675 effective model precisely simulate the experiments up to the failure load. Due to the numerical
676 instabilities the model was not able to simulate the post-critical behaviour of the slabs;
- 677 • Despite the low bond strength of the flexible adhesive (in comparison with the stiff ones) the
678 performance of the flexible strengthening system was quite satisfactory. Due to the compliance of the
679 adhesive the concrete-to-laminate connection exhibits higher effective anchorage length with the
680 smooth bond stress distribution along the laminate. Therefore, the laminate carries high stresses in
681 the vicinity of the critical cross-sections and is able to contribute to the stiffness of the slabs after
682 yielding of the bottom reinforcement. Tests of the flexible adhesives with longer anchorage lengths
683 are required.

Cruz, J.R.; Seręga, S.; Sena-Cruz, J.; Pereira, E.; Kwiecień, A.; Zajac, B. (2020) "Flexural behaviour of NSM CFRP laminate strip systems in concrete using stiff and flexible adhesives" Composites Part B: Engineering, 195: 108042 1-18.

684

685 **ACKNOWLEDGEMENTS**

686 This work was supported by FEDER funds through the Operational Program for Competitiveness Factors
687 – COMPETE and National Funds through FCT (Portuguese Foundation for Science and Technology)
688 under the project FRPLongDur POCI-01-0145-FEDER-016900 (FCT PTDC/ECMEST/1282/2014) and
689 partly financed by the project POCI-01-0145-FEDER-007633. The first and third authors acknowledge
690 the grants SFRH/BD/131259/2017 and SFRH/BSAB/150266/2019 provided by FCT, respectively,
691 financed by European Social Fund and by national funds through the FCT/MCTES. Finally, the authors
692 also like to thank the S&P Clever Reinforcement Ibérica Lda. and SIKA companies for providing the
693 materials.

694

695

696 **REFERENCES**

- 697 [1] Sena-Cruz J, Barros J, Bianco V, Bilotta A, Bournas D, Ceroni F, et al. NSM Systems. In: Pellegrino
698 C, Sena-Cruz J, editors. Design Procedures for the Use of Composites in Strengthening of Reinforced
699 Concrete. RILEM State-of-the-Art Reports 19, Chapter 8, 2015. p. 303-48.
- 700 [2] Sena-Cruz JM, Barros JAO, Coelho MRF, Silva LFFT. Efficiency of different techniques in flexural
701 strengthening of RC beams under monotonic and fatigue loading. *Construction and Building Materials*.
702 2012;29:175-82.
- 703 [3] Al-Saadi NTK, Mohammed A, Al-Mahaidi R, Sanjayan J. Performance of NSM FRP embedded in
704 concrete under monotonic and fatigue loads: state-of-the-art review. *Australian Journal of Structural
705 Engineering*. 2019;20(2):89-114.
- 706 [4] Zhang SS, Yu T, Chen GM. Reinforced concrete beams strengthened in flexure with near-surface
707 mounted (NSM) CFRP strips: Current status and research needs. *Composites Part B: Engineering*.
708 2017;131:30-42.
- 709 [5] Derkowski W, Kwiecień A, Zajac B. CFRP strengthening of bent RC beams using stiff and flexible
710 adhesives. *Technical Transactions. Civil Engineering*. 1-B/2013. Conference, Conference 2013.
- 711 [6] Kwiecień A, Derkowski W, Zajac B. Protection against brittle damage of concrete structure
712 strengthened with CFRP laminates using flexible adhesive - Laboratory test. Conference, Conference
713 2012.
- 714 [7] Kwiecień A. Stiff and flexible adhesives bonding CFRP to masonry substrates—Investigated in pull-
715 off test and Single-Lap test. *Archives of Civil and Mechanical Engineering*. 2012;12(2):228-39.
- 716 [8] D'Antino T, Pisani MA. Evaluation of the effectiveness of current guidelines in determining the
717 strength of RC beams retrofitted by means of NSM reinforcement. *Composite Structures*. 2017;167:166-
718 77.
- 719 [9] ACI. Guide for the design and construction of externally bonded FRP systems for strengthening
720 concrete structures. ACI 440.2R-08. American Concrete Institute Farmington Hills (MI). 2008.

- 721 [10] Society TC. Technical Report 55, Design guidance for strengthening concrete structures using fibre
722 composite materials. 3rd edition, Crowthorne, UK: The Concrete Society. 2012.
- 723 [11] CSA. CSA S806-12, Design and construction of building structures with fiber reinforced polymers.
724 Toronto, Canada: Canadian Standards Association. 2012.
- 725 [12] CNR. CNR. CNR-DT 200 R1/2013. Guide for the design and construction of externally bonded FRP
726 systems for strengthening existing structures. 2013.
- 727 [13] LNEC. Concrete – determination of the elasticity young modulus under compression. E397-1993.
728 Portuguese specification from LNEC. 1993.
- 729 [14] NP EN 12390-3. Testing Hardened Concrete. Part 3: Compressive Strength of Test Specimen.
730 Instituto Português da Qualidade (IPQ). Caparica, Portugal. 2011.
- 731 [15] NP EN EN 1992-1-1. Eurocode 2: design of concrete structures - Part 1-1: general rules and rules for
732 buildings. Instituto Português da Qualidade (IPQ). Caparica, Portugal. 2010.
- 733 [16] NP EN ISO 6892-1. Metallic materials. Tensile testing. Part 1: method of test at room temperature.
734 Instituto Português da Qualidade (IPQ). Caparica, 2012.
- 735 [17] Sena-Cruz J, Jorge M, Branco JM, Cunha VMCF. Bond between glulam and NSM CFRP laminates.
736 Construction and Building Materials. 2013;40:260-9.
- 737 [18] ISO 527-2. Plastics—Determination of Tensile Properties. Part 2: Test Conditions for Moulding and
738 Extrusion Plastics; International Organization for Standardization (ISO): Genève, Switzerland, 2012.
- 739 [19] Correia L, Teixeira T, Michels J, Almeida JAPP, Sena-Cruz J. Flexural behaviour of RC slabs
740 strengthened with prestressed CFRP strips using different anchorage systems. Composites Part B:
741 Engineering. 2015;81:158-70.
- 742 [20] Oehlers DJ, Rashid R, Seracino R. IC debonding resistance of groups of FRP NSM strips in
743 reinforced concrete beams. 2008;22:1574-82.

- 744 [21] Juvandes LFP, Dias SJE, Figueiras JA. Experimental behaviour of concrete slabs strengthened with
745 CFRP systems. In: Proceedings of International Conference Composites in Construction (CCC2001).
746 Conference, Conference 2001. p. 479-85.
- 747 [22] Cruz JR, Sena-Cruz J, Rezazadeh M, Seręga S, Pereira E, Kwiecień A, et al. Bond behavior of NSM
748 CFRP laminate strip systems in concrete using stiff and flexible adhesives. Manuscript submitted to
749 Composites Part B. 2019.
- 750 [23] TNO DIANA BV. DIANA. User's manual.
- 751 [24] Vecchio FJ. Nonlinear finite element analysis of reinforced concrete membranes. ACI Structural
752 Journal. 1989;86(1):26-35.
- 753 [25] Vecchio FJ. Reinforced Concrete Membrane Element Formulations. Journal of Structural
754 Engineering. 1990;116(3):730-50.
- 755 [26] Cornelissen HAW, Hordijk DA, Reinhardt HW. Experimental determination of crack softening
756 characteristics of normalweight and lightweight concrete. Heron. 1986;31(2):45–56.
- 757 [27] Hordijk DA. Local Approach to Fatigue of Concrete. PhD thesis. Delft University of Technology.
758 1991.
- 759 [28] Borst Rd. Smearred cracking, plasticity, creep and thermal loading - a unified approach. Computer
760 Methods in Applied Mechanics and Engineering. 1987;62(1):89–110.
- 761 [29] Rots JG. Computational modeling of concrete fracture. PhD thesis. Delft University of Technology.
762 1988.
- 763 [30] EN 1992-1-1. Eurocode 2: Design of concrete structures - Part 1-1: General rules and
764 recommendations for buildings. CEN - European Committee for Standardization. 2004.
- 765 [31] FIB. Fib Model Code for Concrete Structures 2010. Federation internationale du beton. Ernst &
766 Sohn, a Wiley brand; 2013.
- 767 [32] Feenstra FH, Borst R. Aspects of robust computational modeling for plain and reinforced concrete.
768 Heron. 1993;38(4):3-76.

- 769 [33] Jansen DC, Shah SP. Effect of Length on Compressive Strain Softening of Concrete. Journal of
770 Engineering Mechanics. 1997;123(1):25-35.
- 771 [34] Markeset G, Hillerborg A. Softening of concrete in compression - Localization and size effects.
772 Cement and Concrete Research. 1995;25(4):702-8.
- 773 [35] Mier JGMv, Shah SP, Arnaud M, Balayssac JP, Bascoul A, Choi S, et al. Strain-softening of
774 concrete in uniaxial compression. Materials and Structures. 1997;30(4):195-209.
- 775 [36] Vonk RA. Softening of concrete loaded in compression. PhD thesis. Eindhoven University of
776 Technology. 1992.
- 777 [37] Kwiecień A, Gams M, Rousakis T, Viskovic A, Korelc J. Validation of a New Hyperviscoelastic
778 Model for Deformable Polymers Used for Joints between RC Frames and Masonry Infills. Engineering
779 Transactions. 2017;65(1):113-21.
- 780 [38] Shima H, Chou LL, Okamura H. Micro and Macro Models for Bond in Reinforced Concrete. Journal
781 of the Faculty of Engineering, The University of Tokyo (B). 1987;22:133-94.
- 782
- 783

784 **LIST OF TABLES**

785 **Table 1** – Experimental program.

786 **Table 2** – Material characterization (average values).

787 **Table 3** – Main results obtained from the flexural slab tests.

788 **Table 4** – Mechanical properties of concrete adopted in calculations.

789 **Table 5** – Effective mechanical parameters for flexible adhesives.

790 **Table 6** – Experimental vs predicted cracking, yielding and ultimate loads.

791

792

Table 1 – Experimental program.

Slab's denomination	Type of adhesive	Pre-cracking	CFRP cross-section geometry, $t_f \times w_f$ [mm]
SL_REF	--	--	--
SL_ADH1_U	Adhesive 1 (ADH1)	No	1.4×20
SL_ADH1_C		Yes	
SL_ADH2_U	Adhesive 2 (ADH2)	No	
SL_ADH2_C		Yes	
SL_ADH3_U	Adhesive 3 (ADH3)	No	
SL_ADH3_C		Yes	

793

794

Table 2 – Material characterization (average values).

Concrete			
Curing age	E_c [GPa]	f_c [MPa]	
28 days	27.0 (0.5%)	35.4 (4.8 %)	
110 days	28.3 (2.5%)	38.5 (2.1%)	
Steel			
Steel bar diameter	f_y [MPa]	f_{ts} [MPa]	
Ø6	631.6 (3.4 %)	781.0 (2.4 %)	
Ø8	546.8 (5.3 %)	669.1 (5.6 %)	
CFRP			
Cross-section geometry [mm]	E_f [GPa]	f_f [MPa]	ϵ_{fmax} [$\times 10^{-3}$]
1.4×20 ^a	161.8 (0.9%)	2784.0 (3.9%)	1.7 (3.0%)
Adhesive			
Type of adhesive	E_a [GPa]	f_a [MPa]	ϵ_{amax} [$\times 10^{-3}$]
ADH1	11.7 (0.51%)	25.6 (7.40%)	3.0 (10.91%)
ADH2	7.6 (6.15%)	17.2 (5.43%)	2.5 (13.16%)
ADH3 ^b	0.008	2.2	450.0

795

796 Notes:

797 The values in brackets are the corresponding coefficients of variation (CoV).

798 ^a Results collected from [17].

799 ^b Results collected from [7].

800

801

Table 3 – Main results obtained from the flexural slab tests.

Slab's denomination	Flexural stiffness			Crack initiation		Yielding		Maximum			Ductility parameter	Residual force ratio	FM
	K_I	K_{II}	K_{III}	δ_{cr}	F_{cr}	δ_y	F_y	δ_{max}	F_{max}	ε_{fmax}	δ_{max}/δ_y	F_r/F_{max}	--
	[kN/mm]			[mm]	[kN]	[mm]	[kN]	[mm]	[kN]	[10 ⁻³]	[-]	[%]	--
SL_REF	7.75	0.78	0.01	0.71	7.57	20.17	21.47	158.43 ^a	23.56 ^a	-	-	-	-
SL_ADH1_U	9.57	1.10	0.40	1.25	10.86 (43%)	21.85	31.93 (49%)	74.04	52.87 (124%)	12.06	3.39	46.36	F
SL_ADH2_U	8.95	1.07	0.41	1.35	10.52 (39%)	22.47	31.11 (45%)	74.95	52.08 (121%)	12.49	3.34	49.10	F
SL_ADH3_U	7.94	1.28	0.34	1.58	10.86 (43%)	20.79	27.35 (27%)	72.24	42.71 (81%)	8.46	3.47	77.33	D
SL_ADH1_C	6.30 ^b	1.92	0.41	1.32 ^b	7.16 ^b	18.95	31.58 (47%)	68.87	51.53 (119%)	12.46	3.63	51.56	F
SL_ADH2_C	6.03 ^b	1.91	0.40	0.99 ^b	7.78 ^b	17.36	30.47 (42%)	69.33	51.06 (117%)	12.02	3.99	48.73	F
SL_ADH3_C	5.38 ^b	1.81	0.34	1.06 ^b	6.18 ^b	13.97	24.61 (15%)	69.54	41.82 (78%)	8.33	4.98	80.13	D

802

803 Notes:

804 FM = Failure modes: F = CFRP failure; D = Debonding of the CFRP laminate due to cohesive failure of the
 805 adhesive; the force F_r corresponds to a 90 mm of mid-span vertical displacement for slabs SL_ADH1 and SL_ADH2
 806 and 120 mm for slabs SL_ADH3; the values between parentheses represent the increase in load carrying capacity in
 807 each phase compared to SL_REF.

808 ^a Maximum value reached during the test without failure of the slab (by concrete crushing or failure of the
 809 longitudinal tensile steel bars).

810 ^b Values obtained from the pre-cracking phase (see Section 2.4).

811

812

Table 4 – Mechanical properties of concrete adopted in calculations.

E_c	ν	f_c	f_t	G_{ft}	G_{fc}	β
[GPa]	[-]	[MPa]	[MPa]	[N/mm]	[N/mm]	[-]
28.3	0.2	38.5	2.9	0.14	25.0	0.15

813

814

815

Table 5 – Effective mechanical parameters for flexible adhesives.

s_1	s_2	s_3	τ_m	τ_f	α
[mm]	[mm]	[mm]	[MPa]	[MPa]	[-]
1.5	1.6	2.4	1.3	0.6	0.5

816

817

Table 6 – Experimental vs predicted cracking, yielding and ultimate loads.

	SL_REF	SL_ADH1_U	SL_ADH1_C	SL_ADH2_U	SL_ADH2_C	SL_ADH3_U	SL_ADH3_C
$F_{cr.exp}$ [kN]	7.6	10.9	7.2	10.5	7.8	10.9	6.2
$F_{cr.num}$ [kN]	9.6	10.1	9.6	10.1	9.6	9.6	9.6
$\frac{F_{cr.exp}}{F_{cr.num}}$ [-]	0.79	1.08	0.75	1.04	0.81	1.14	0.65
$F_{y.exp}$ [kN]	21.5	31.9	31.6	31.1	30.5	27.4	24.6
$F_{y.num}$ [kN]	21.1	31.7	31.9	31.7	31.9	27.0	25.5
$\frac{F_{y.exp}}{F_{y.num}}$ [-]	1.02	1.01	0.99	0.98	0.96	1.01	0.96
$F_{max.exp}$ [kN]	23.6	52.9	51.5	52.1	51.1	42.7	41.8
$F_{max.num}$ [kN]	27.6	54.3	53.2	54.3	53.2	42.9*	43.0*
$\frac{F_{max.exp}}{F_{max.num}}$ [-]	0.86	0.97	0.97	0.96	0.96	54.9**	54.9**
						0.99*	0.97*
						0.78**	0.76**

818

819 Notes:

820 *exp*=Experimental; *num*=Numerical modelling

821 *values calculated for effective bond-slip law

822 **values calculated for average bond-slip law

823

824 **LIST OF FIGURES**

825 **Figure 1** – Geometry, reinforcement and strengthening detailing, test configuration, and instrumentation
826 of the slabs: (a) lateral view; (b) elevation; (c) cross section; (d) groove' details; (e-f) photos during the
827 execution of tests. Note: all dimensions are in millimetres.

828 **Figure 2** – Application of the adhesive: (a) Adhesive 1; (b) Adhesive 2; (c) Adhesive 3.

829 **Figure 3** – Force vs. mid-span displacement obtained on the cracked (a) and uncracked (b) series.

830 **Figure 4** – Force vs. mid-span CFRP strain obtained on the (a) uncracked and (b) pre-cracked series.

831 **Figure 5** – Force vs. mid-span steel strain obtained on the (a) uncracked and (b) pre-cracked series

832 **Figure 6** – Force vs. mid-span concrete strain obtained on the (a) uncracked and (b) cracked series.

833 **Figure 7** – Failure modes: (a) CFRP laminate failure (SL_ADH1 and SL_ADH2); (b) debonding at
834 laminate-adhesive interface (SL_ADH3); (c) cohesive failure of ADH3 at the ends of the strengthening
835 system (SL_ADH3); (d) cohesive failure of the adhesive between the mid-span section and the ends of
836 the strengthening (SL_ADH3).

837 **Figure 8** – Crack width evolution on the (a) cracked and (b) uncracked series.

838 **Figure 9** – Crack pattern of each slab after the test on lateral and bottom surfaces. Notes: on reference
839 and uncracked slabs, the cracks were marked using black colour; on the cracked slabs, the cracks caused
840 by the pre-cracking process were marked with red colour and the cracks resulting of the test up to failure
841 were marked using black colour.

842 **Figure 10** – Average crack distance of each slab.

843 **Figure 11** – Force at: (a) crack initiation; (b) bottom steel yielding. Notes: the values between parentheses
844 are the percentage increase to SL_REF at this phase of the test.

845 **Figure 12** – Maximum force (a) and maximum CFRP strain (b). Note: the values between parentheses are
846 the percentage increase to SL_REF at this phase of the test.

847 **Figure 13** – Finite element model adopted for the simulation of the slabs: (a) concrete and laminate,
848 (b) steel reinforcement.

849 **Figure 14** – Uniaxial stress-strain relationship for concrete.

850 **Figure 15** – Comparison of bond-slip laws for Adhesive 3, average according to DPT (red line) [22],
851 effective (black line).

852 **Figure 16** – (a) bond-slip law for steel reinforcement, (b) stress-strain relationship for steel
853 reinforcement, (c) stress-strain relationship for CFRP laminate.

854 **Figure 17** – Comparison between experimental and numerical modelling of the load vs. displacement
855 behaviour for slabs with stiff adhesives: (a) slabs uncracked before strengthening, (b) slabs preloaded and
856 cracked before strengthening.

857 **Figure 18** – Comparison between experimental and numerical modelling of the load vs. displacement
858 behaviour for slabs with flexible adhesive: (a) slab uncracked before strengthening, (b) slab preloaded and
859 cracked before strengthening.

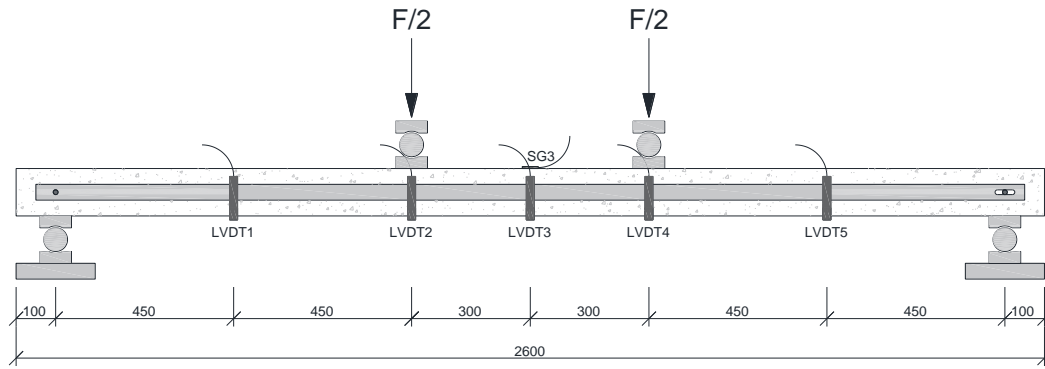
860 **Figure 19** – Averaged slip rate between CFRP laminate and concrete for SL_ADH3_U slab.

861 **Figure 20** – Comparison between the experimental and numerical crack pattern for slab SL_ADH2_U.

862 **Figure 21** – Development of slips (a), (c) and bond stresses (b), (d) for slabs SL_ADH1_U (stiff
863 adhesive) and SL_ADH3_U (flexible adhesive) – external load of 40 kN.

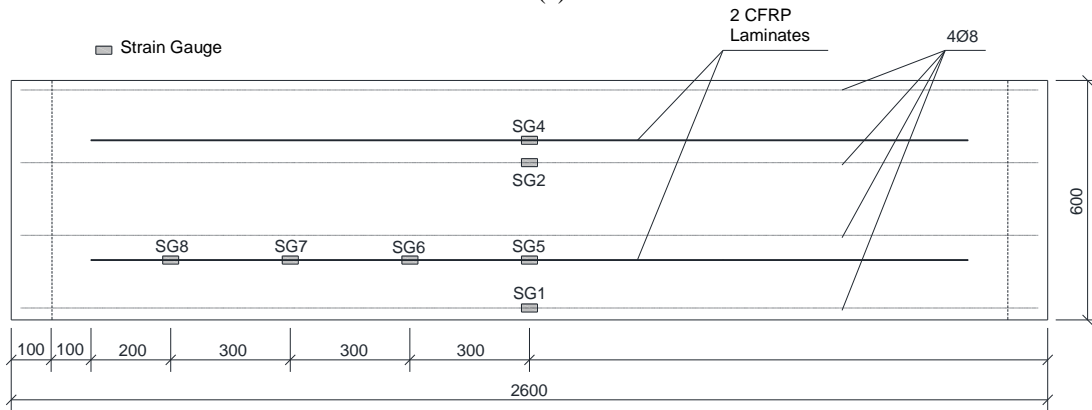
864

865
866



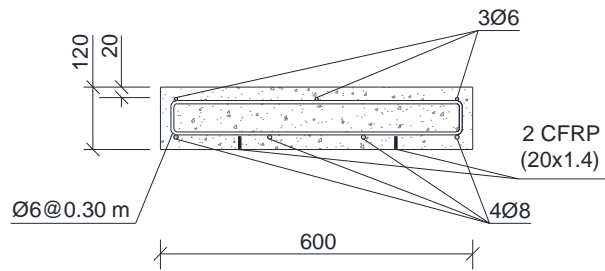
(a)

867
868

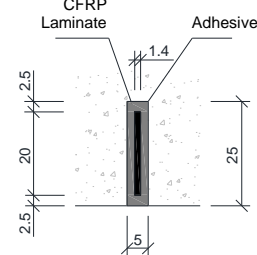


(b)

869
870
871



(c)



(d)

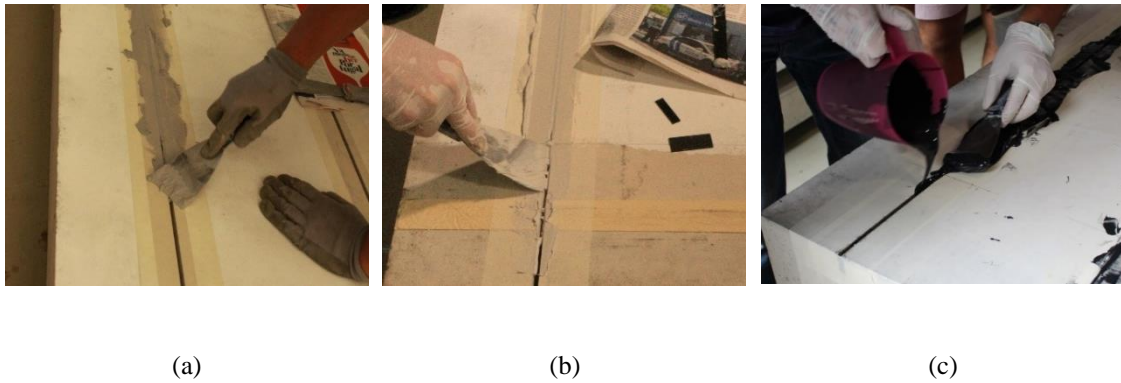


(e)



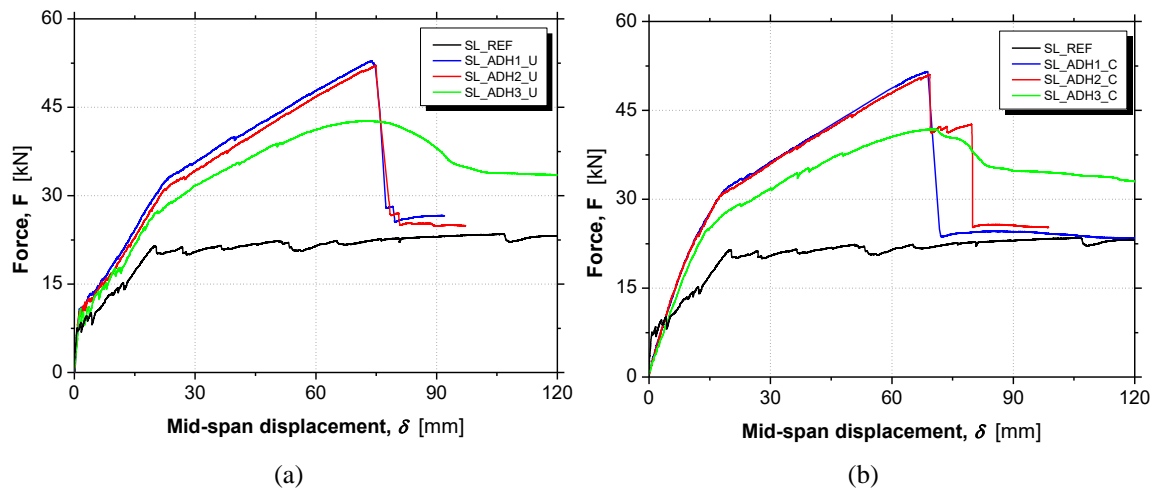
(f)

872 **Figure 1** – Geometry, reinforcement and strengthening detailing, test configuration, and instrumentation
873 of the slabs: (a) lateral view; (b) elevation; (c) cross section; (d) groove' details; (e-f) photos during the
874 execution of tests. Note: all dimensions are in millimetres.



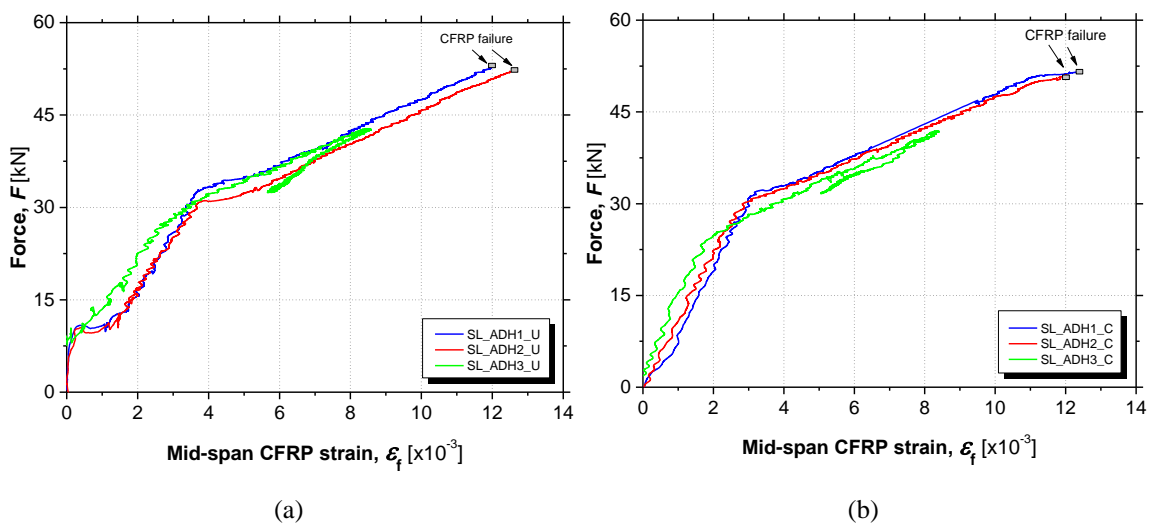
875 **Figure 2** – Application of the adhesive: (a) Adhesive 1; (b) Adhesive 2; (c) Adhesive 3.

876

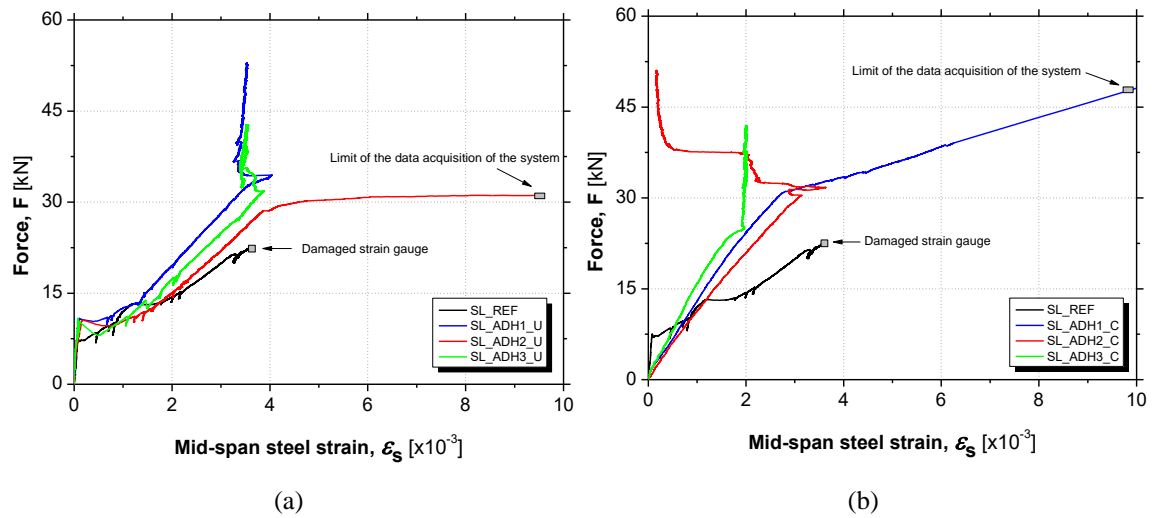


877 **Figure 3** – Force vs. mid-span displacement obtained on the cracked (a) and uncracked (b) series.

878

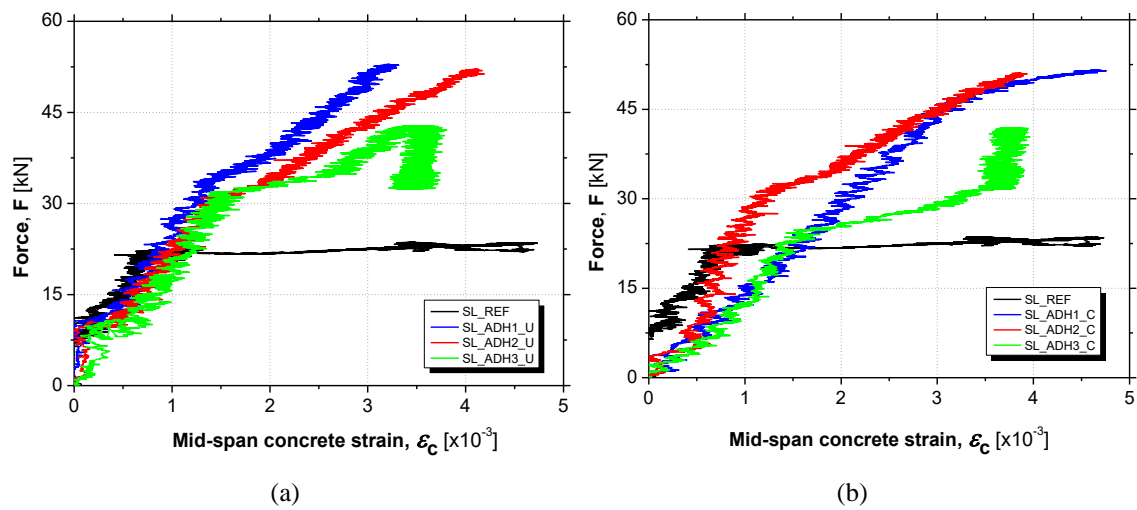


879 **Figure 4** – Force vs. mid-span CFRP strain obtained on the (a) uncracked and (b) pre-cracked series.



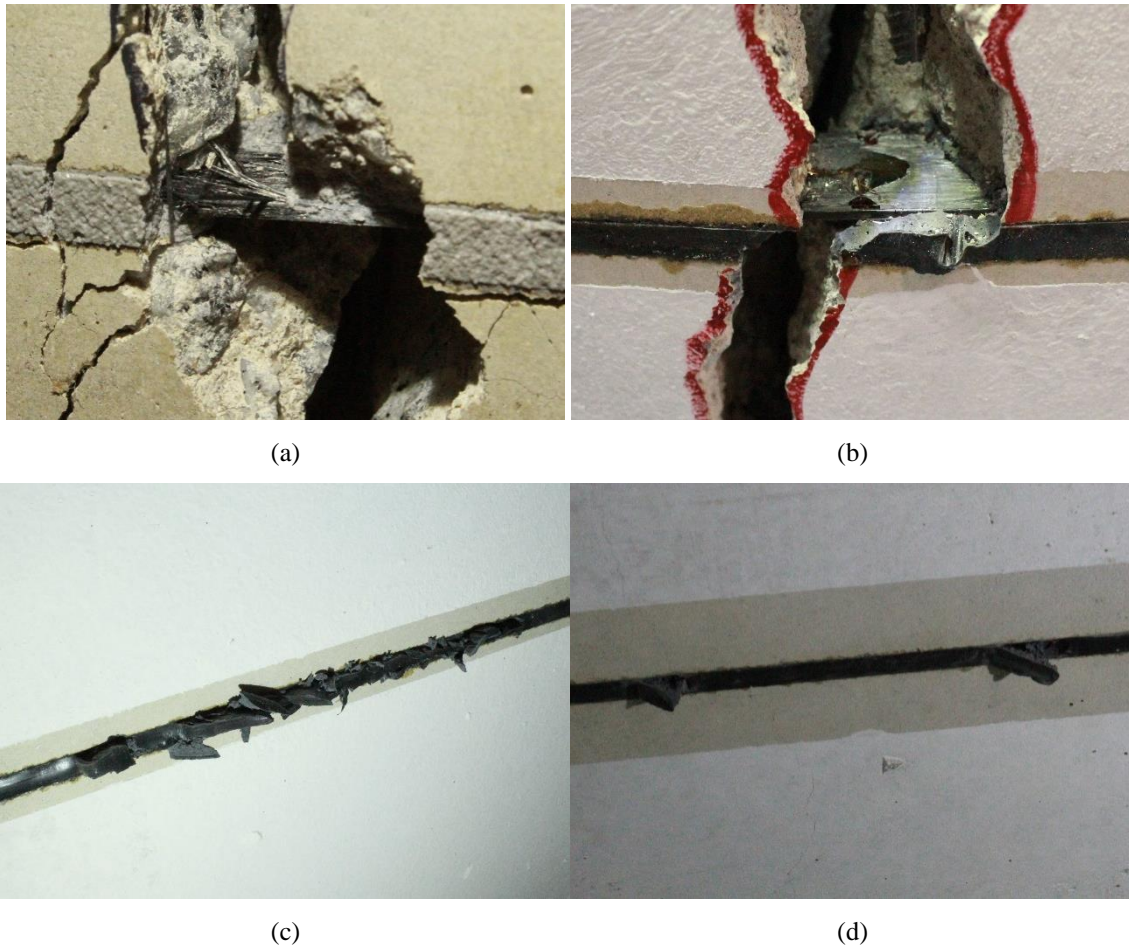
880 **Figure 5** – Force vs. mid-span steel strain obtained on the (a) uncracked and (b) pre-cracked series

881

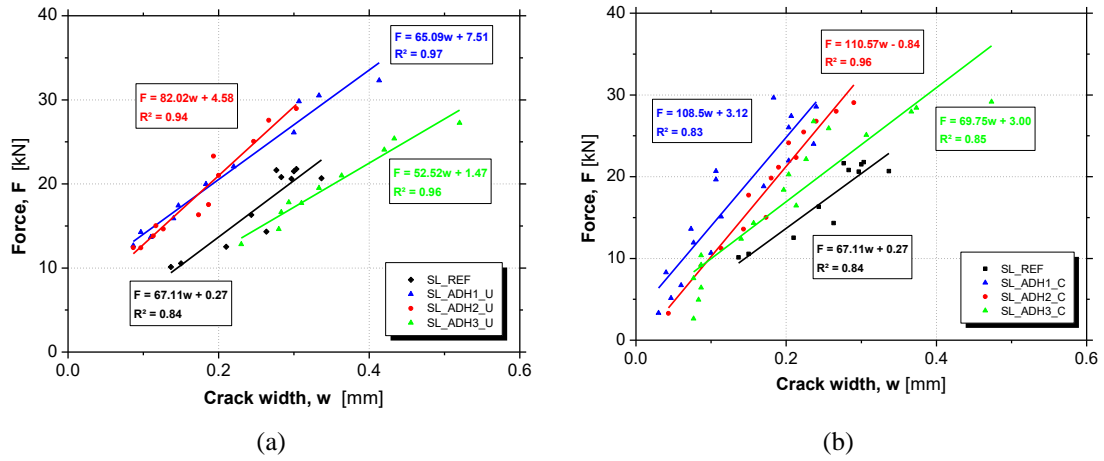


882 **Figure 6** – Force vs. mid-span concrete strain obtained on the (a) uncracked and (b) cracked series.

883



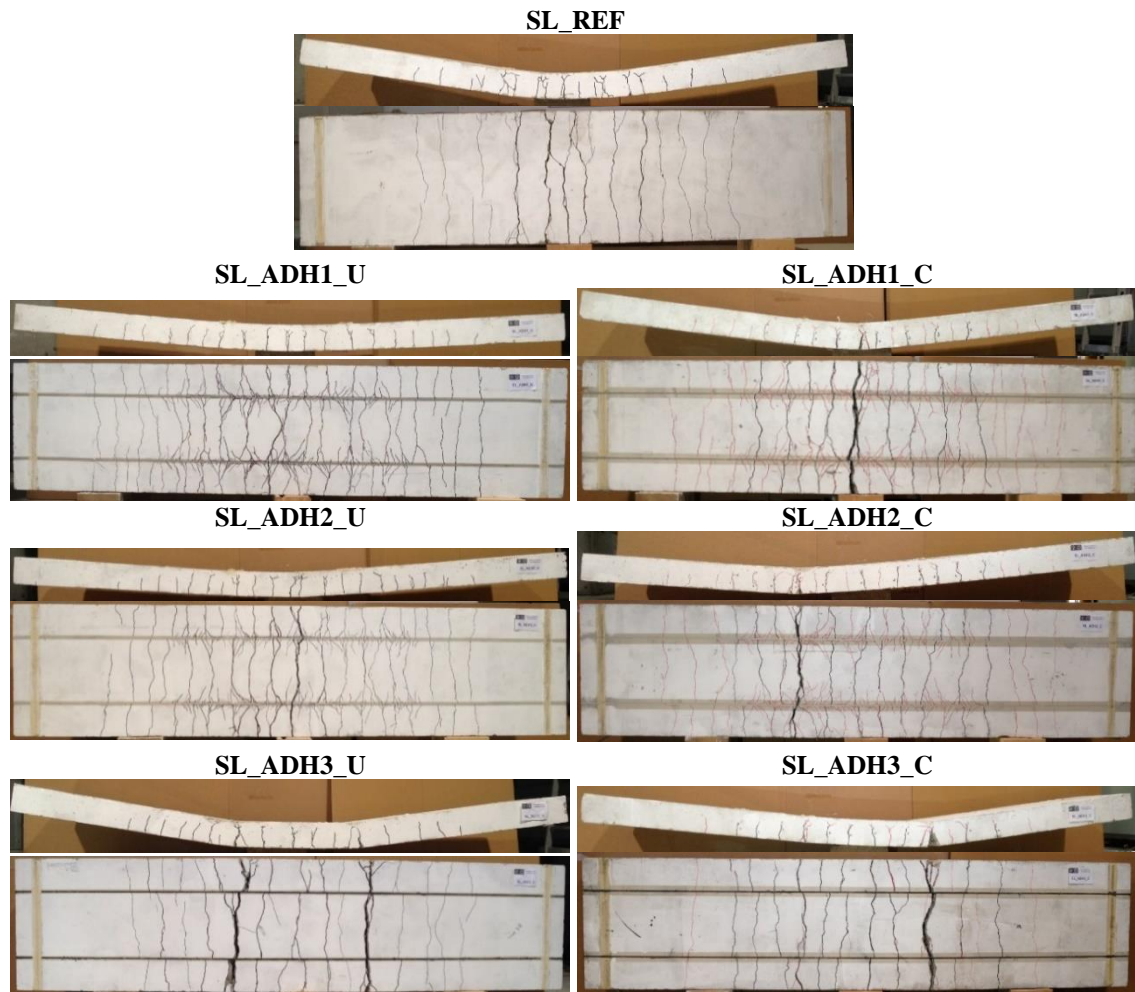
884 **Figure 7** – Failure modes: (a) CFRP laminate failure (SL_ADH1 and SL_ADH2); (b) debonding at
885 laminate-adhesive interface (SL_ADH3); (c) cohesive failure of ADH3 at the ends of the strengthening
886 system (SL_ADH3); (d) cohesive failure of the adhesive between the mid-span section and the ends of
887 the strengthening (SL_ADH3).
888



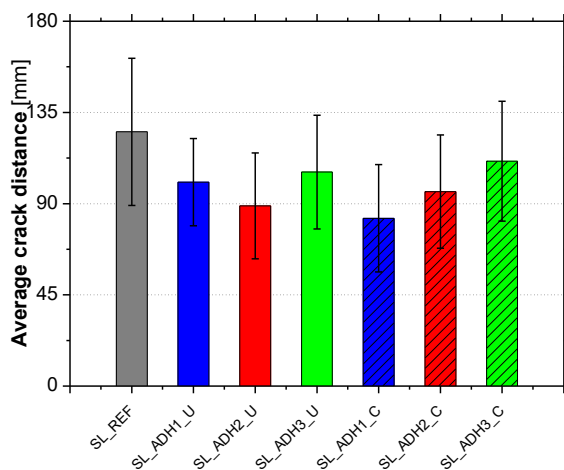
889

Figure 8 – Crack width evolution on the (a) cracked and (b) uncracked series.

890



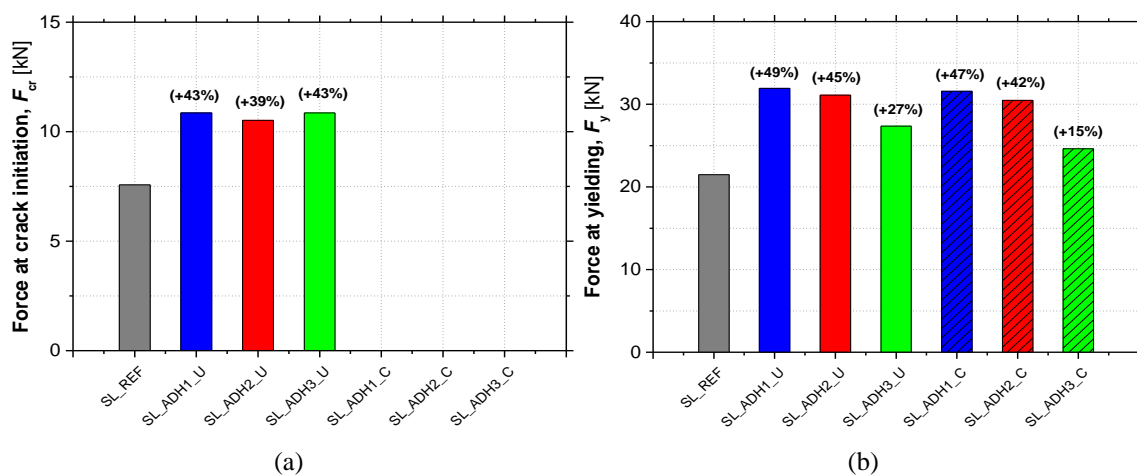
891 **Figure 9** – Crack pattern of each slab after the test on lateral and bottom surfaces. Notes: on reference
892 and uncracked slabs, the cracks were marked using black colour; on the cracked slabs, the cracks caused
893 by the pre-cracking process were marked with red colour and the cracks resulting of the test up to failure
894 were marked using black colour.
895



896

Figure 10 – Average crack distance of each slab.

897



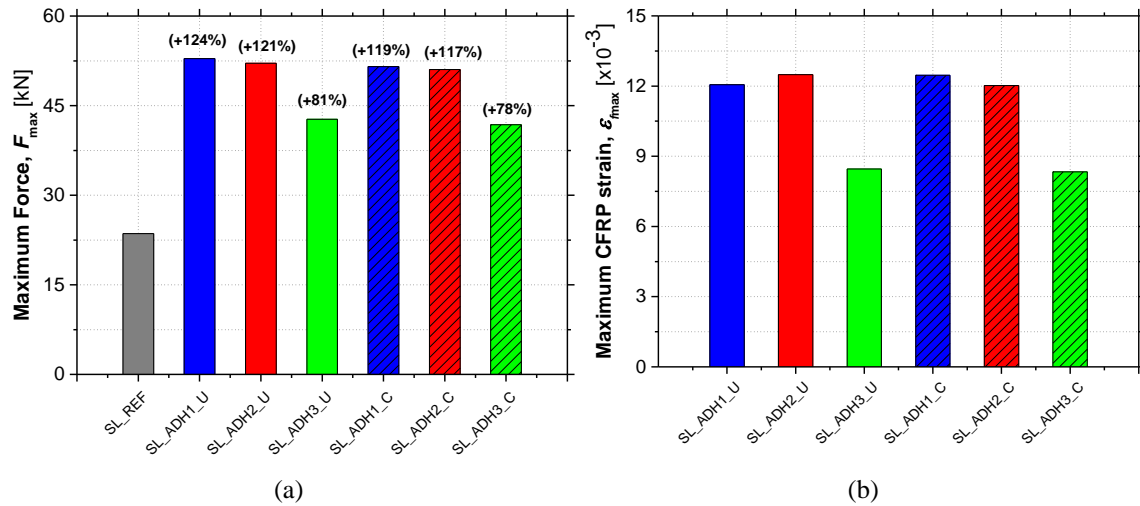
898

Figure 11 – Force at: (a) crack initiation; (b) bottom steel yielding. Notes: the values between parentheses

899

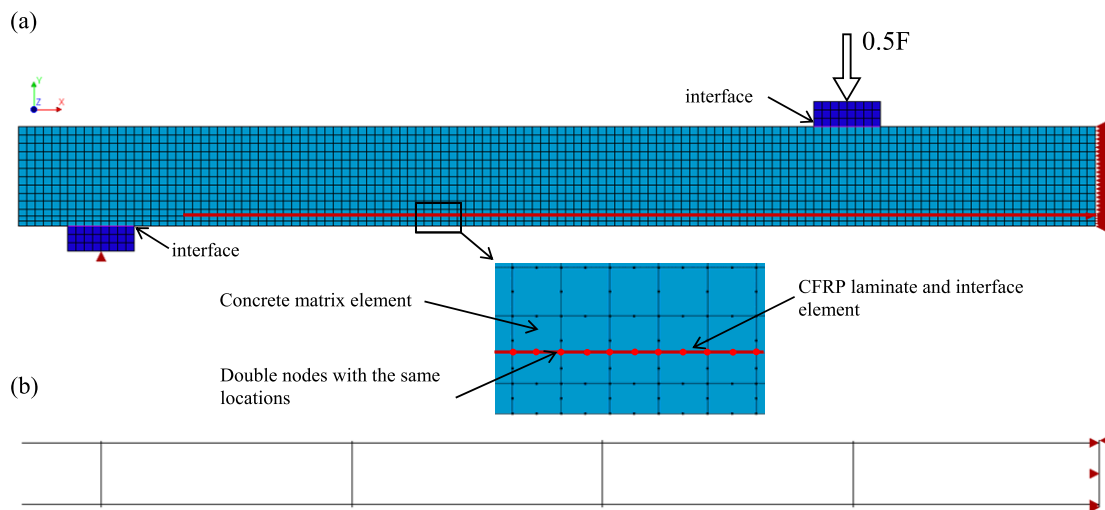
are the percentage increase to SL_REF at this phase of the test.

900



901 **Figure 12** – Maximum force (a) and maximum CFRP strain (b). Note: the values between parentheses are
 902 the percentage increase to SL_REF at this phase of the test.

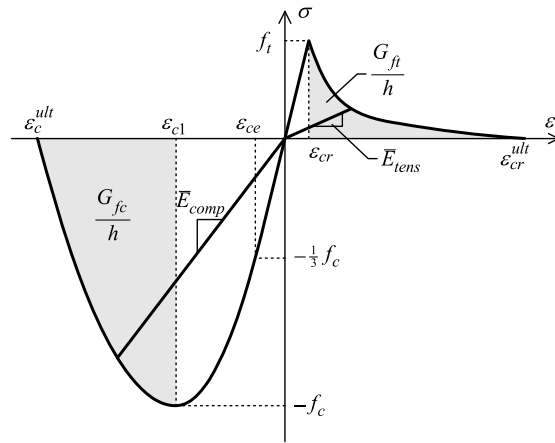
903



904

905 **Figure 13** – Finite element model adopted for the simulation of the slabs: (a) concrete and laminate,
 906 (b) steel reinforcement.

907

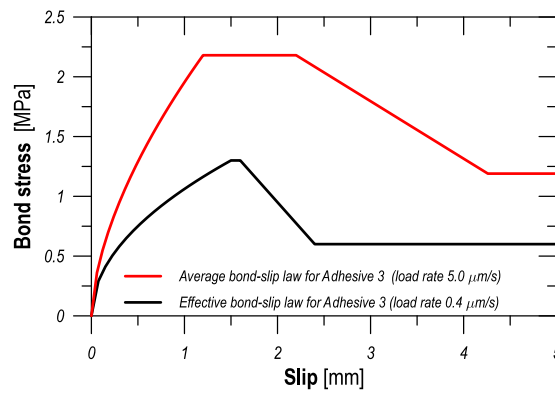


908

909

Figure 14 – Uniaxial stress-strain relationship for concrete.

910



911

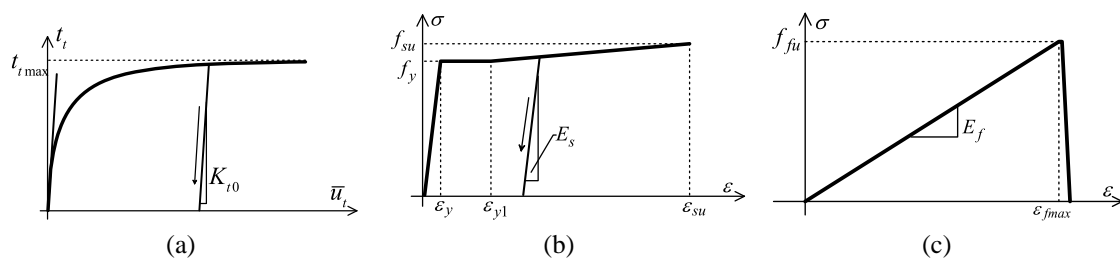
912

Figure 15 – Comparison of bond-slip laws for Adhesive 3, average according to DPT (red line) [22],

913

effective (black line).

914



915

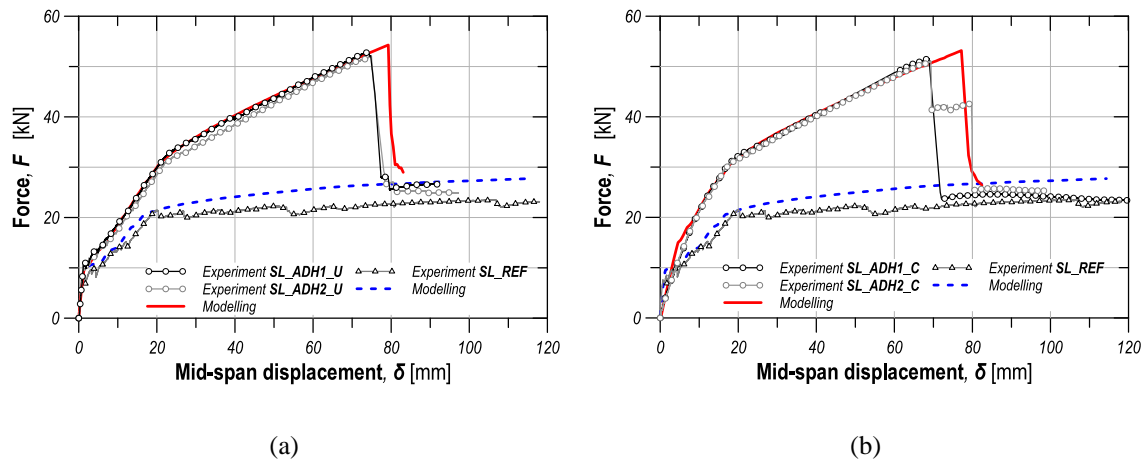
Figure 16 – (a) bond-slip law for steel reinforcement, (b) stress-strain relationship for steel

916

reinforcement, (c) stress-strain relationship for CFRP laminate.

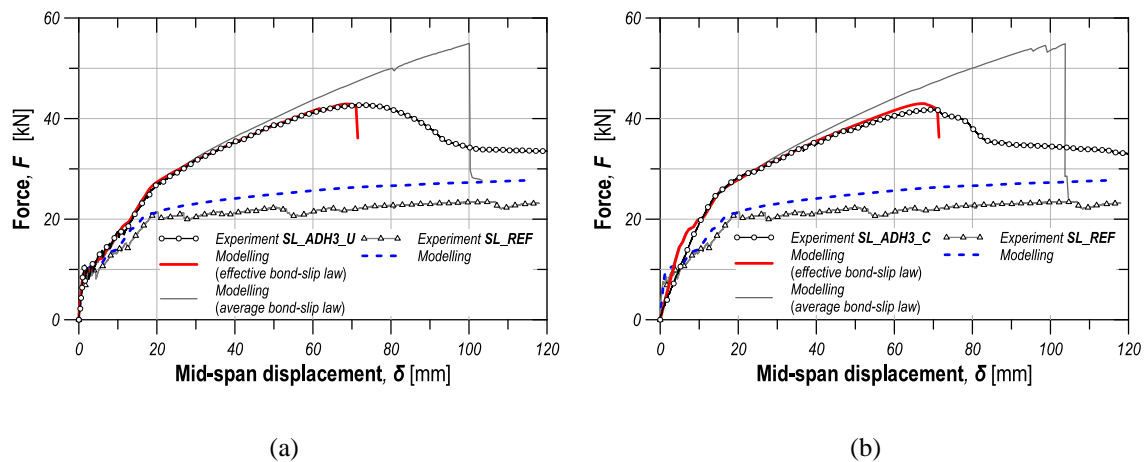
917

918



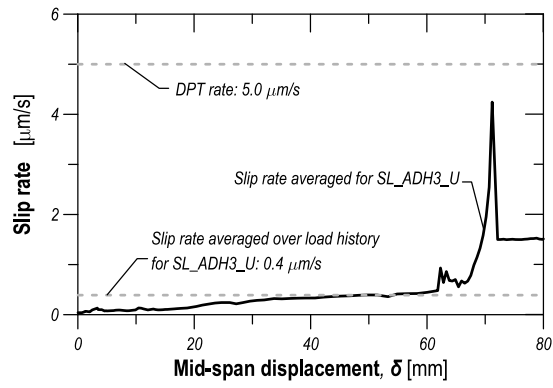
919 **Figure 17** – Comparison between experimental and numerical modelling of the load vs. displacement
 920 behaviour for slabs with stiff adhesives: (a) slabs uncracked before strengthening, (b) slabs preloaded and
 921 cracked before strengthening.

922



923 **Figure 18** – Comparison between experimental and numerical modelling of the load vs. displacement
 924 behaviour for slabs with flexible adhesive: (a) slab uncracked before strengthening, (b) slab preloaded and
 925 cracked before strengthening.

926



927

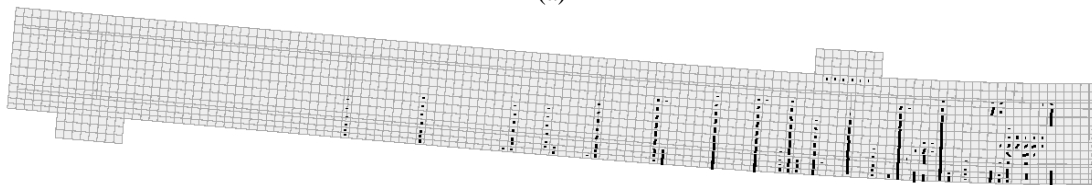
928

Figure 19 – Averaged slip rate between CFRP laminate and concrete for SL_ADH3_U slab.

929



(a)

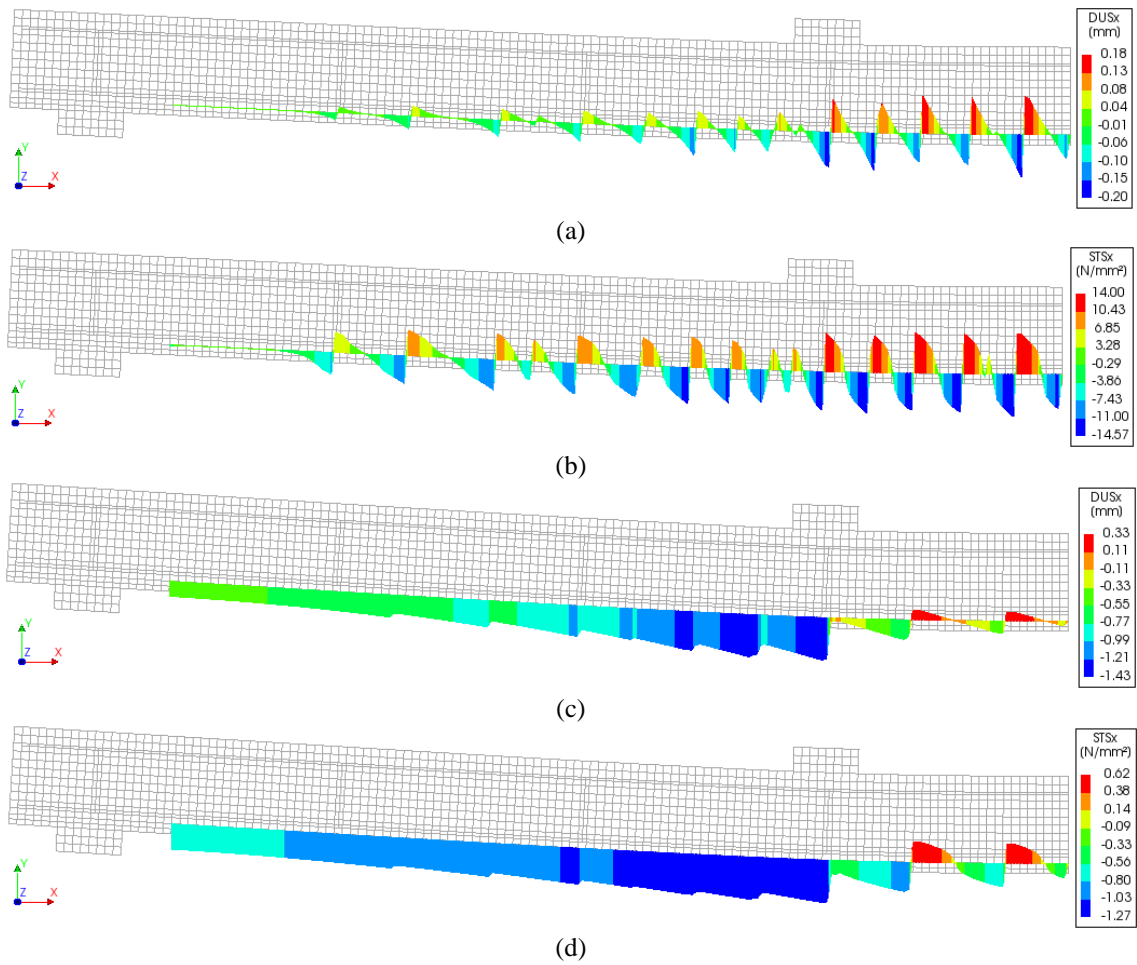


(b)

930

Figure 20 – Comparison between the experimental and numerical crack pattern for slab SL_ADH2_U.

931



932

933

Figure 21 – Development of slips (a), (c) and bond stresses (b), (d) for slabs SL_ADH1_U (stiff

934

adhesive) and SL_ADH3_U (flexible adhesive) – external load of 40 kN.